

Meteoroid Impact Detection for Exploration of Asteroids (MIDEA): A concept for asteroid prospecting

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**Stanford
University**

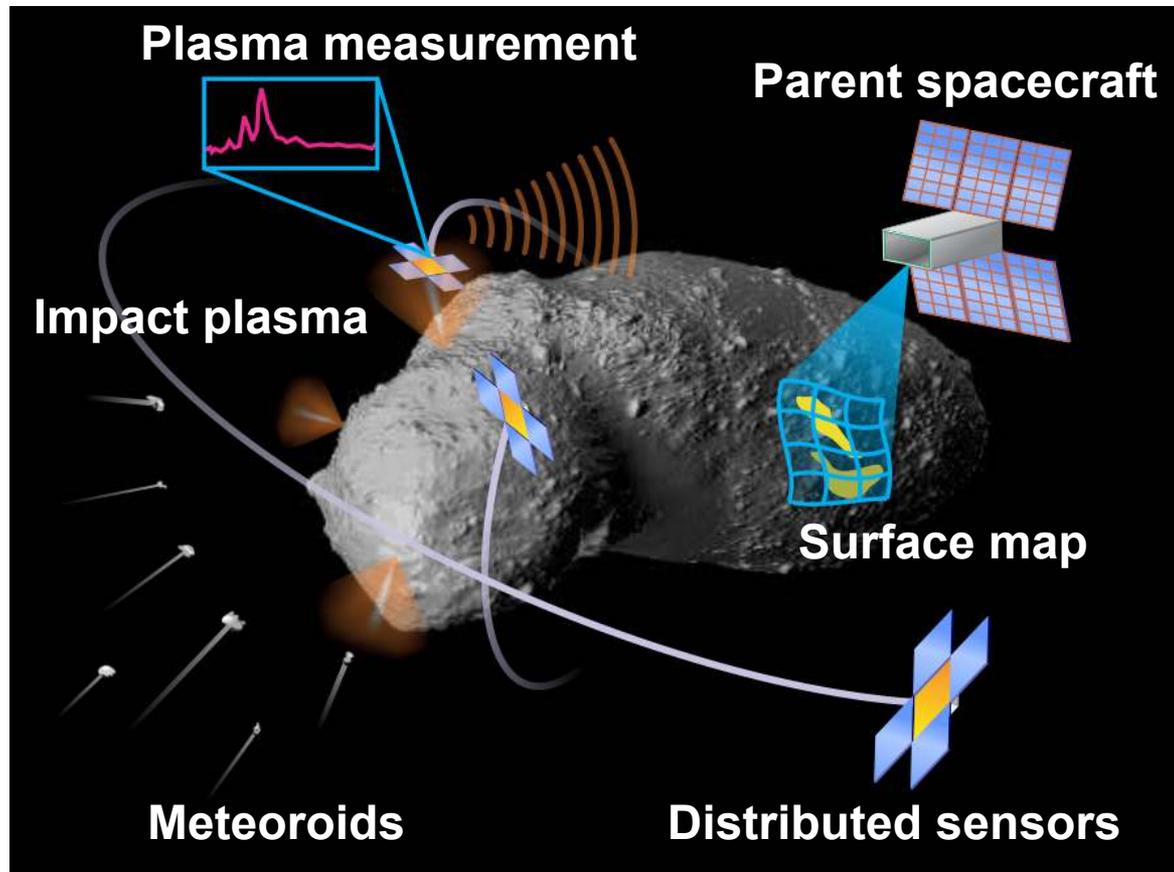
Motivation

- **Solar system science**
 - Meteoroid and dust dynamics
 - Geology and chemistry of primitive bodies
- **In situ resource characterization/utilization**
 - Candidate asteroid selection for retrieval
 - Target surface regions for exploitation



Concept overview

- Use natural meteoroid population to probe asteroid surface through impacts
- Measure impact plasma to characterize surface

