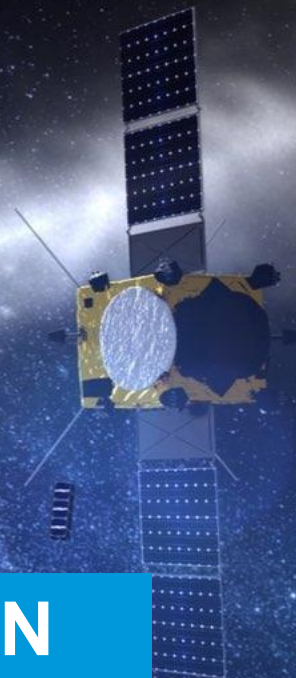




aim



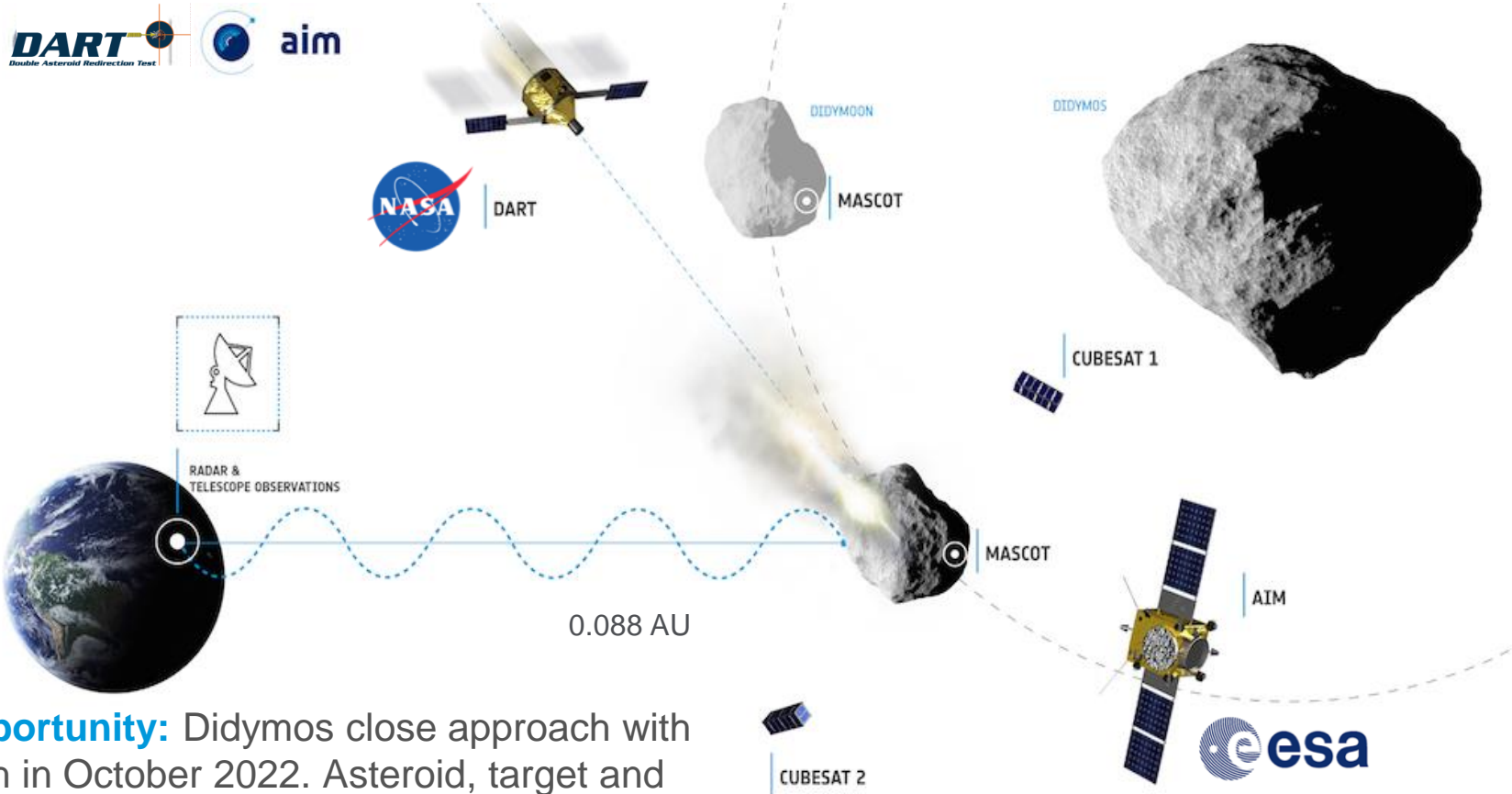
→ ASTEROID IMPACT MISSION

ESA UNCLASSIFIED - For Official Use



European Space Agency

AIDA INTERNATIONAL COOPERATION

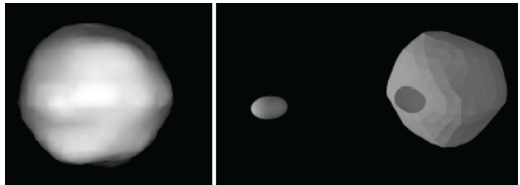
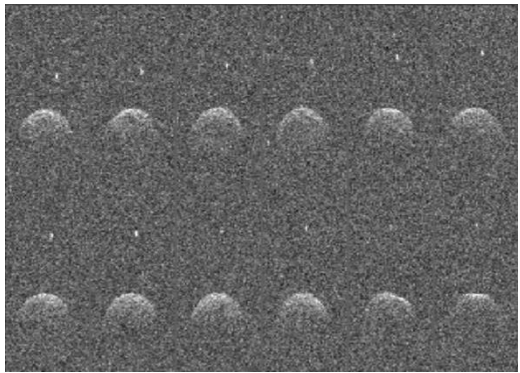


→ **opportunity:** Didymos close approach with Earth in October 2022. Asteroid, target and impact date fixed



DIDYMOS: A PERFECT TARGET

- Asteroid observed by ground telescopes and radars
- Heliocentric orbit well known
- Shape and size of primary well known (not Didymoon)
- Orbit plane orientation to be confirmed in 1Q 2017 (observations planned with European observatories)
- Didymoon size representative of a potentially hazardous object (generating



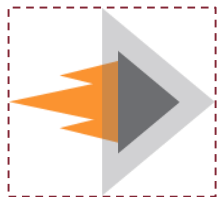
Chelyabinsk meteor (Feb 2013): 1500 injuries, 7200 damaged



AIM: A UNIQUE OPPORTUNITY



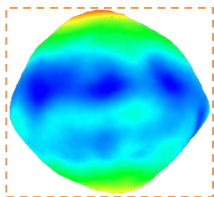
→ SPEED



→

- Fast “return on investments” (2 yrs) + double return with DART
- Asteroid operations: 6 months to achieve all objectives
- Demonstrate approach integrating platform-payload-operations teams for faster implementation
- Technology “firsts” enabling future:
