Meteoroid Impact Detections by the Gaia spacecraft at L2

Gaia Flight Control Team European Space Operations Centre Germany

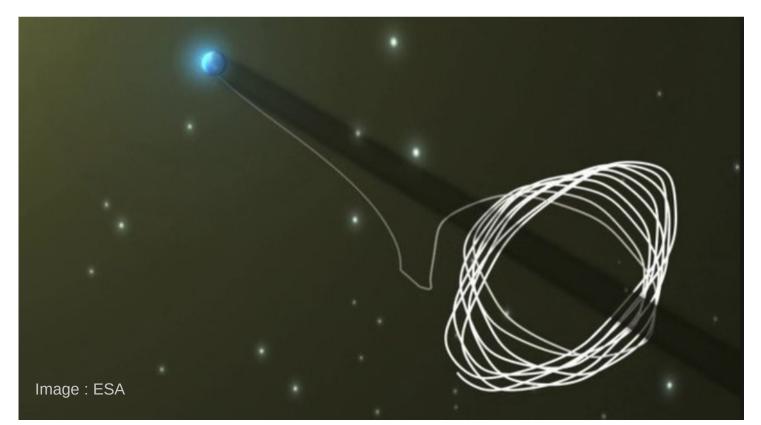
Meteoroids 2016

Presentation Contents

- Introduction to Gaia
- Measurable meteoroid influences
- Meteoroid detection by rate disturbance
- Meteoroid detection by thermal disturbance
- Meteoroid properties calculation
- Meteoroid origin determination
- Conclusion

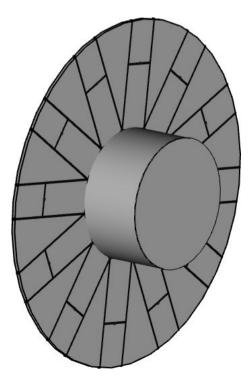
Introduction to Gaia

- Gaia is the European Space Agency's astrometry mission to measure the positions of >1 billion stars in the Milky Way galaxy
- In a halo orbit around the second Sun-Earth Lagrange point (L2) 1.5 million km away from Earth, it has been conducting routine measurements since July 2014 and will continue to at least 2019.
- Gaia is a survey mission which equipped with two 35m telescopes (~f35) separated by a fixed angle of 106.5°



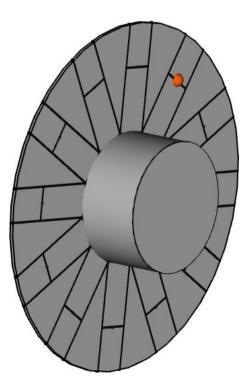
Measurable meteoroid influences on Gaia

• Gaia is shaped like a top-hat with a cylinder containing the service and payload modules and a disk that performs the function of a sun-shield.



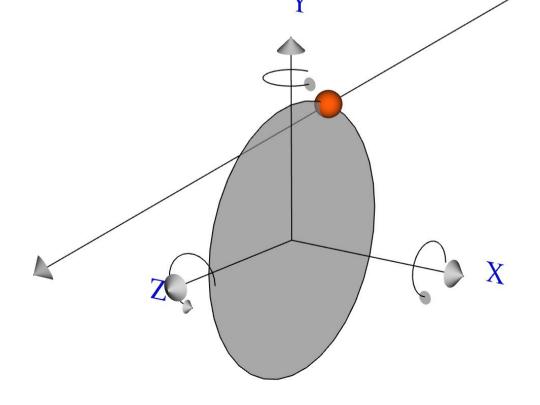
Measurable meteoroid influences on Gaia

- Gaia is shaped like a top-hat with a cylinder containing the service and payload modules and a disk that performs the function of a sun-shield.
- Meteoroid impact energy dissipation causes local heating and structural deformation