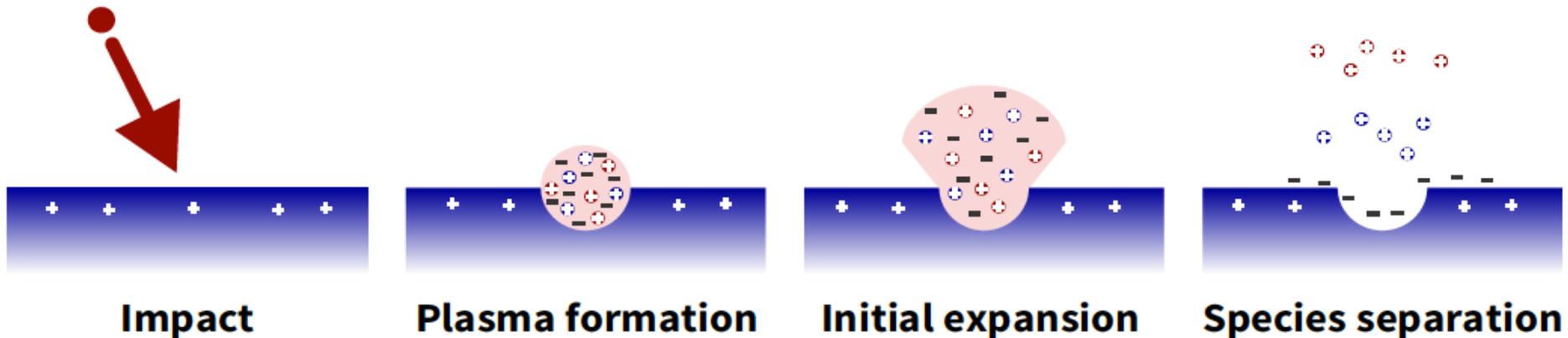


Concept overview

- **Meteoroid Impact Detection for Exploration of Asteroids: MIDEA**
- **Targeting 100–300 m asteroids**
- **Nanogram-sized meteoroid impactors**
- **Single small (6–12U) parent spacecraft to asteroid**
- **5–10 deployable free-flying sensors orbiting at 100–300 m altitude**

Hypervelocity impact

- Impact at speed greater than sound speed in the material
- Results in vaporization and ionization of the impactor and target