LEO spacecraft architectures (swarms) and applications, debris-removal, sample-return, mining and human exploration missions
- Based on currently funded developments for:
on-board autonomy, CubeSats, advanced GNC, laser comm
- New industries to demonstrate technical capabilities in deep-space, stepping stone to future missions (e.g. Cosmic Vision)

AIM: A UNIQUE OPPORTUNITY



- First mission to a binary body, Solar System formation
- Impact dynamics beyond laboratory scale
- Probe the interior structure of small bodies (first time)
- Provide “ground-truth” for observations (radar, optical, meteorites)
- First mission to demonstrate planetary defence



- Public engagement and outreach similar or even beyond Rosetta (DART impact event visible from ground)
- Opportunity to provide visibility to space programmes at large
- Opportunity to enhance governments’ support in space activities

ASTEROID IMPACT MISSION



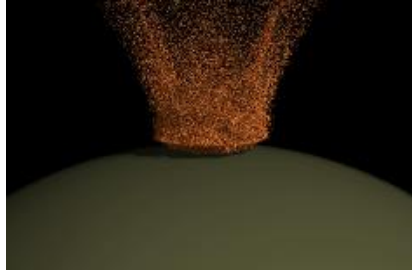
Interdisciplinary mission of opportunity to explore and **demonstrate technologies for future deep-space missions** while addressing **planetary defence objectives** and performing **asteroid scientific investigations**.