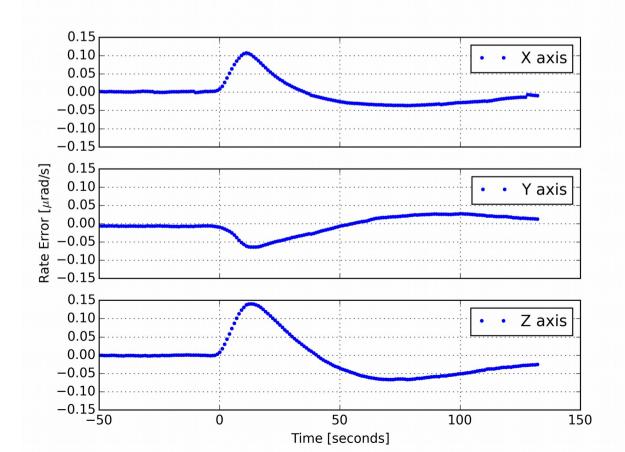


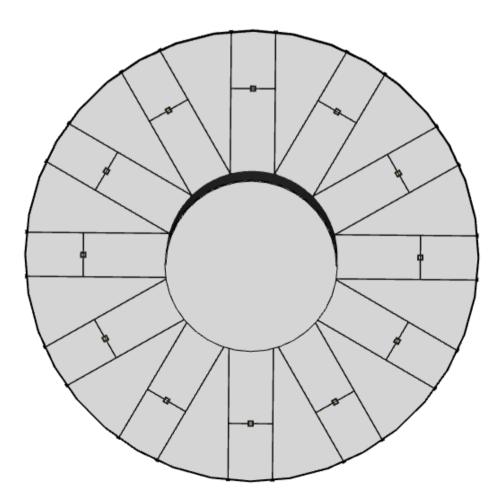
Measurable meteoroid influences on Gaia

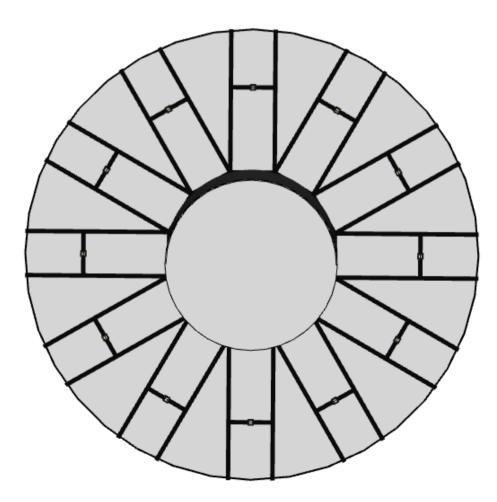
- Gaia is shaped like a top-hat with a cylinder containing the service and payload modules and a disk that performs the function of a sun-shield.
- Meteoroid impact energy dissipation causes local heating and structural deformation
- Meteoroid impact forces cause rotations in 3 axes (and linear delta V)

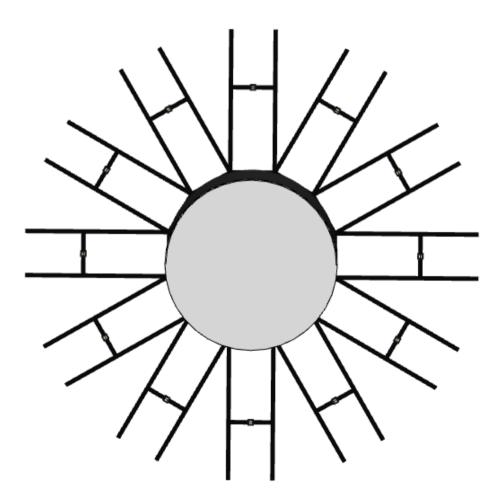


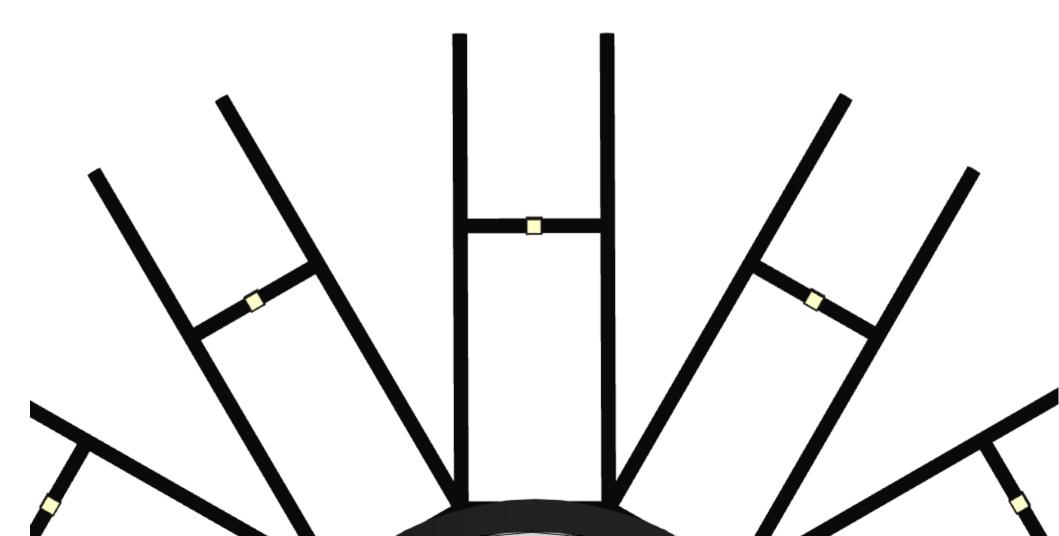
- From the payload data the attitude computer calculates body rate errors and applies corrective torques with the cold gas thrusters
 - Reaction time typically 10 seconds
 - To create these spacecraft rates requires ~100pJ of energy