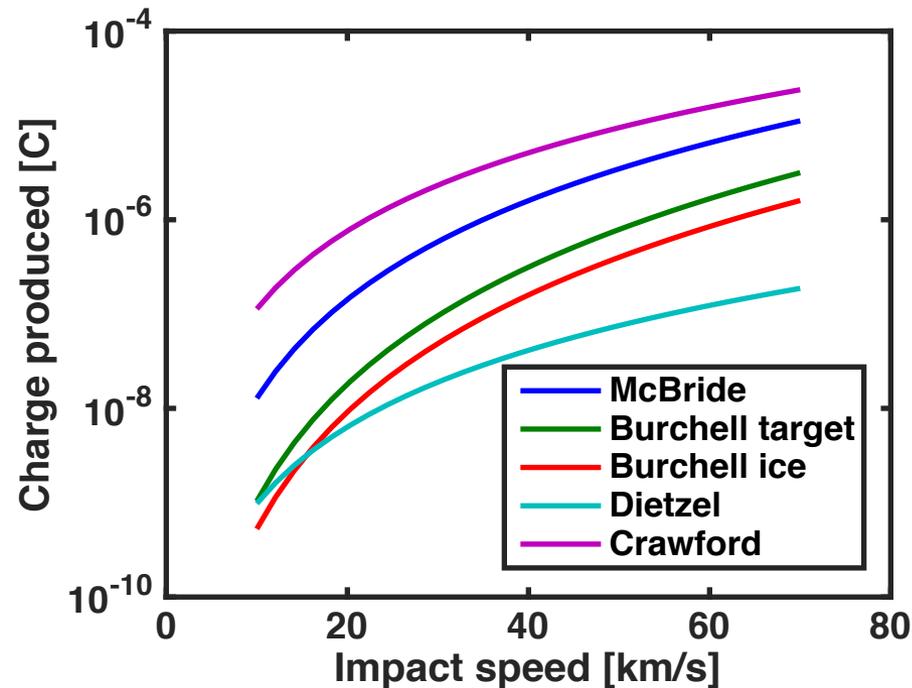


Impact ionization

- Charge production based on power law

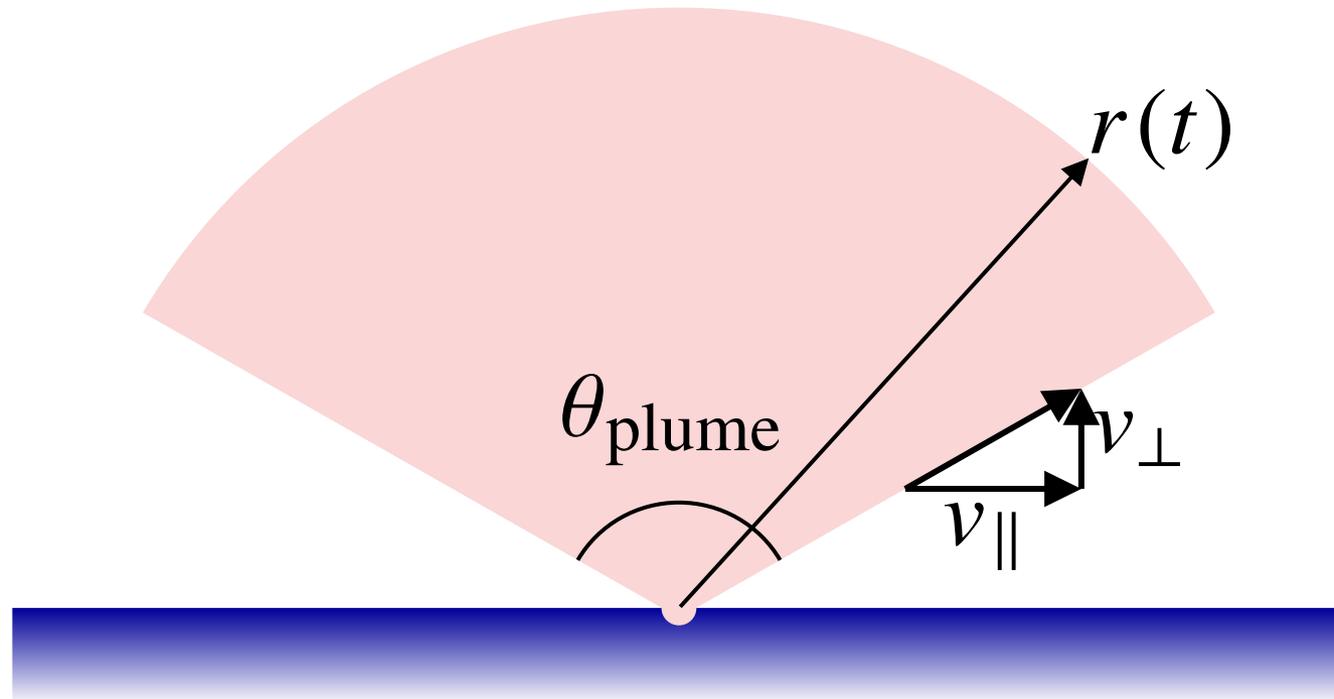
$$Q = C m^\alpha v^\beta$$

- Wide range of experimental values obtained for C and β



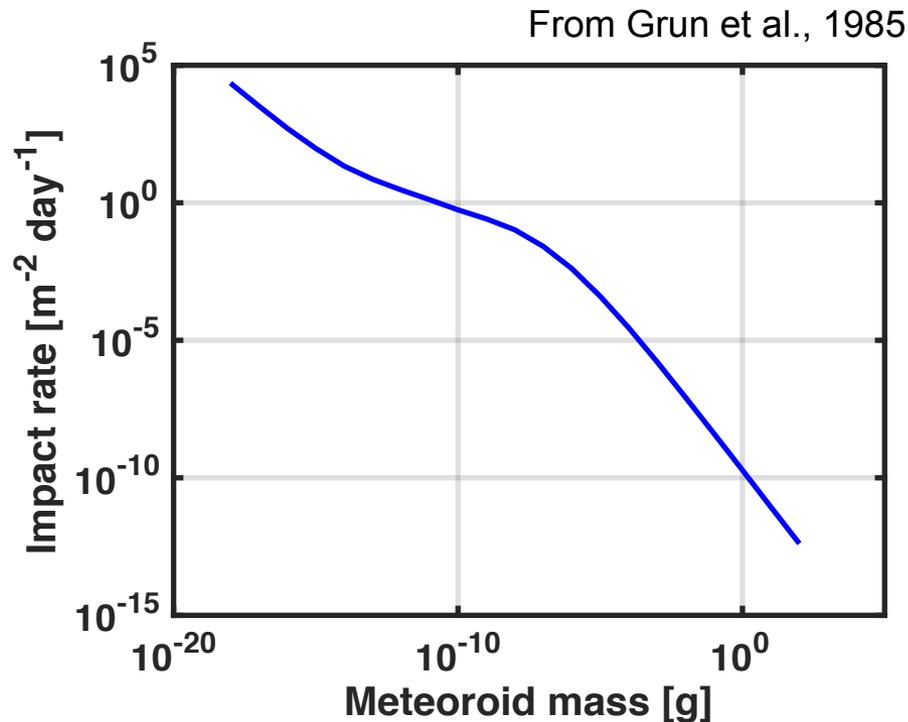
Plasma expansion

- Nearly hemispherical plume expanding at 20 km/s
- Density falls off with distance (inverse cubic)
- Expected detection threshold at about 300–500 m
- Determine composition through ion time-of-flight



Grun interplanetary flux model

- Applicable near 1 AU, 10^{-18} to 100 g
- Based on lunar craters, zodiacal light, in situ measurements



Meteoroid Engineering Model

- Based on meteor radar, calibrated to Grun model near 1 AU
- Applicable from 0.2 to 2 AU, 10^{-6} to 10 g