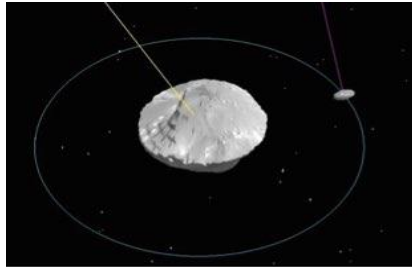
AIM "FIRSTS"



First mission to deploy **CubeSats** in deep-space and operate them through advanced **inter-satellite link** with embedded metrology, first semi-autonomous deployment of **micro-lander**, first demonstration of **interplanetary optical communication**



First mission to fully **measure** and **characterize** **asteroid** deflection, results enabling the validation of models to be applied to other asteroids if necessary.

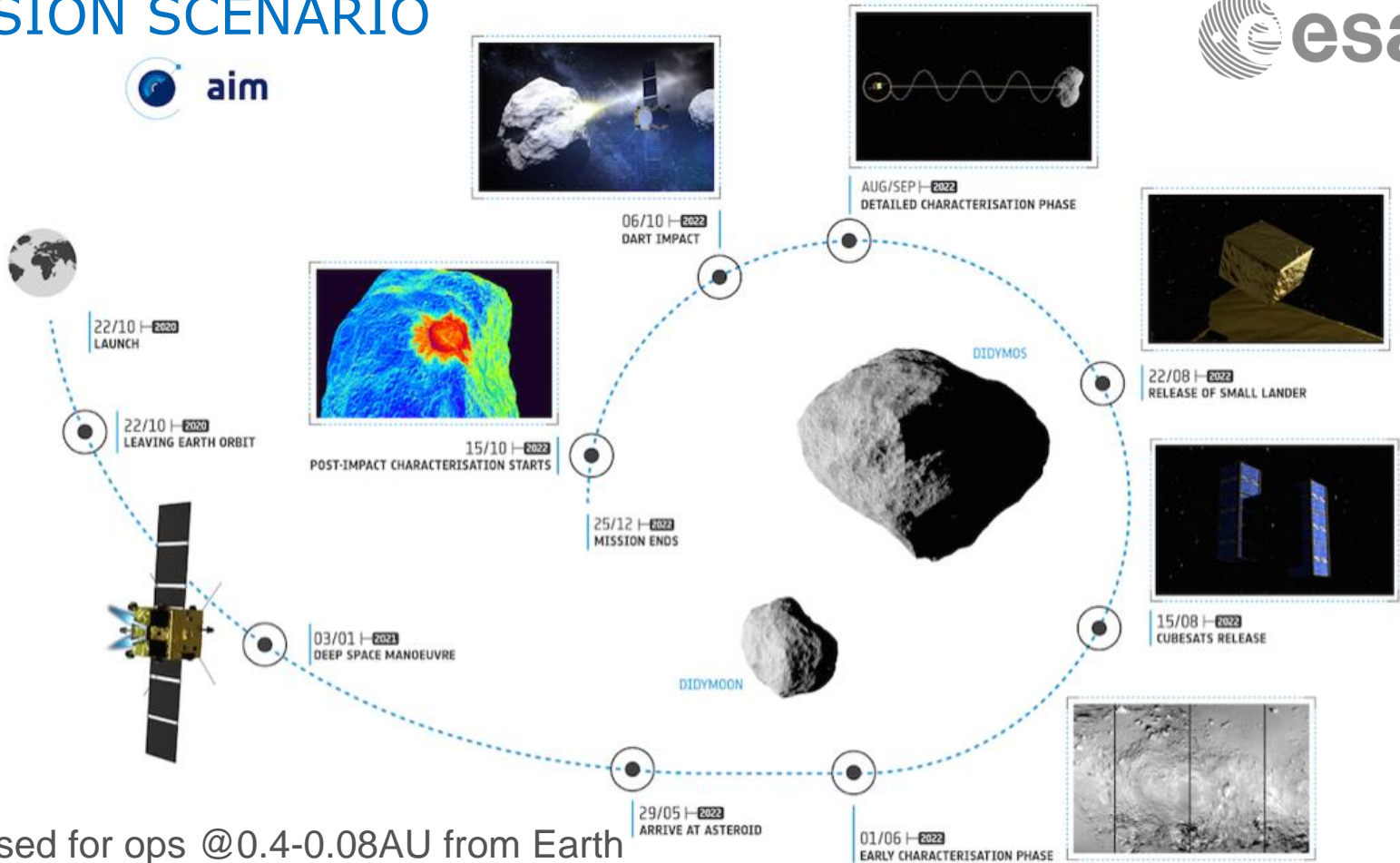


First mission to **study** a **binary asteroid**, its **origins** and sound the **interior structure** providing clues of its formation process.

