Voyage 2050 White Paper

AMS-100
The Next Generation Magnetic Spectrometer in Space –
An International Science Platform for Physics and Astrophysics at
Lagrange Point 2

Mission type:
Large (L) mission

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1 Introduction

The past decade has witnessed a remarkable explosion of dramatic paradigm-challenging results in particle astrophysics. The most impactful of these results have emerged from a new generation of high-precision space-based instruments, such as PAMELA [1], AMS-02 [2], and Fermi-LAT [3]. These instruments – in particular, the magnetic spectrometers PAMELA and AMS-02 – have revealed unexpected features in the cosmic-ray matter [4, 5] and antimatter [6, 7] flux spectra that have challenged much of our traditional understanding of particle astrophysics, across a range of topics. These topics include questions of cosmic-ray origins, high-energy particle acceleration and propagation mechanisms, and the nature of dark matter. Perhaps the most intriguing result has been the recent observation of a few candidate cosmic-ray antihelium events [8], which, if confirmed, would have profound implications for our understanding of the matter-antimatter asymmetry of the universe.

Addressing these topics, and extending current measurements to higher energies, requires powerful new instrumentation with extended exposure times. And while pure calorimeters, such as the presently deployed DAMPE [9] and CALET [10] can provide some answers, only magnetic spectrometers, with their ability to differentiate charge sign, and even measure particle mass, can properly address the full scope of open questions.

The project described in this white paper is A next-generation Magnetic Spectrometer with a geometrical acceptance of 100 m$^2$ sr [11] (referred to as AMS-100 throughout, Fig. 1). It is conceived to be an international platform for precision particle astrophysics and fundamental physics, at energy scales well beyond the reach of all current detector payloads. Achieving this will require overcoming a number of key technical challenges, though in many cases, technology already exists that can be adapted to provide the needed performance.

The heart of the instrument is a thin, large-volume high-temperature superconducting (HTS) solenoid magnet, which provides a uniform 1 Tesla field within the tracking volume. It is cooled passively to 50 K to 60 K. An expandable compensation coil balances the magnetic moment of the solenoid and allows the attitude control of the instrument within the heliospheric magnetic field. When instrumented with proven silicon-strip and scintillating fiber technologies, the spectrometer will achieve a maximum detectable rigidity (MDR, defined by $\Delta R/R = 1$, where $\Delta R$ is the uncertainty of the rigidity measurement) of 100 TV, with an effective acceptance of 100 m$^2$ sr. A deep ($70\ X_0/4\ \lambda_f$) central calorimeter will provide energy measurements and particle identification, with innovative ‘cubic’ detector elements enabling a large acceptance of 30 m$^2$ sr.

Combined, this instrumentation will vastly improve on existing measurements and allow us to probe, with high statistics and high precision, the positron and electron spectra to 10 TeV, the antiproton spectrum to 10 TV, and the nuclear cosmic-ray component to $10^{16}$ eV, past the cosmic-ray knee.

For the first time, this instrument will have the acceptance and resolution to probe the cosmic-ray antideuteron spectrum up to 8 GeV/n with high precision. AMS-100 will vastly expand our sensitivity to heavy cosmic antimatter ($Z \leq -2$). Covering most of the sky continuously, AMS-100 will provide high-resolution survey measurements of $\gamma$-rays to energies beyond the TeV scale, with an angular resolution of 4$''$ at 1 TeV and 0$''$.4 at 10 TeV, comparable to X-ray telescopes [12].

The magnet and detector systems will be designed with no consumables, allowing for an extended 10-year payload lifetime at its thermally-favorable orbital location at the Sun-
Earth Lagrange Point 2 (L2). A plausible timeline for instrument definition, design, development, and testing would target a launch date in 2039, though this requires an early commitment from the agencies and the community to perform the necessary R&D tasks. This will include some level of underlying technology development, as well as a pathfinder mission to test the high temperature superconducting solenoid magnet system at L2.

2 Key Science Goals and Objectives

AMS-100 is conceived as a wide-focus platform for precision high energy astrophysics and physics across a broad range of topics. The size, variety, and sensitivity of its instrumentation are chosen to allow it to address key open questions in particle astrophysics, make important contributions to multi-messenger astrophysics, and open a large window of discovery space. In addition to the topics discussed in more detail below, the science possible with AMS-100 includes the measurements of various isotopes and heavy (trans-iron) nuclei in cosmic rays, strangelets [13], magnetic monopoles [14], particles with fractional charges [15], evaporating primordial black holes [16, 17], search for signatures of dark matter annihilation or decay in γ-ray lines [18, 19], search for axions [20, 21], and tests of quantum gravity by precisely measuring the energy and arrival time of photons from γ-ray bursts [22], to mention a few examples that can be covered in principle with unprecedented sensitivity by such a powerful instrument. In the following, detector acceptances have been calculated using GEANT4 [23].
Figure 2: Cosmic-ray proton spectrum. Expected data from AMS-100 (blue) (statistical uncertainties only), for the case that the proton flux is described by a power law with several smooth breaks, inserted for the purpose of illustration (dashed curve). Recent magnetic spectrometer measurements from BESS [29], PAMELA [30], and AMS-02 [24]. Recent calorimeter measurements from ATIC-2 [31], CALET [32], and CREAM-III [33].