- Would like to compare with IMEM

Finished processing state vector 8 out of 7

Description  Meteoroid Engineering Model for use in determining the flux and speed of the background meteoroid environment for spacecraft in orbit about the Sun or on interplanetary transfer orbits. 

State Vector Input Filename: IR2_0_4\UserFiles\ItokawaHello2016sm.txt

Results Output Filename: IR2_0_4\UserFiles\ItokawaHello2016smOut.

Select a Spacecraft Orientation: Cube, ram faces velocity vector <default>

Enter the Limiting Mass (Log of Mass (g)): -6

Output Coordinate Frame:
 Body Fixed (default) J2K Ecliptic

Output Standard Deviation Files ? Yes No

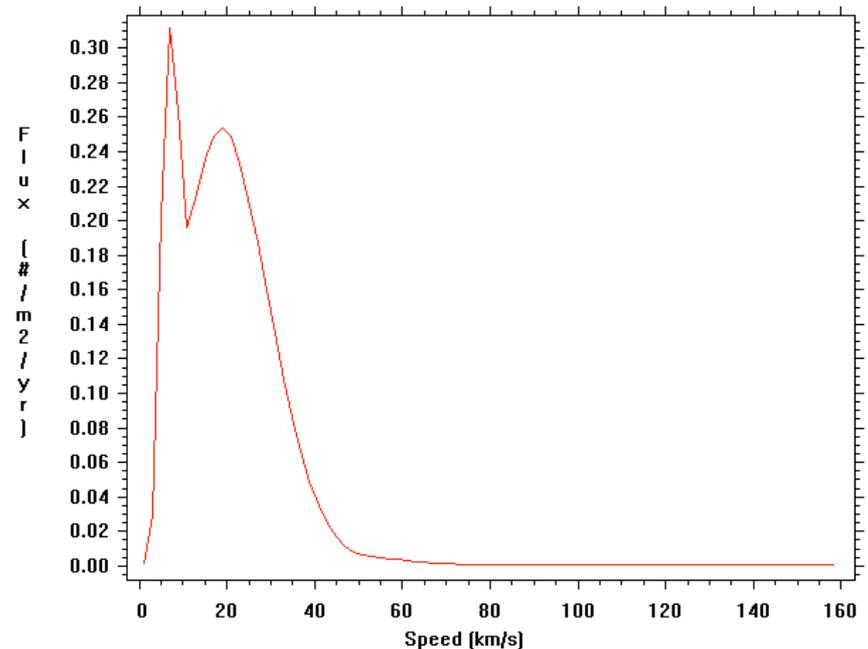
Output Intermediate Distribution Files ? Yes No