AIM MISSION SCENARIO



Ariane 6
maiden flight
under study

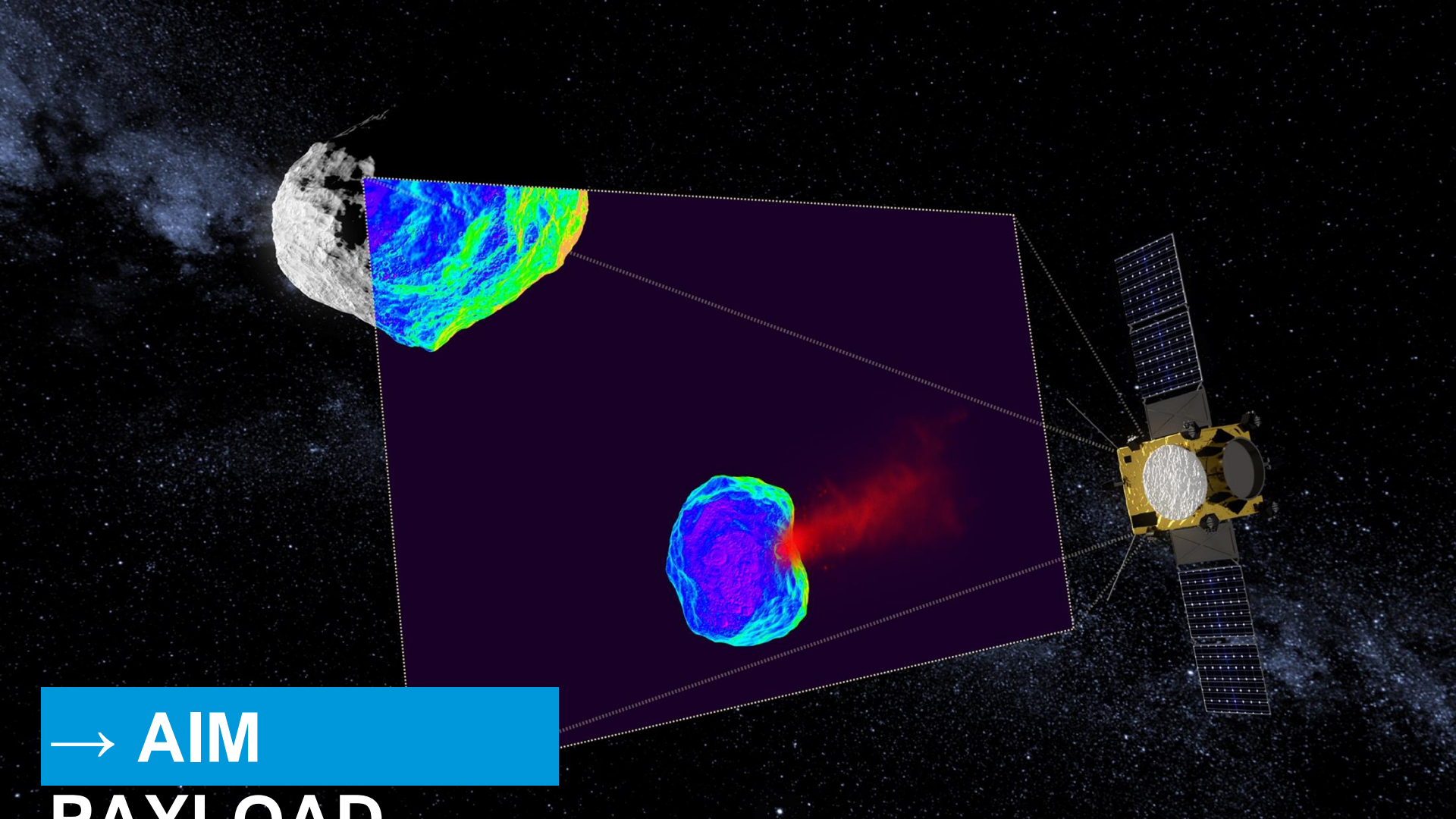


Direct escape
1.5 years cruise
0.5 years ops

Platform optimised for ops @0.4-0.08AU from Earth



European Space Agency

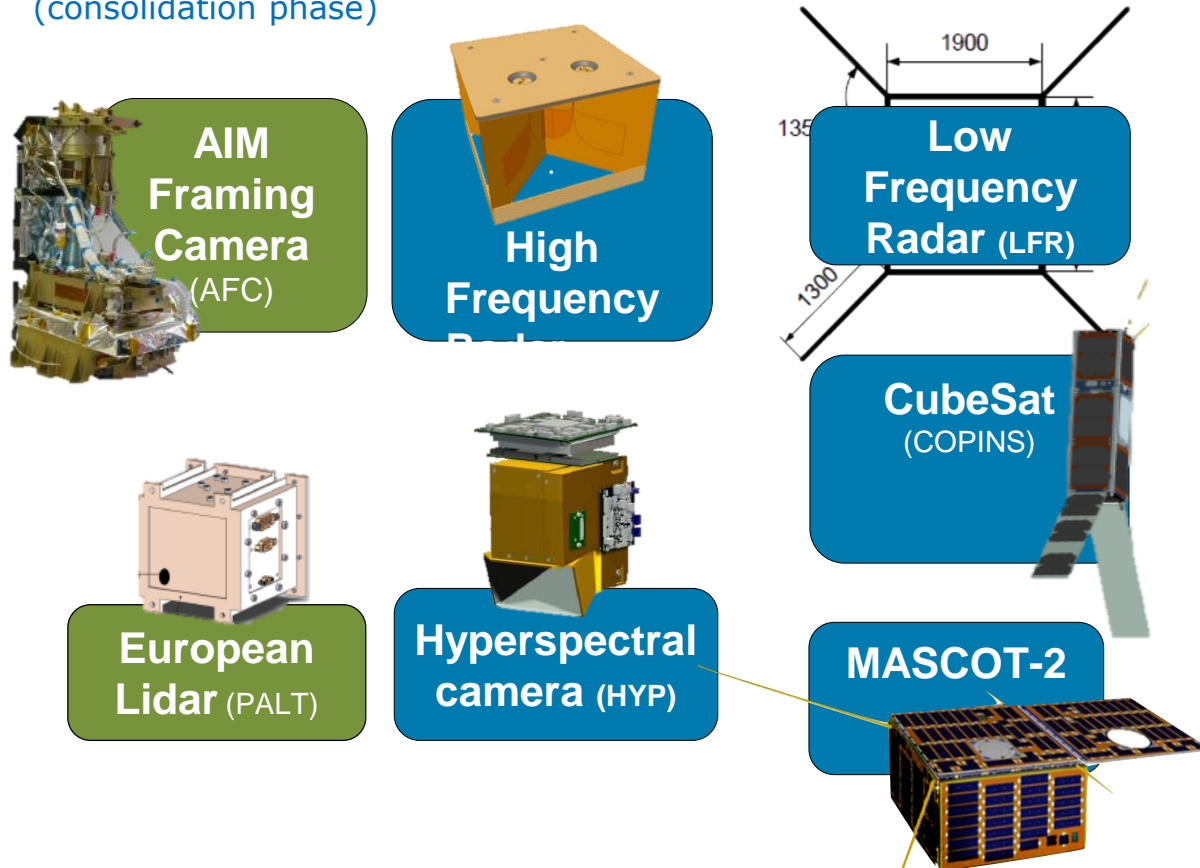


→ AIM

PAYLOAD

AIM REFERENCE PAYLOAD

(consolidation phase)



- Several options studied in detail to prepare for proper **interfaces** and **proximity operations**.
- Announcement for payload opportunities to be released in **Jan 2017** for any remaining mass following CM16 subscriptions by ESA Member States.