2.1 Protons and other nuclei

Protons are the most abundant species in cosmic rays. PAMELA and AMS-02 have reported a spectral break above $\sim 200$ GV in protons and other light nuclei [24–26]. Such spectral breaks encode information about the sources and propagation history of cosmic rays [27, 28], but thus far, no coherent and well-accepted description of the various observed features has emerged. AMS-100 will have the size and energy reach to directly measure, for the first time ever, protons and light nuclei in cosmic rays up to and through the energy of the cosmic-ray knee (Fig. 2), as well as heavier nuclei with vastly improved statistics. The detailed spectral and composition studies enabled by AMS-100 through the knee region will directly address several unresolved decades-old questions in cosmic-ray astrophysics including, for instance, what is the maximum energy that can be reached by galactic cosmic-ray accelerators. This information also forms an essential context for the other studies detailed below, such as the origin of cosmic-ray positrons, electrons, antiprotons, and antimatter. In addition, these direct measurements at the highest energies will allow us to investigate the change of the chemical composition of cosmic rays at the knee and gather invaluable information about the transition from galactic to extragalactic cosmic rays.

2.2 Positrons and Electrons

The unexpected excess of high-energy positrons observed above the predicted yield from cosmic-ray collisions has been one of the most exciting developments in high-energy astrophysics in the last generation. Possible interpretations range from new effects in the acceleration and propagation of cosmic rays [34–36] to acceleration of positrons to high energies in astrophysical objects [37–45] and to dark matter [46–54] as a new source of
Figure 3: Cosmic-ray positron spectrum. Expected data from AMS-100 (stat. uncertainties only) for two different scenarios: a) The spectrum is described by a power law plus a source term with an exponential cutoff (blue circles, lower curve at high energy). b) The spectrum is described by power laws with spectral breaks and the last break is at 300 GeV (blue squares, upper curve at high energy). The dashed green curve shows the expected spectrum from a) without the source term. Recent experimental data from PAMELA [55] and AMS-02 [7] are shown.

cosmic-ray positrons. The latest data on the positron flux from AMS-02 show a spectral break at 300 GeV followed by a sharp drop [7]. The detailed understanding of the shape of the spectrum above this energy is the key to deduce the origin of these high energy positrons.

A generic source term, that describes the contribution of the new source responsible for the positron excess, is given by a power law with an exponential cutoff (e.g., Ref. [7]). AMS-100 will be able to precisely measure the cosmic-ray positron spectrum up to 10 TeV (Fig. 3).

If the origin of the source term is a process producing electrons and positrons in equal amounts, the effect should also be detectable in the cosmic-ray electron spectrum. Both pulsar models and dark matter models generically predict such a charge-symmetric source term. H.E.S.S. [60], VERITAS [62], and DAMPE [58] have observed a spectral break of the combined electron and positron flux at about 1 TeV followed by a sharp drop, which might be related to this question. AMS-100 will be able to precisely measure the cosmic-ray electron spectrum up to 20 TeV (Fig. 4) and detect features associated with the local sources of electrons predicted in propagation models. Identifying such features will shed light on the origin of positrons, electrons, and other cosmic-ray species.

2.3 Antiprotons

A key discriminator between various models for the origin of positrons is the presence or lack of antiprotons. Pulsars will not generate these antiparticles, but according to most
Figure 4: Cosmic-ray electron spectrum. Expected data from AMS-100 in blue (stat. uncertainties only) for two different scenarios: a) Broken power law plus a charge symmetric source term as obtained a fit to the positron flux (blue circles, lower curve at high energy). b) The broken power law continues without any further spectral break at high energies (blue squares, upper curve at high energy). The dashed green curve shows the derived spectrum from a) without the source term. Recent experimental data from PAMELA [56] and AMS-02 [5] are shown. For comparison also the recent calorimetric measurements of the combined \((e^+ + e^-)\) flux by CALET [57], DAMPE [58], FERMI [59], and H.E.S.S. [60, 61] are shown as they extend to higher energies and provide an upper limit for the electron flux.

dark matter models, pair annihilations of dark matter particles will create antiprotons. Therefore, antiproton measurements may provide support to the dark matter hypothesis for the origin of the positron excess or rule it out. Independently, they provide another crucial probe of the processes in the interstellar medium, as well as production and acceleration of secondary species in the sources [63]. AMS-100 will be able to measure the antiproton spectrum up to the 10 TeV energy scale and provide precise information on the spectral shape. Hence it will shed light on many questions associated with the origin of cosmic rays and with the nature of dark matter (Fig. 5).

2.4 Antihelium

AMS-02 has shown both \(^3\text{He}\) and \(^4\text{He}\) candidate events at a CERN Colloquium in 2018 [8]. These unexpected events are observed in AMS-02 at a rate of 1 event/year or 1 \(^4\text{He}\) event in 100 million \(^4\text{He}\) events, well above the rate of secondary \(^4\text{He}\) production in coalescence models. Their origin is presently unclear, however, the statistically significant detection of unambiguous \(^4\text{He}\) events could have the most profound implications for physics and astrophysics. Progress in this direction requires new instrumentation achieving high-confidence particle identification, a solid understanding of potential backgrounds, and powerful systematic checks. AMS-100 will have the performance and exposure to do exactly this –
extrapolating the AMS-02 candidate $^4\text{He}$ event rate to the AMS-100 acceptance results in the prediction of $\sim 1000$ $^4\text{He}$ events/year. The precision measurement of the spectral shape of the $^4\text{He}$ flux would allow tests of the origin of $^4\text{He}$. Additionally, the rotational symmetry of AMS-100 allows detailed systematic cross-checks of such a result equivalent to inverting the magnetic field.