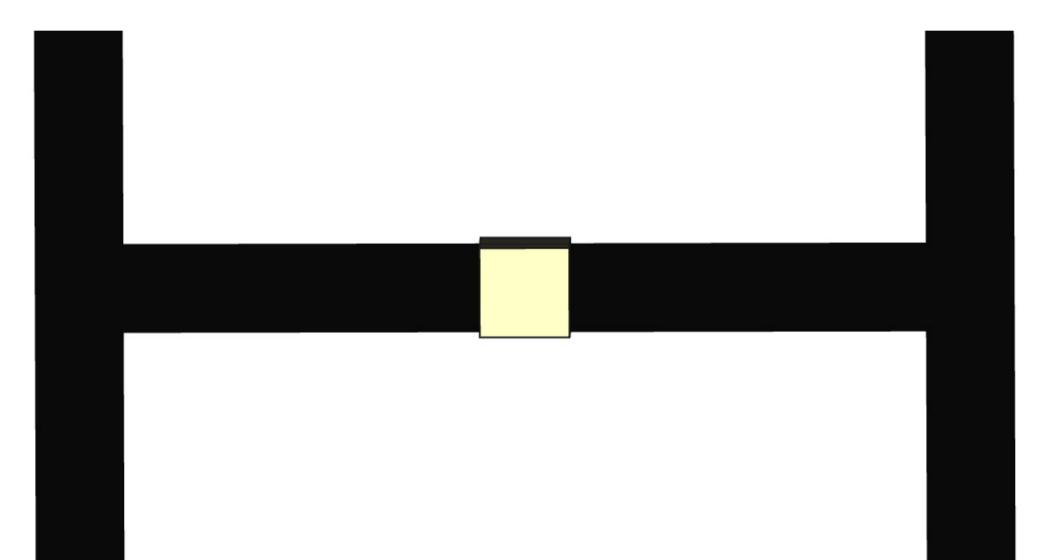


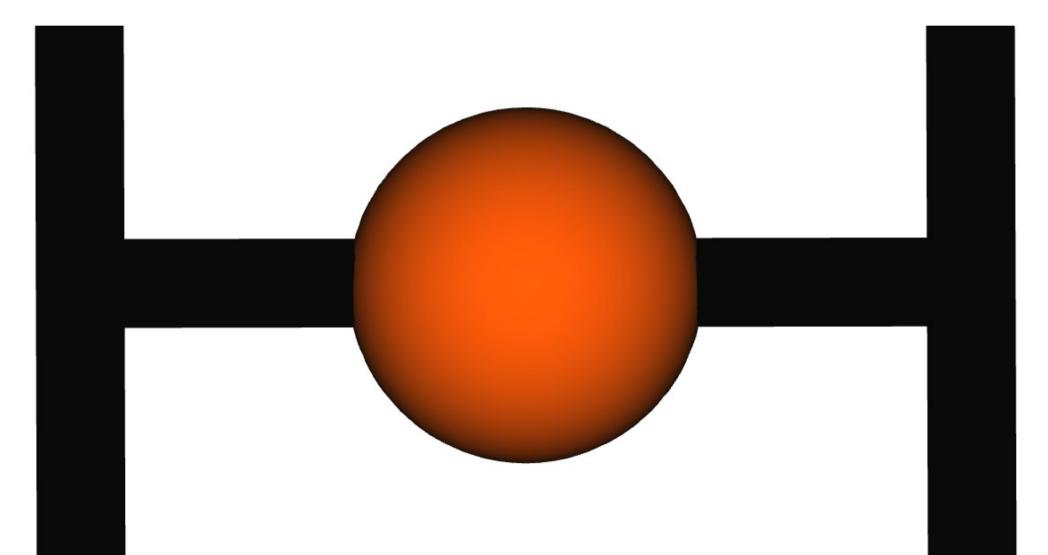






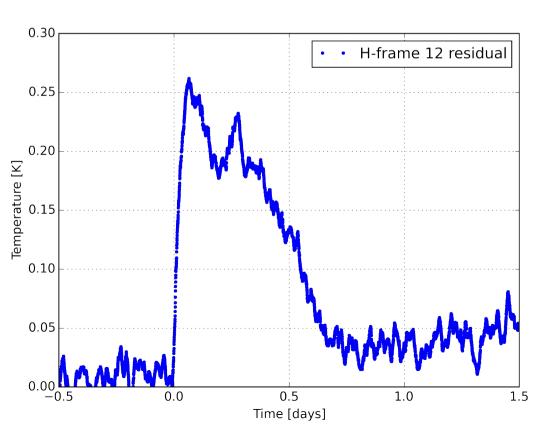




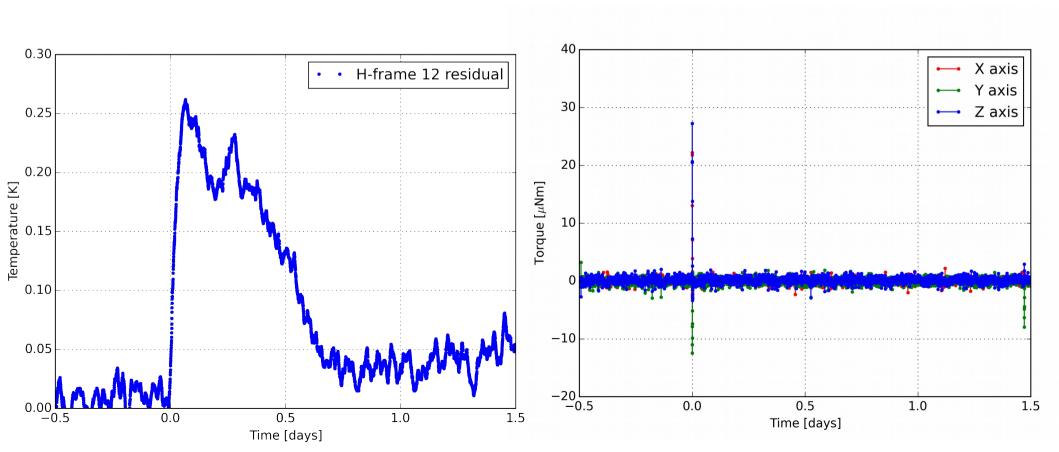


- The sun-shield temperature is monitored at 12 points equally distributed on the support structure
- Sun-shield temperature changes are very predictable and depend on;
 - Sun distance & solar output (it is possible to measure changes due to sunspots)
 - Sun azimuth (spacecraft rotation phase)
 - Ageing term
 - Operations activities
- Removing the effects of these influences leaves residuals which appear to be caused by meteoroid strikes
 - Permanent offsets that are due to structural deformation of the sun-shield
 - Temporary offsets that are due to dissipated energy of the impact

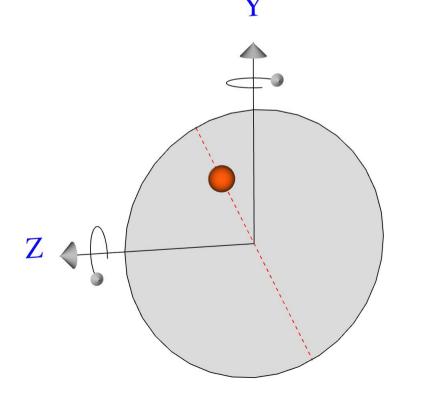
• The first observation of a thermal disturbance was initially mysterious