Legend:

Potential provider companies

(country)

Built-in AIM S/C (GNC subsystem)



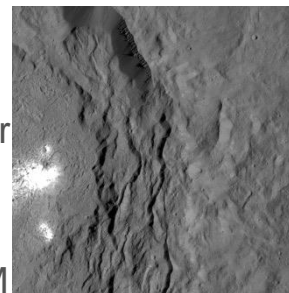
European Space Agency



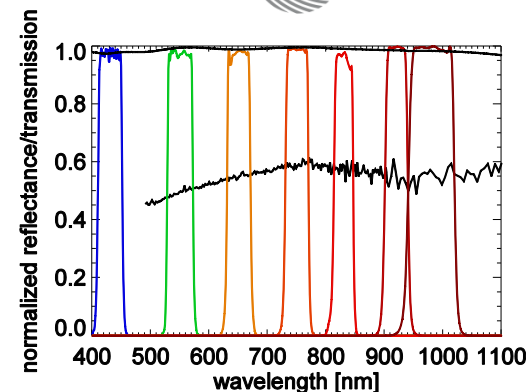
Flight Spares of the DAWN cameras

(5.5° FOV, 93.7 μ rad/pixel, 400-1000 nm, 7 filters)

- spacecraft GNC system, provided by MPI for solar system research
- Used for spacecraft navigation but also science
- Navigation currently being tested at GMV with QM

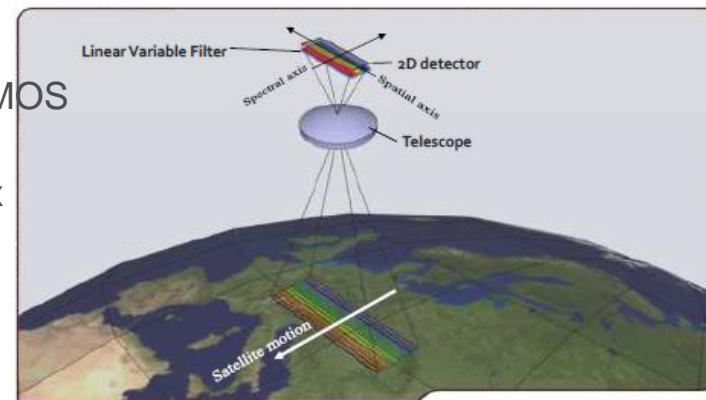


DAWN FC image of Ceres



Compact Hyperspectral imager

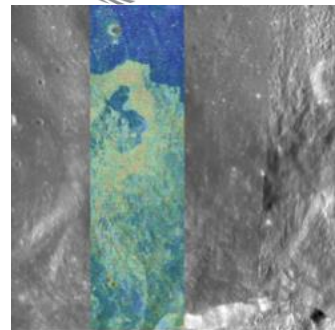
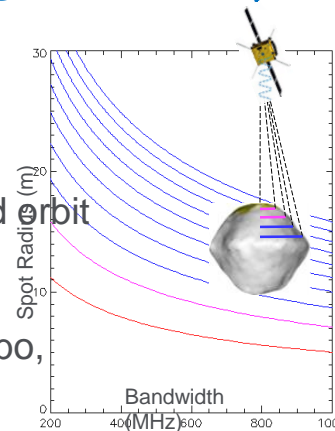
- Grating spectrometer or linear filter fixed on CMOS detector
- Large detector, 7 x 9 deg. FOV at 8 arcsec/pix
- Spectral resolution 5-10 nm
- Wavelength range 470-950 nm
- Developed for Earth observation



PLANETARY RESOURCES INC. (LUX), AMOS (BE), VITO (BE),
COSINE (NL)

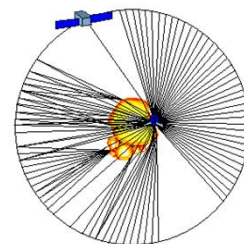


- ★ determine structure and layering of shallow sub-surface
- ★ support asteroid mass determination, shape modelling and characterisation
- ★ observe ejecta cloud
- ★ support ground-based bi-static radar measurements Arecibo, Goldstone, SRT



Instrument design based on CONSERT (Rosetta)

- Spare components available and TRL6
- Radar type: Bistatic radar (between AIM and MASCOT-2)
- Carrier frequency: 60 MHz
- Bistatic operation through the secondary asteroid



IPAG (FR), LATMOS (FR), Univ. Dresden (DE), ROB (BE), Antwerp Space (BE), Astronica (BE), and the European Space Agency (ESA) for their support and cooperation through the secondary asteroid mission.