### 2.5 Antideuterons

Antideuterons potentially are the most sensitive probe for dark matter in cosmic rays [71, 72]. While antiprotons are predominantly produced in secondary interactions in the interstellar medium, antideuterons at low energy have no other known origin. No antideuterons have ever been identified in cosmic rays. The current best limit has been set by BESS [73], excluding a flux of $1.9 \times 10^{-4}$ (m$^2$ s sr GeV/n)$^{-1}$ between 0.17 GeV/n and 1.15 GeV/n at the 95% confidence level. The expected sensitivity of AMS-100 is $3 \times 10^{-11}$ (m$^2$ s sr GeV/n)$^{-1}$ in the energy range between 0.1 GeV/n and 8 GeV/n. It is compared to other experiments...
Figure 6: Differential sensitivity of AMS-100 to antideuterons in cosmic rays for a mission time of 10 years and a single-layer ToF time resolution of 20 ps, with a logarithmic binning of 20 bins per decade (thick dashed red curve). Sensitivities for time resolutions of 10 ps, 30 ps, 40 ps and 50 ps are shown by thin dashed red curves (from right to left). The red symbols show the expected data from AMS-100 if the antideuteron flux follows the dark matter model of Ref. [66] with statistical uncertainties (which are smaller than the symbol size). The solid black curve shows the antideuteron flux expected from secondary production by charged cosmic rays interacting with the interstellar material, as derived in Ref. [67] for the EPOS LHC interaction model. Data for the other $Z = -1$ particles in cosmic rays, from AMS-02 [6, 68] and BESS-Polar [64], are shown to indicate the signal to background ratios for the antideuteron measurement.

in Table 1. At this level of sensitivity, it is no longer useful to quote an integral sensitivity, which is related to the chances of observing a certain number of events anywhere inside a given energy range. Instead, we calculate a differential sensitivity, which can be directly compared to model predictions for the differential $\bar{D}$ flux. We choose a logarithmic energy binning with 20 bins per decade and calculate the sensitivity individually for each bin. It is defined as the 95% confidence level limit that will be set in case no $\bar{D}$ events are observed in the given bin. The differential sensitivity for antideuterons is shown in Fig. 6. AMS-100 will be the first instrument to measure the cosmic-ray antideuteron spectrum with thousands of events, even in the case that antideuterons originate only from secondary production. AMS-100 will have the sensitivity to distinguish between antideuterons originating in dark matter annihilations and those produced in interactions within the interstellar medium, due to the different spectral shapes expected for these components. While it is not clear if antideuterons from dark matter annihilation exist, the observation of antideuterons from secondary production would allow us to set additional constraints on the $^3\text{He}$ and $^4\text{He}$ rates in cosmic rays: Within the coalescence model [74], every nucleon in the antimatter particle reduces the production rate by a factor $\simeq 10^3$-10$^4$ depending on the energy, i.e. we expect $N(\bar{p}) : N(\bar{D}) : N(\bar{^3}\text{He}) : N(\bar{^4}\text{He}) \approx 1 : 10^{-3} : 10^{-4} : 10^{-6} : 10^{-7} : 10^{-9} : 10^{-10}$ in cosmic rays if there is no new source for one of these antimatter species. A simultaneous
Figure 7: Left: Sketch of a $\gamma$ conversion in the AMS-100 main solenoid and of a $\gamma$ reconstructed in the calorimeter. Right: Expected angular resolution (68% containment) for photons converted in the AMS-100 barrel, based on a Geant4 simulation of the multiple scattering in the detector material, and for the endcap $\gamma$ converters (Fig. 12), which follow the design of GAMMA-400 [70]. The resolution function of Fermi-LAT [74] is shown for comparison.

measurement of these sensitive probes for new physics is therefore required to derive a coherent picture.

2.6 High-energy Gamma-Rays

Building on the success of current-generation $\gamma$-ray detectors such as Fermi-LAT [75], AMS-100 will allow detailed studies of $\gamma$-ray sources and the diffuse $\gamma$-ray emission up to the $\simeq 10$ TeV scale. It has an acceptance of 30 m$^2$ sr for photons reconstructed in the calorimeter system. Due to a dedicated pre-shower detector enveloping the central calorimeter, the expected angular resolution is compatible to the one of Fermi-LAT. In addition, a similar acceptance is obtained from photon conversions in the thin main solenoid, resulting in a total acceptance for diffuse photons of up to 60 m$^2$ sr.

At low energies the angular resolution for converted photons is limited by multiple scattering of the resulting electron-positron pairs. But at high energies, the direction of the photon can be reconstructed with high accuracy due to the good spatial resolution and long lever arm of the main device for particle tracking, the silicon tracker (Fig. 7). This will resolve structures in $\gamma$-ray sources with angular resolution similar to today’s best X-ray telescopes. Interesting targets include galactic supernova remnants [78, 79], pulsar wind nebulae [80], and blazars. For converted photons perpendicular to the $z$-axis the effective area reaches 2.5 m$^2$.

Due to the rotational symmetry of its barrel detector, the presence of dedicated $\gamma$ converters in the endcap opposite the service module, and its location far from the shadow of the Earth, AMS-100 will be able to monitor almost the entire sky continuously. Combined with its large effective area, this will make it a prime instrument for instantaneous observation of transient sources, e.g. $\gamma$-ray bursts or photons emitted in conjunction with
gravitational wave events, as well as for monitoring blazar variability [81]. In combination with ground-based experiments, it will allow completing the multi-messenger network for modern astronomy combining the observation of gravitational waves, cosmic-ray neutrinos and GeV-TeV γ-rays. AMS-100 can serve as a trigger for the Cherenkov Telescope Array [82] and similar ground-based observatories for the detailed follow-up investigation of transient sources.