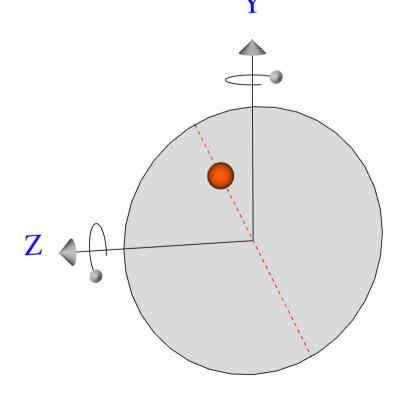
- The first observation of a thermal disturbance was initially mysterious
- It was quickly realised to coincide with a large attitude disturbance

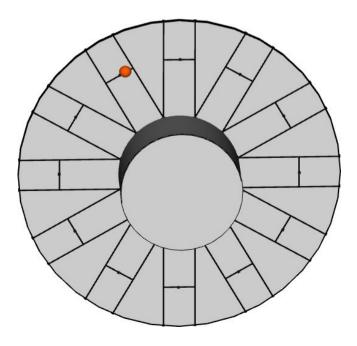


- The first observation of thermal disturbance was initially mysterious
- It was quickly realised to coincide with a large attitude disturbance
- Confirmation of cause and effect by calculation of impact diagonal from Y/Z spacecraft rates and correlation with location of temperature monitor
 - Additionally now know which side of disk was impacted
 - And length of lever arm

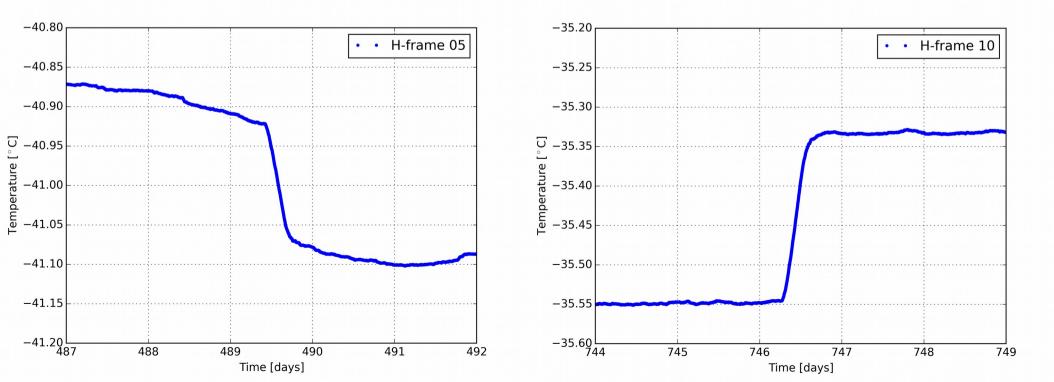


- The first observation of thermal disturbance was initially mysterious
- It was quickly realised to coincide with a large attitude disturbance
- Confirmation of cause and effect by calculation of impact diagonal from Y/Z spacecraft rates and correlation with location of temperature monitor
 - Additionally now know which side of disk was impacted
 - And length of lever arm



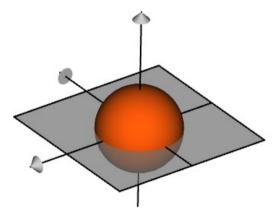


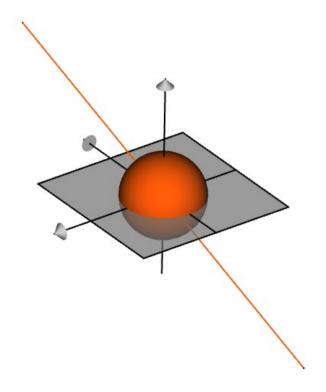
- Permanent changes to local thermal balance have also been observed to occur coincidentally with attitude disturbances
 - changes are due to structural deformation
- No temporary heating has been observed for these events
 - Did not impact hard structure (only insulation)? Did not stop?
- Approximately 1 thermal event observed per month.