THERMAL IMAGER (TIRI), MASCOT-2 μ LANDER

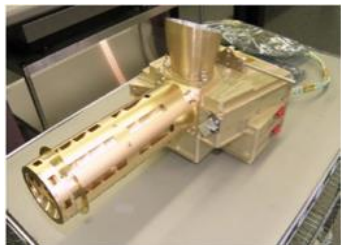
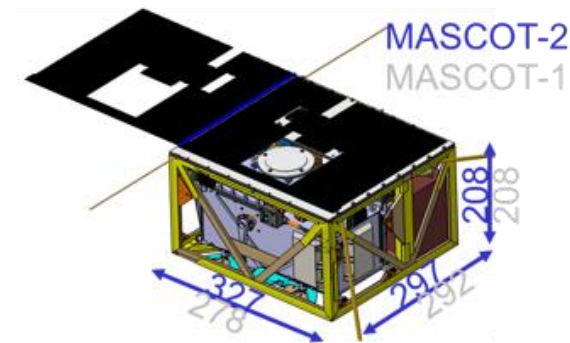
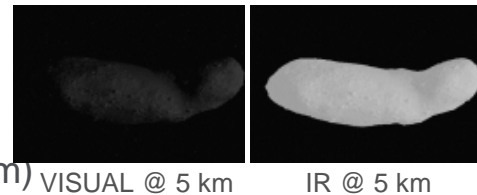


Figure: MERTIS

TIRI strawman design

- Heritage: MERTIS (Bepi-Colombo), MAIR, HIBRIS, AMS
 - Temperature range: 200 K – 450 K
 - Spectral range: 8 μ m – 13 μ m (spectral resolution 0.3 μ m)
 - Spatial resolution (goal): 2 m @ 10 km
 - Field of view: ~5 deg., similar to cameras
 - Thermal and physical surface properties
- COSINE (NL), GMV (PT), GMV (RO), SODERN (F), MPI (D), DLR (D)



MASCOT-2 μ lander

- Development based on MASCOT-1 currently on JAXA's Hayabusa-2 mission
- Size: 33 x 30 x 21 cm
- Mass: 15 kg
- Deployable solar generator cover (supports orientation)
- 3 months operational lifetime
- Carries: μ -camera (CAM), low-frequency radar (LFR), radiometer

DLR (DE), SSC (SE), Cobham Gaisler (SE), CBK (PL), Astronika (PL), COSINE (NL), CGS (I), SELEX (I), POLIMI (I), Space-X (CH), CSEM (CH), MCSE (CH)



COPINS: A CASE FOR CUBESATS IN DEEP SPACE



ASPECT



- Vis-NIR imaging spectrometer
- Space Weathering
- Shock experiment
- Plume

VTT (FI), Univ. Helsinki (FI), Aalto Univ. (FI), CAS (CZ)

AGEX



- Mechanical properties of surface material
- Seismic properties of sub-surface
- Determine kinematics prior and

ROB (BE), ISAE (FR), Antw. Space (BE), EMXYS (ES)

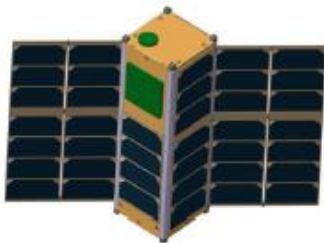
PALS



- Characterize magnetization
- Composition of volatiles
- Volatiles released from DART impact
- Super-resolution imaging
- DART collision and plume

IFR (SE), AAC (SE), DLR (DE), IEEC (ES), KTH (SE)

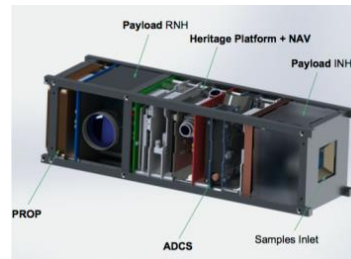
CUBATA



- Gravity field
- Observe DART impact
- Perform seismology
- Velocity field of the ejecta