The physics program of AMS-100 will cover a large variety of galactic and extragalactic γ-ray sources. One example is the study of γ-ray pair halos around blazars, e.g. [83]. TeV γ-rays emitted from the jets of blazars produce pairs of electrons and positrons through interactions with the extragalactic background light (EBL). These electrons and positrons further lose their energy through synchrotron and inverse Compton emission, thus initiating a cascade of lower-energy electrons, positrons and γ-rays. Depending on the properties of the intergalactic magnetic fields, γ-rays from such cascades can be observed in the form of extended γ-ray halos. With its improved diffuse sensitivity, AMS-100 would be able to detect or constrain the existence of pair halos and thus put new bounds on the strength and correlation length of the intergalactic magnetic field.

One can also search for spectral features in the γ-ray emission of blazars due to attenuation from the EBL. This allows drawing conclusions on axion-photon couplings [20, 21]. Measuring blazar spectra to higher energies with AMS-100 extends the sensitive parameter space to smaller couplings.

The excellent timing and pointing capabilities of AMS-100 make it an ideal instrument to test Lorentz invariance violation (LIV) by precisely measuring the energy and arrival time of photons from γ-ray bursts [22]. Deviations of the group velocity of photons from the speed of light, which could be realised in models of quantum gravity, would manifest themselves in different arrival times of photons of different energies from the same transient event. Given the energy reach of AMS-100, the observation of very high-energy γ-rays in conjunction with X-ray instruments would increase the sensitivity to LIV by orders of magnitude compared to existing measurements.

3 Technical Overview and Technology Drivers

Achieving the science goals described above requires an ambitious detector design to dramatically improve on the current state-of-the-art in sensitivity, precision, and energy reach.

3.1 Design considerations

Due to the strong dependence of the cosmic-ray flux Φ on energy E, approximated by Φ ∝ E⁻³, every increase in energy reach by a factor of 10 requires an increase in geometrical acceptance by a factor of 1000. Simply scaling the dimensions for the telescope-like geometries of PAMELA or AMS-02 would not allow significantly enhancing the geometrical acceptance and the energy reach at the same time. Increasing the height would enhance the energy reach but reduce the acceptance. Increasing the diameter would enhance the acceptance but reduce the magnetic field for a fixed magnet wall thickness and hence the energy reach. This dilemma can only be overcome by moving to a different detector geometry. A possible solution has been pioneered successfully by the BESS experiment [84] with a thin solenoid. The key here is the fact that the central magnetic field for a long
solenoid only depends on the number of turns, the current and the length, but not on the
radius. Therefore, for a solenoid of given wall thickness and instrumented with a tracking
detector on the inside like a classical collider experiment, both the geometrical acceptance
and the MDR increase quadratically with the radius if the diameter-to-length ratio stays
constant. If placed far away from Earth, such an instrument has an angular acceptance of
up to $4\pi$ sr due to its rotational symmetry, superior to any telescope-like geometry.

3.2 Magnet Design

The geometrical acceptance of 100 m$^2$ sr defines the dimensions of the 3 mm thin main
solenoid. It has a length of 6 m and a diameter of 4 m and creates a central magnetic field
of 1 Tesla along the z-axis. As the magnet will be operated at 50 K to 60 K, the only option
is to construct it from second-generation rare-earth barium copper oxide (REBCO) high
temperature superconducting tapes [85, 86]. These HTS tapes have a typical thickness
of $\sim 0.1$ mm and can carry high current densities even at field strengths of 30 T [87] and
tolerate severe mechanical stresses [88, 89] thanks to a 30 $\mu$m to 100 $\mu$m thick Hastelloy
substrate. Recently, REBCO wire has become available in extended lengths [90]. A typically
20 $\mu$m thick copper stabilizer completes the HTS tapes (for more details, see for example
Refs. [91–93]) which can be easily soldered for joints. It has been shown in Ref. [94] that
increasing the stabilizer thickness aids in reducing the magnet temperature at a quench.
For the AMS-100 magnets, we assume that the copper stabilizer will be replaced by an
equivalent aluminum stabilizer to minimize the material budget [95].

Quench protection and understanding the dynamics of the quench process in HTS
tapes [96] are the key for the long term stable operation of such a magnet in space. As one possible option, HTS coils can be protected from an irreversible quench by winding
them from tapes without additional insulation [97–99], thus allowing the current to flow
in the radial direction in case of a thermal runaway.

Generally REBCO tapes are available in piece lengths of 300 m to 500 m with joint
resistances of less than 20 n$\Omega$ [100]. For a 450 m long REBCO tape at $T = 50$ K and a
magnetic field of 1 T, a critical current of $I_c = 1000$ A/cm-wide, equivalent to $I_c = 1200$ A
for 1.2 cm wide tape, has been reported in 2019 [90].

The key parameters of the magnet system for AMS-100 are given in Table 2. Progress on
the critical current $I_c$ for REBCO tapes, as expected in the coming years, will proportionally
reduce the number of layers required to obtain a central magnetic field of 1 T and will hence
allow reducing the weight and the material budget of the coils even further. The magnetic
field is visualized in Fig. 8.