Run Type:
 Sequential Random

Enter the number of random draws from the state vector file: 7 * See Help for determining appropriate values

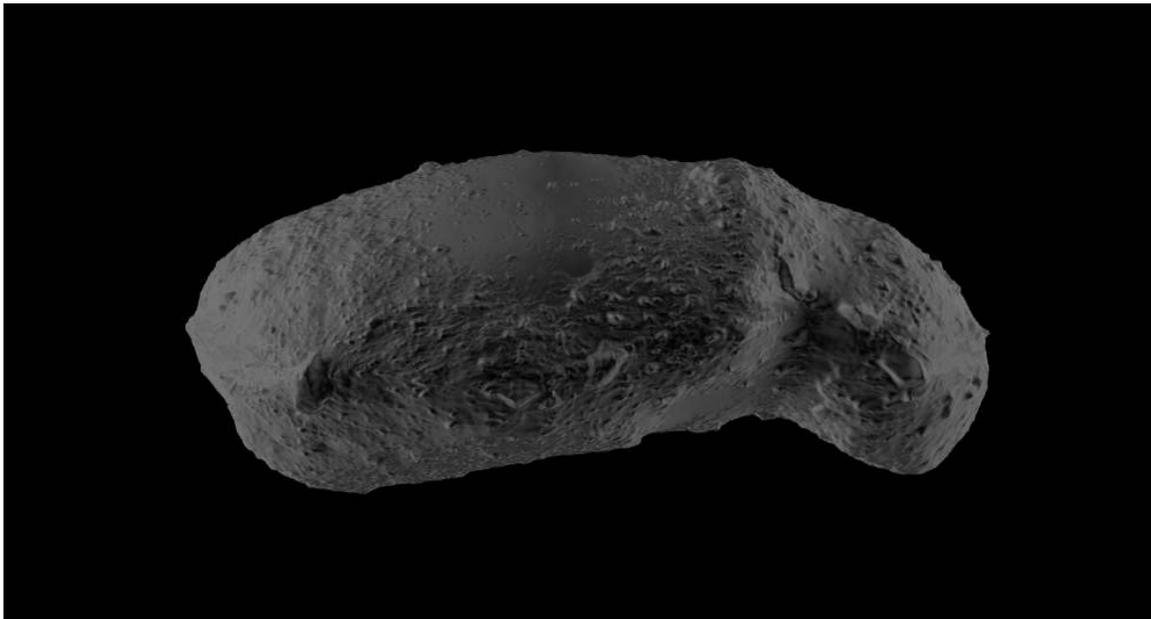
Output Resolution:
 1 deg x 1 deg x 2 km/s 4 deg x 4 deg x 4 km/s 5 deg x 5 deg x 5 km/s

Finished All Post Processing

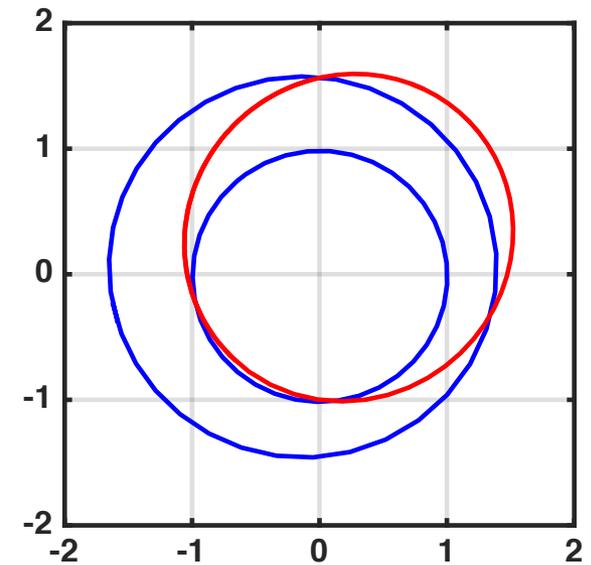


Reference asteroid: Itokawa

- **Discovered in 1998, visited by Hayabusa in 2005**
- **S-type asteroid, possibly contact binary**
- **535 x 294 x 209 m**
- **0.95 to 1.7 AU orbit**