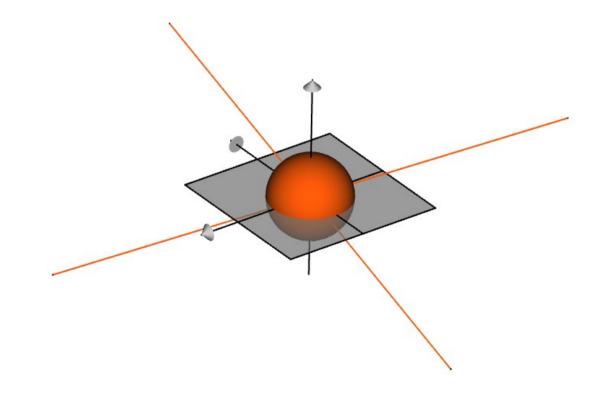


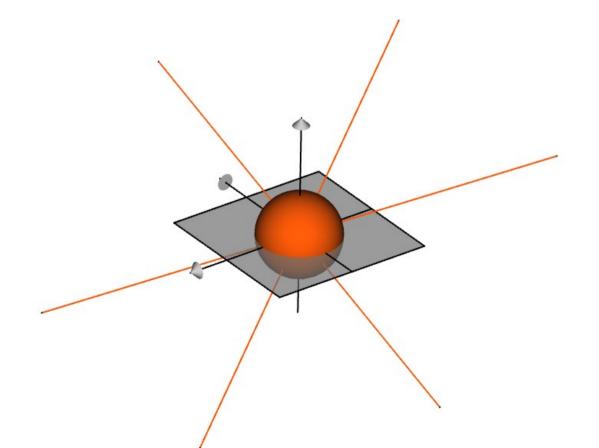
Meteoroid properties calculation

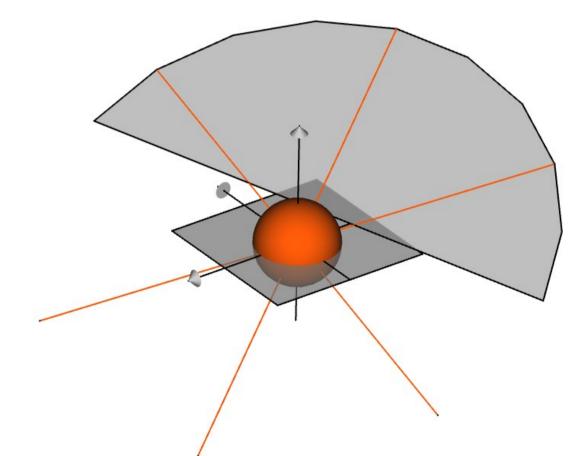
- The energy from the impactor is dissipated through several mechanisms of which only the angular momentum and thermal effects are measurable by Gaia.
- From impacts with a temporary heating effect it is possible to calculate a lower limit for the energy of the impactor
 - Assuming that thermistor location in a 350g Aluminium box has reached thermal equilibrium
 - with heat capacity just under 1J/gK
 - ΔT of 0.25K => Energy = 84J
- The heat energy is many (12) orders of magnitude greater than the angular momentum energy
- From the energy it is possible to calculate the mass, assuming a typical impact velocities;
 - At 20km/s 84J => Mass = 0.42mg
 - At 60km/s 84J=> Mass = 0.05mg



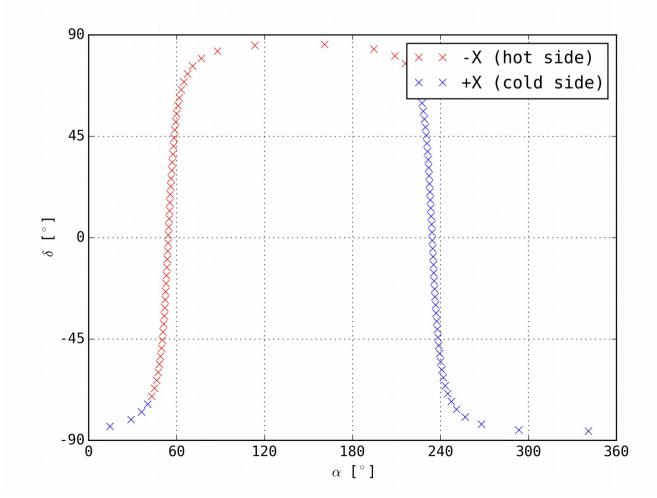








- Thermal measurements provide a point of impact. Imparted rates from these impacts can be used to calculate an arc of possible impact trajectories
- Plotted onto the sky shows the arc of possible origins



Conclusion

- Gaia is detecting meteoroids in-situ at L2
- Energy limits of some impacts can be measured
- Directional information can be used to restrict possible source origins
- Data could be used to check models

Conclusion

- Gaia is detecting meteoroids in-situ at L2
- Energy limits of some impacts can be measured
- Directional information can be used to restrict possible source origins
- Data could be used to check models