The thin solenoid is cooled by radiation to deep space and operated in thermal equilibrium
at a temperature of 50 K to 60 K behind a sunshield. A simplified thermal model taking only
radiation into account is shown in Fig. 9. The main solenoid is insulated thermally from the
other detector components by multi-layer insulation. The obtained magnet temperatures
leave some margin for conductive thermal loads which have to be taken into account in
the final thermo-mechanical design. Similar to all other detectors inside the main solenoid,
the silicon tracker temperature will be kept constant at 200 K using a two-phase cooling
system or heat pipes connected to the radiator opposite the sunshield. This temperature
of 200 K might have to be adjusted within the overall thermo-mechanical model to ensure
a stable operating temperature for the main solenoid of 50 K to 60 K. All sub-detector
systems are designed to have a better signal-to-noise ratio at such low temperatures than at
Main Compensation solenoid  Compensation coil

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Main solenoid</th>
<th>Compensation coil</th>
</tr>
</thead>
<tbody>
<tr>
<td>Inner radius</td>
<td>2.0 m</td>
<td>6.0 m</td>
</tr>
<tr>
<td>Length</td>
<td>6.0 m</td>
<td>1.2 m</td>
</tr>
<tr>
<td>Current</td>
<td>500 A</td>
<td>1500 A</td>
</tr>
<tr>
<td>Temperature</td>
<td>50 - 60 K</td>
<td>30 - 40 K</td>
</tr>
<tr>
<td>HTS tape width</td>
<td>12 mm</td>
<td>12 mm</td>
</tr>
<tr>
<td>HTS tape layers</td>
<td>22</td>
<td>4</td>
</tr>
<tr>
<td>$B_z$ at center</td>
<td>1.0 T</td>
<td>-0.06 T</td>
</tr>
<tr>
<td>Stored energy</td>
<td>37 MJ</td>
<td>4.5 MJ</td>
</tr>
<tr>
<td>Magnetic moment</td>
<td>70 MA m$^2$</td>
<td>-70 MA m$^2$</td>
</tr>
<tr>
<td>Coil thickness</td>
<td>3.0 mm</td>
<td>0.5 mm</td>
</tr>
<tr>
<td>Mass</td>
<td>1.2 t</td>
<td>0.13 t</td>
</tr>
<tr>
<td>Volume</td>
<td>75 m$^3$</td>
<td>136 m$^3$</td>
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<tr>
<td>Material budget</td>
<td>0.12 $X_0$</td>
<td>0.02 $X_0$</td>
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<tr>
<td>Wire length</td>
<td>0.012 $\lambda_I$</td>
<td>0.002 $\lambda_I$</td>
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<td>Hoop stress $\sigma_\theta$</td>
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<td>250 kPa</td>
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<tr>
<td>$\sigma_R$</td>
<td>-130 kPa</td>
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<td>$\sigma_Z$</td>
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<td>-79 kPa</td>
</tr>
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</table>

Table 2: Main parameters of the AMS-100 magnet system. The mechanical stresses are denoted by $\sigma$ and calculated according to the formulae given by Iwasa [102].

room temperature and first laboratory tests of various detector components down to liquid nitrogen temperatures have already been performed successfully at RWTH Aachen [101].

Particularly for the sensitivity to antimatter in cosmic rays, the small wall thickness of the main solenoid and its support structure are of key importance. One option for this that we have studied in more detail consists of two lightweight aluminum honeycomb structures with a height of 10 mm each that sandwich the magnet and have carbon fiber face sheets on the outside (Fig. 10 (left)). The coil would be assembled on a temporary support and afterwards the outer honeycomb and carbon fiber face sheets would be laminated directly onto the outer side of the magnet. In the next step, the temporary inner support would be removed and the inner honeycomb and carbon fiber face sheets would be laminated. The total material budget of this structure would be equivalent to a solid aluminum cylinder of 3 mm thickness ($0.04 \ X_0$). The further optimisation of this lightweight magnet support structure will have to take all components of the instrument and the constraints from the thermal model into account.

It has never been demonstrated that a HTS magnet with a lightweight support structure can be operated in space. In particular, the vibrations during the rocket launch are a concern. We have therefore started to produce first prototypes (Fig. 10 (right)) of thin HTS pancakes to performe space qualification tests including vibration and thermo-vacuum tests.

For the operation of a large solenoid in deep space, the interaction with the interplanetary magnetic field (IMF) is a major concern. The IMF has a complicated time-dependent structure. Due to the rotation of the Sun (period of 25.4 days), its magnetic field winds up into a large rotating spiral. The heliospheric magnetic field changes polarity every $\approx 11$
Figure 8: Magnetic field lines in the AMS-100 magnet system (black) and amplitude of the \( z \)-component of the magnetic field (color map). The compensation coil cancels the magnetic moment of the main solenoid, without substantially affecting the magnetic field inside the main solenoid.

Figure 9: Simplified thermal model for AMS-100 taking only the radiation between the surfaces, the Sun and deep space into account. The color scale indicates temperature in Kelvin. Left: Warm side facing the Sun. Right: Cold side facing deep space.
Figure 10: Left: Photograph of a structural test article for the lightweight support of the AMS-100 main solenoid. The central layer is mechanically equivalent to the expected magnet. Right: Photograph of a 20 layer HTS test pancake with a diameter of 80 mm.