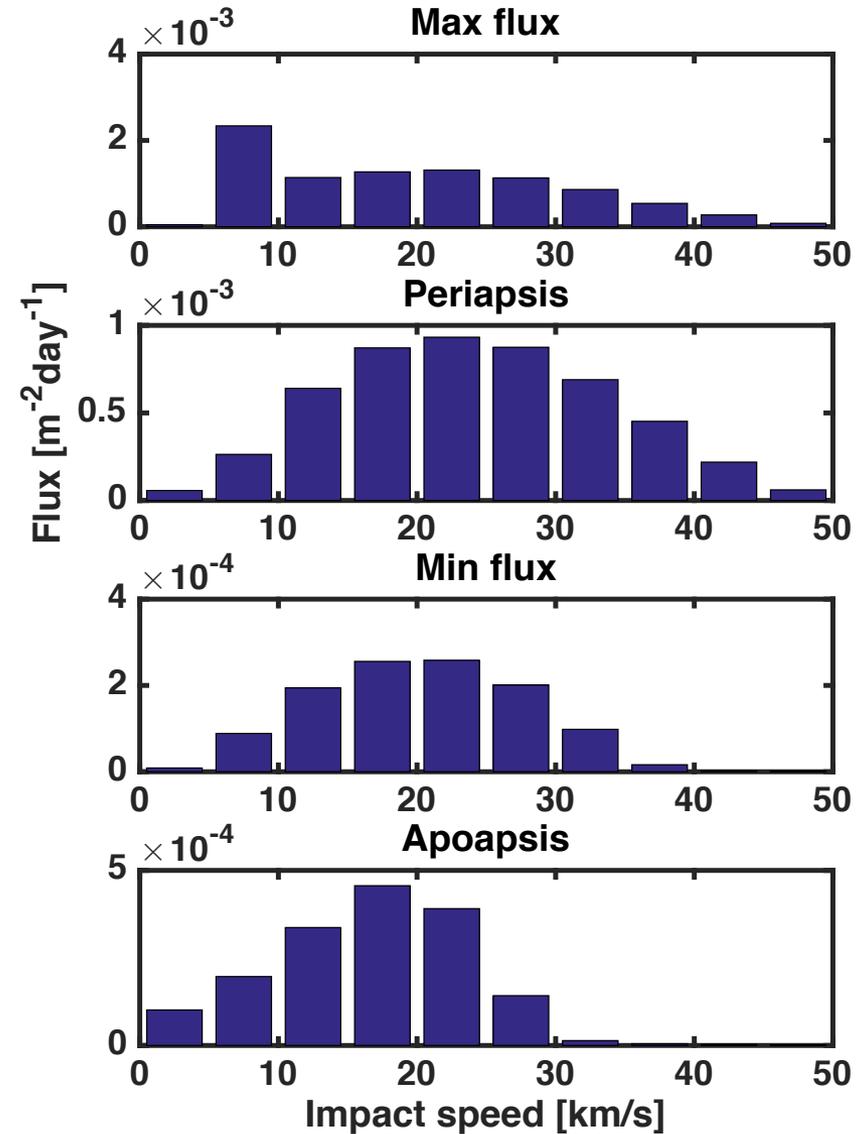
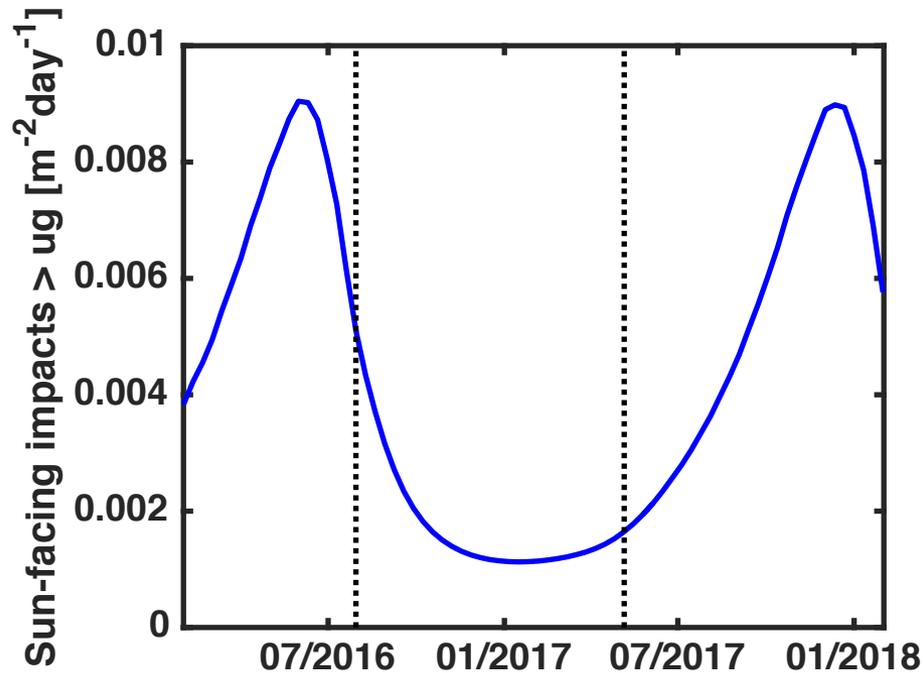