Figure 11: Properties of the solar magnetic field based on recent measurements by the ACE/MAG instrument [103] at Lagrange Point 1 (L1) and resulting angular momentum accumulated by the main solenoid of AMS-100 without a compensation coil. From top to bottom: $|\mathbf{B}|$, norm of the interplanetary magnetic field; $B_t$, its transverse component relative to the line between the Sun and L1; $\theta$, the angle between the magnetic field vector and this line. The horizontal blue lines mark the values of $\theta$ calculated for a simple Parker spiral field geometry for the two heliospheric polarities; $\sum L_t$, angular momentum accumulated by the main solenoid in transverse direction.

years [104]. It is distorted at the orbit of AMS-100 around L2 by the geomagnetic field and by solar flares. Due to the solar wind, the magnetic field at L2 still has an average strength of 6 nT, varying between 0 and 35 nT. In combination with the large magnetic moment of the AMS-100 main solenoid, this causes an average torque of 0.4 N m. Based on measurements of the heliospheric magnetic field at Lagrange Point 1, which is very close to L2 on heliospheric scales, we can derive the expected angular momentum as a function of time (Fig. 11). Even though the magnetic field reverses polarity periodically, the accumulated angular momentum reaches a value on the order of $10^6$ N m s over the
course of one year. Such a large angular momentum cannot be balanced by reaction wheels or control moment gyroscopes. Instead, a compensation coil is needed with opposite field direction to balance the magnetic dipole moment of the main solenoid (Fig. 8).

With a diameter of 12 m, the compensation coil has to be an expandable coil, as it has been studied for radiation shielding in space in [105]. It will consist of 0.5 mm of HTS tape embedded and reinforced by 1 mm thick Kevlar or Zylon layers. The support structure of this coil is designed to avoid small bending radii for the HTS tape. The Lorentz force will push the compensation coil outwards when the coil is powered. This movement will be supported by expanding booms. The very small additional material budget of the compensation coil will have negligible impact on the detector performance. The position of the compensation coil is stabilized mechanically by Kevlar or Zylon straps connecting it to the main solenoid.

An alternative solution not requiring the compensation coil to be expandable consists in two static compensation coils, one at each end of the main solenoid. These compensation coils would each have a length of 0.4 m and a diameter of 8 m so that they fit into a SLS-sized fairing. With a weight of 250 kg each they would be connected mechanically to the end flanges of the main solenoid by lightweight radial carbon fiber rods. These design options have to be studied in more detail to decide on the optimal solution.

Compensating the magnetic dipole moment of the main solenoid requires a regulation of the current in the magnets at the ppm level, similar to the precision achieved for the current regulation of the LHC quadrupole magnets [106].

### 3.3 Instrumentation

The AMS-100 detector will be located on the cold side of the spacecraft’s sunshield (Fig. 1). The overall architecture is symmetrical and cylindrical, which provides a very large solid angle acceptance (Fig. 12).

The main solenoid is instrumented both on the outside and on the inside with a 3-layer high-resolution scintillating fiber (SciFi) tracker [107, 108] and a 2-layer time of flight system (ToF). The SciFi tracker is assumed to have a single point resolution of 40 µm. These sub-detectors will provide fast information on the incoming particles, as undistorted by the instrument as possible.

The inner detector consists of a silicon tracker, similar in design to the AMS-02 silicon tracker [109], followed by a pre-shower detector and a Lutetium-Yttrium oxyorthosilicate (LYSO) crystal calorimeter [110] with an outer radius of 40 cm. In addition to the SciFi-Tracker modules and ToF-detectors, the endcap opposite the service module is instrumented with γ converters to allow the reconstruction of low-energy photons with good angular resolution. These converters consist of silicon detector layers interleaved with thin tungsten layers as proposed for GAMMA-400 [111].

AMS-100 has a geometrical acceptance of 100 m² sr, i.e. 1000 times the acceptance of AMS-02. The instrument will monitor most of the sky continuously and will orbit around
Figure 12: Schematic view of the AMS-100 detector and its response to protons and positrons. The magnetic field inside the main solenoid is oriented in the $z$-direction, i.e. the bottom left view shows the bending plane of the magnet, and a transverse view is shown on the bottom right. The upper panel shows a zoom into the bending plane view.

the Sun in one year, together with Earth and L2. This will guarantee homogenous sky coverage for $\gamma$-ray astronomy. The weight estimate of the instrument is given in Table 3. It has eight million readout channels in total and an estimated total power consumption of 15 kW.

3.3.1 Event trigger

Reducing the 2 MHz rate of incoming particles to an acceptable level of a few kHz for the higher level DAQ systems and to a data rate of $\sim$28 Mbps [112] for the transfer to Earth with on-board computers will be a major challenge. To overcome it, the fast information provided by the outer detector (ToF-system and SciFi-tracker) will be used for the trigger decisions, in combination with calorimeter measurements: The track segments of the higher energy particles reconstructed in the SciFi tracker will provide a first estimate of the particle’s rigidity up to the TV scale, and the ToF signal amplitudes will determine the particle’s charge. This will allow the configuration of flexible trigger menus. For example, light nuclei with rigidity below 100 GV have to be mostly rejected. Charged particles with an energy below $\sim$ 100 MeV will be deflected by the magnetic field of the main solenoid.
and will not be able to enter the detector volume. Prescaled random triggers will be used to estimate the related trigger efficiencies. In addition, those SciFi- and ToF-layers located outside the main solenoid will be used to veto charged particles when reconstructing $\gamma$-rays.

### 3.3.2 Silicon tracker

The silicon tracker is assumed to have a single point resolution of 5 $\mu$m in the bending plane for $|Z|$ = 1 particles. It consists of six double layers arranged in cylindrical geometry (Fig. 12) leading to a maximum of 24 measurement points for a single track. For comparison, the CMS barrel silicon tracker [113] has an outer radius of 1.2 m and consists of 10 layers, providing up to 20 measured points for a cosmic muon going through the instrument. In combination with the 4 m diameter of the magnet and the magnetic field of 1 T, the AMS-100 silicon tracker provides an MDR of 100 TV.

### 3.3.3 Time-of-Flight system

To reconstruct particle masses and thus identify isotopes in cosmic rays, a high-resolution ToF-system is required. Such systems constructed from small scintillator rods with time resolution of 30 ps to 50 ps are presently under construction [114, 115]. We assume here that the time resolution of the PANDA ToF can be significantly improved using a larger coverage of the scintillator rods with SiPMs and operating the detector at 200 K. For $|Z|$ = 1 particles, we target for a time resolution of 20 ps for a single scintillator rod leading to a time resolution of 15 ps for the 4-layer ToF system as a result of the R&D process.

### 3.3.4 Calorimetry

The pre-shower detector and the LYSO crystal calorimeter are used to separate electromagnetic and hadronic showers, and to measure the energy of electrons, positrons and photons, as well as protons and ions beyond the MDR. The crystal calorimeter is inspired by the design of the HERD detector [110] and allows the three-dimensional reconstruction of the shower shape. The pre-shower detector consists of 12 silicon detector layers interleaved with thin tungsten layers to provide good angular resolution for the measurement of $\gamma$-rays and to limit the backspash of the calorimeter into the silicon tracker. This combination of pre-shower detector and crystal calorimeter has a depth of $70 X_0$, or $4 \lambda_I$, for particles incident in the bending plane of the main solenoid and hitting the calorimeter centrally. The geometrical acceptance of this system allows the measurement of cosmic nuclei with energies above 100 TV up to the cosmic-ray knee at the PeV scale (Fig. 13). With today’s accelerators, AMS-100 can only be calibrated up to 400 GeV. In orbit, the energy scale of the calorimeter system will be calibrated in the energy range from 100 GeV to 100 TeV using the rigidity measurement of charged cosmic rays in the spectrometer.

### 3.4 Spacecraft requirements

The instrument will be installed on a spacecraft and operated for at least ten years at L2. This positioning is necessary to create a stable cold environment for the operation of the HTS magnet. In a low-Earth orbit, the interaction of the residual magnetic moment with the geomagnetic field would result in a complete loss of attitude control. In addition, the
shadow of the Earth would reduce the field of view and the geomagnetic cutoff would limit the sensitivity to low-energy cosmic antimatter, in particular to antideuterons from dark matter annihilations.

To fulfill the science requirements, the full payload has a mass of $\sim 43 \text{ t}$ (dominated by the calorimeter) and hence will require new heavy-lift launch capabilities such as NASA’s Space Launch System (SLS) or China’s Long March 9 rocket, which are under development for human missions to Mars. Figure 14 illustrates the launch configuration in an SLS fairing.

Support tube and service module  The main structural element is a central 3 cm thick carbon support tube with an outer radius of 44 cm around the calorimeter. It will mechanically stabilize the detector during the launch and connect the service module to the launch adapter, which is the interface to the rocket. The main solenoid and the other subdetectors are connected to the central support tube by lightweight carbon fiber structures. Services are routed in the volumes between the barrel and the endcap detectors to the service module. The service module accommodates the DAQ system, the power distribution system, the telecommunication system, the attitude control, the thermal control system, and an electric propulsion system to keep a stable orbit around L2. A combination of reaction wheels and electric propulsion is used to keep the orientation of the sunshield stable with respect to the Sun.
Figure 14: AMS-100 launch configuration in an SLS-Block 2 fairing. The compensation coil, the sunshield, the solar cells, and the electric propulsion system are folded up. The service module is located at the top for structural reasons.

**Sunshield**  The sunshield is a key component of AMS-100, allowing the HTS magnet to operate without cryogens. It has a radius of 9 m and is designed similar to the concept developed for the James Webb Space Telescope [112]. The dimensions of the sunshield are chosen such that a pointing accuracy of a few degrees towards the Sun is sufficient to keep the magnet system cool. Other than for thermal reasons, the orientation of the instrument has no impact on the physics program. Star trackers will be used to monitor the orientation to provide precision information for the $\gamma$-ray astronomy program.

4 AMS-100 Pathfinder

The technical complexity of the AMS-100 project requires a pathfinder mission, similar to the AMS-01 flight on Space Shuttle Discovery in 1998 [116], or to the ongoing LISA program. This pathfinder mission has to demonstrate the stable operation of a HTS magnet in space for the first time, including the expandable compensation coil technology. It has to be operated at L2 to verify the thermo-mechanical design and to demonstrate the sufficient attitude control inside the time-varying interplanetary magnetic field. Testing the quench probability of the magnet system in this environment and the impact of a quench on the instrument is of key importance. The successful test will qualify similar HTS magnet configurations as radiation shield for a crew compartment for interplanetary manned space flights as discussed in Ref. [105].

Given the effort of a space mission at L2, a purely technical demonstrator mission would be a waste of resources. Therefore the AMS-100 pathfinder is anticipated to be a prototype at the 10% scale level of AMS-100, i.e. the length and the radius of the main solenoid are reduced by a factor 2 to $L = 3$ m and $R = 2$ m, so that the instrumented volume is reduced by nearly an order of magnitude. Its weight is estimated to be 5 t and its detector concept is in all other aspects very similar to AMS-100. The central calorimeter has to
be removed due to weight constraints as other components like the service module do not scale accordingly. With these dimensions and weight, the AMS-100 pathfinder can be launched to L2 with an Ariane 5 or a rocket of similar scale.