Generated from Gaskell Itokawa Shape Model



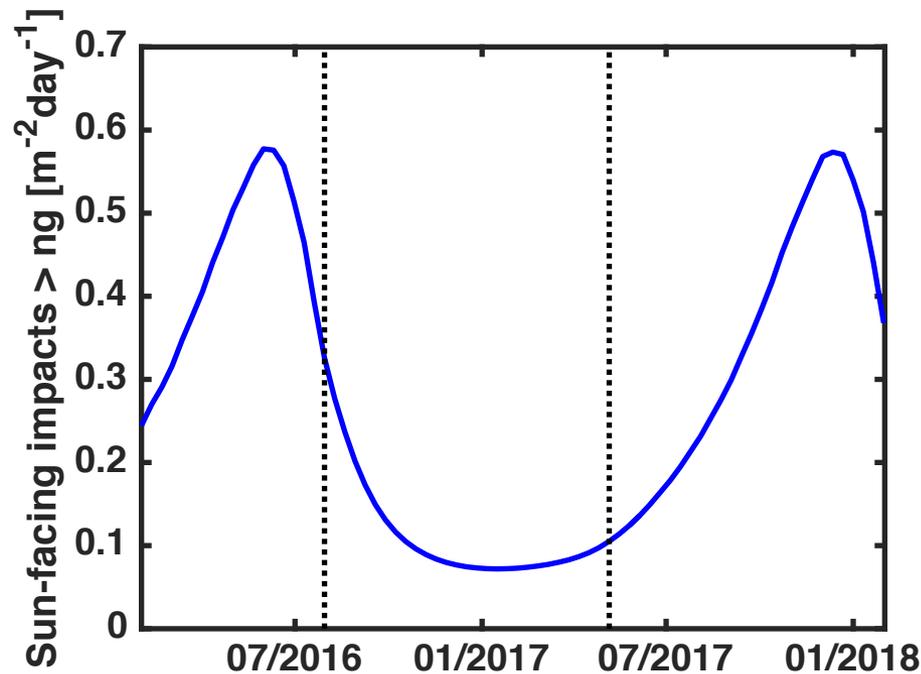
Itokawa meteoroid environment

- Sun-facing flux



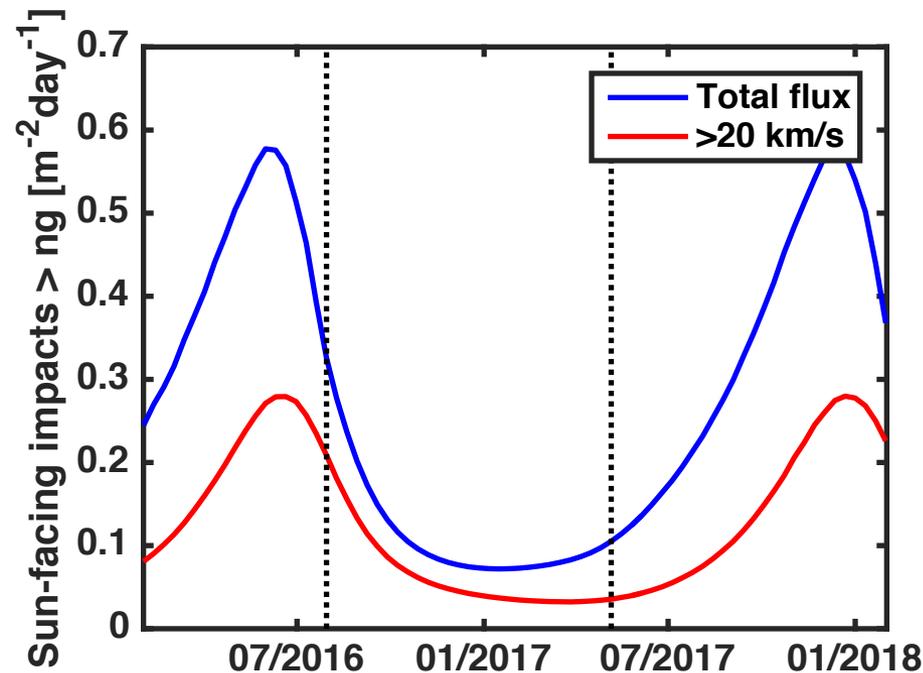
Itokawa meteoroid environment

- Sun-facing flux
- Scale from MEM microgram limit to nanogram (~5–10 μm) using Grun model



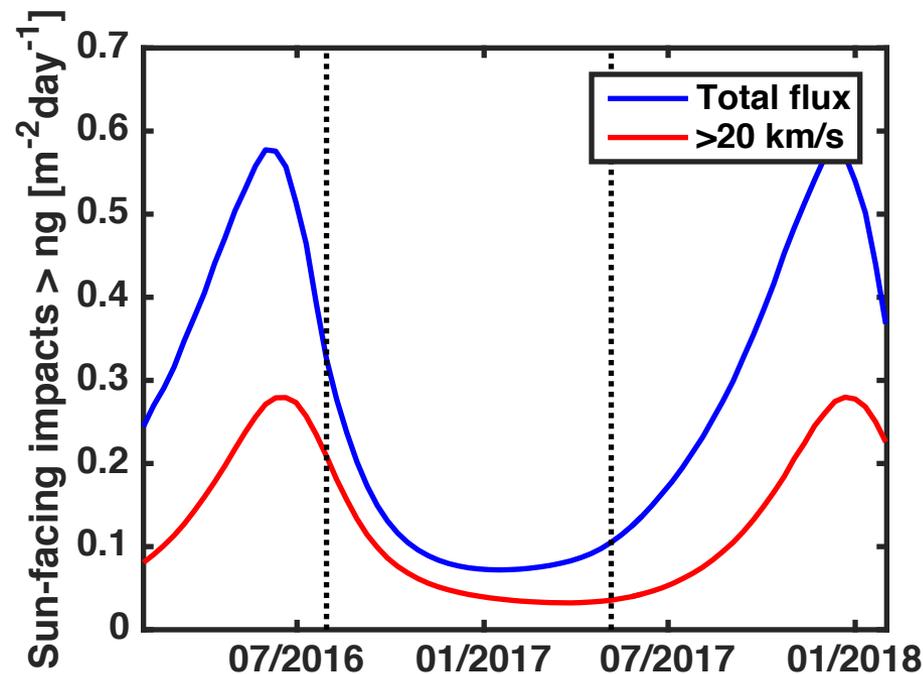
Itokawa meteoroid environment

- Sun-facing flux
- Scale from MEM microgram limit to nanogram ($\sim 5\text{--}10\ \mu\text{m}$) using Grun model
- Apply velocity threshold of 20 km/s



Mission parameters

- 0.1–0.3 impacts per square meter per day
- One impact detection every few seconds
- Every square meter covered in 3–10 days



Potential meteoroid research

- **Identification of new meteoroid streams intersecting asteroid path**
- **Determination of background flux**
- **Selection of asteroid targets corresponding to intersection with known meteoroid streams**

Conclusions

- Meteoroids, typically treated as a threat, can serve as useful probes into composition of asteroids
- Nanogram-sized meteoroids impact at reasonably frequent rate
- Impact plasma sensors provide a low-mass/cost alternative to spectroscopy for preliminary exploration