For the physics program of the pathfinder mission, the key performance parameters are a geometrical acceptance of $20 \text{ m}^2\text{sr}$ and an MDR of $5 \text{ TV}$. The sensitivity for heavy cosmic antimatter particles would be reduced compared to AMS-100 by an order of magnitude, but compared to AMS-02 this 10% scale pathfinder already has a $100 \times$ higher sensitivity to heavy cosmic antimatter particles and completely independent systematic uncertainties, due to its different detector geometry, detector technology and orbit.

## 5 Cost and schedule

The AMS-100 project falls into the ESA class L category, i.e. the full mission requires a budget of 1000 M€ to 1200 M€. The AMS-100 pathfinder mission falls into the ESA class M category, i.e. it requires a budget below 500 M€, with an estimated instrument cost of 150 M€.

The scale of the project requires a large international collaboration as successfully demonstrated by, e.g. the AMS-02 project on the International Space Station and the LHC experiments at CERN. An international consortium is currently forming to pursue the AMS-100 concept. A similar white paper to this one has been submitted in July 2019 for the Decadal Survey on Astronomy and Astrophysics (Astro2020) [117]. We strongly encourage the agencies to consider long term research objectives of this scale, as international research projects.

A possible timeline for the AMS-100 project is given in table 4. The important milestones for the R&D-Phase are the first successful space qualification test of a high temperature superconducting solenoid and the verification of the achievable time resolution of the ToF system. The detailed technical design report requires a valid thermo-mechanical model for the mission including a detailed concept of the detector electronics, DAQ system and data handling. With this white paper, we invite contributions from groups interested in participating in the project and encourage the community to support initial funding for the R&D stage of the mission, which we consider of the highest priority for future progress in astroparticle physics.

### Table 4: Estimated schedule for the AMS-100 project.

<table>
<thead>
<tr>
<th>Activity</th>
<th>Start Year</th>
<th>End Year</th>
</tr>
</thead>
<tbody>
<tr>
<td>Research and Development</td>
<td>2019</td>
<td>2021</td>
</tr>
<tr>
<td>Technical Design Report</td>
<td>2020</td>
<td>2022</td>
</tr>
<tr>
<td>Pathfinder Construction</td>
<td>2023</td>
<td>2028</td>
</tr>
<tr>
<td>Pathfinder Launch</td>
<td>2029</td>
<td></td>
</tr>
<tr>
<td>Pathfinder Science Ops</td>
<td>2030</td>
<td>2036</td>
</tr>
<tr>
<td>AMS-100 Construction</td>
<td>2031</td>
<td>2038</td>
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<tr>
<td>AMS-100 Launch</td>
<td>2039</td>
<td></td>
</tr>
<tr>
<td>AMS-100 Science Ops</td>
<td>2040</td>
<td>2050</td>
</tr>
</tbody>
</table>

## 6 Summary

The only magnetic spectrometer in space today, AMS-02, has collected more than 140 billion cosmic rays since 2011 and will continue to take data for the lifetime of the ISS, i.e. the next decade. AMS-100 is an ambitious project for the following decade which requires pushing today’s technology to its limits in several fields. Many demanding technical
questions need to be worked out in detail to make such a large space mission possible. These questions are of similar complexity as the ones that had to be solved to realize AMS-02 after the proposal in 1994 [118]. The AMS-100 concept as outlined in this article (Tab. 5) has the potential to improve the sensitivity of AMS-02 by a factor of 1000. This means that we will reproduce 20 years of AMS-02 data within the first week of operation at Lagrange Point 2. In the second week, we will start exploring completely new territory in precision cosmic-ray physics.

<table>
<thead>
<tr>
<th>Quantity</th>
<th>Value</th>
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</thead>
<tbody>
<tr>
<td>Acceptance</td>
<td>100 m² sr</td>
</tr>
<tr>
<td>MDR</td>
<td>100 TV</td>
</tr>
<tr>
<td>Material budget of main solenoid</td>
<td>0.12 X₀</td>
</tr>
<tr>
<td>Calorimeter depth</td>
<td>70 X₀, 4 λ_I</td>
</tr>
<tr>
<td>Energy reach</td>
<td>10¹⁶ eV</td>
</tr>
<tr>
<td></td>
<td>10 TeV</td>
</tr>
<tr>
<td></td>
<td>8 GeV/n</td>
</tr>
<tr>
<td>Angular resolution</td>
<td>4'' for photons at 1 TeV</td>
</tr>
<tr>
<td></td>
<td>0''4 for photons at 10 TeV</td>
</tr>
<tr>
<td>Spatial resolution (SciFi)</td>
<td>40 μm</td>
</tr>
<tr>
<td>Spatial resolution (Si-Tracker)</td>
<td>5 μm</td>
</tr>
<tr>
<td>Time resolution of single ToF bar</td>
<td>20 ps</td>
</tr>
<tr>
<td>Incoming particle rate</td>
<td>2 MHz</td>
</tr>
<tr>
<td>High-level trigger rate</td>
<td>few kHz</td>
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<tr>
<td>Downlink data rate</td>
<td>~28 Mbps</td>
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<tr>
<td>Instrument weight</td>
<td>43 t</td>
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<tr>
<td>Number of readout channels</td>
<td>8 million</td>
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<tr>
<td>Power consumption</td>
<td>15 kW</td>
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<tr>
<td>Mission flight time</td>
<td>10 years</td>
</tr>
</tbody>
</table>

Table 5: Important quantities in the AMS-100 design.
References


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