GMV (ES), Sapienza Univ. Roma (IT), INTA (ES)

DUSTCUBE

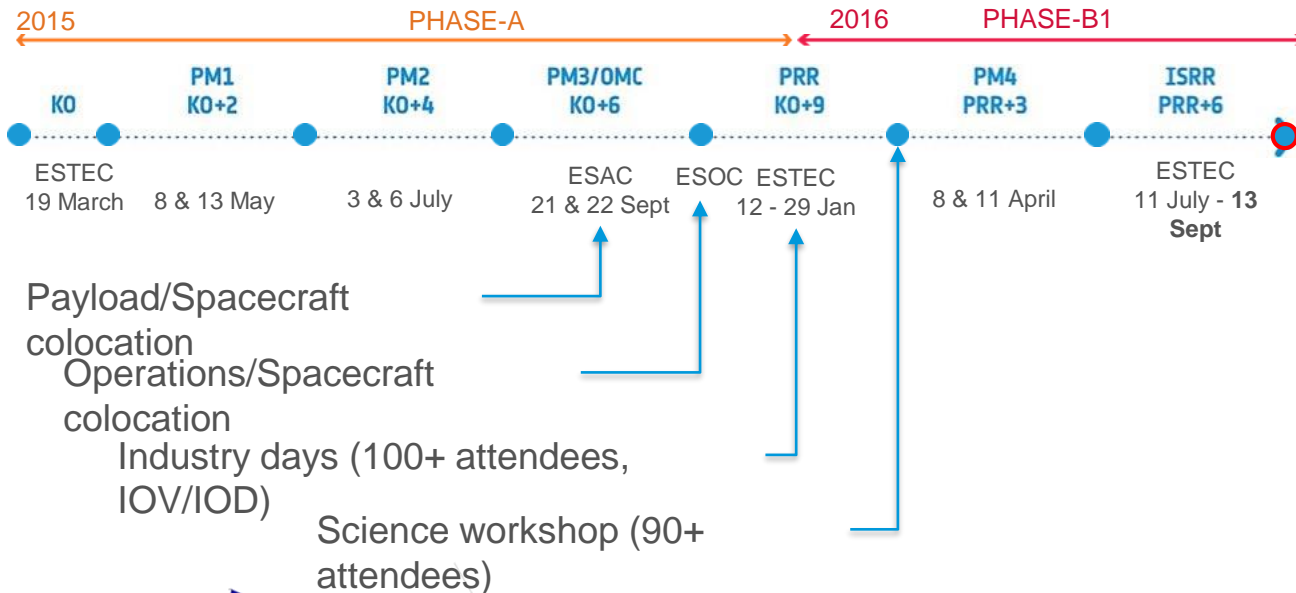


- Dust properties with Nephelometer
- Mineralogical composition
- Compliment com demo
- Reflectance of the

Univ. Vigo (ES), UniBO (I) Micos (CH), Univ. Bern (CH)



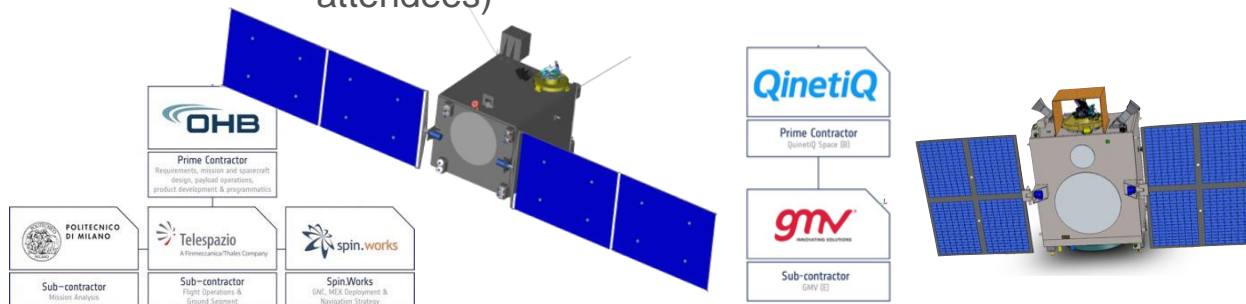
SYSTEM ACTIVITIES: PHASE B1 COMPLETED



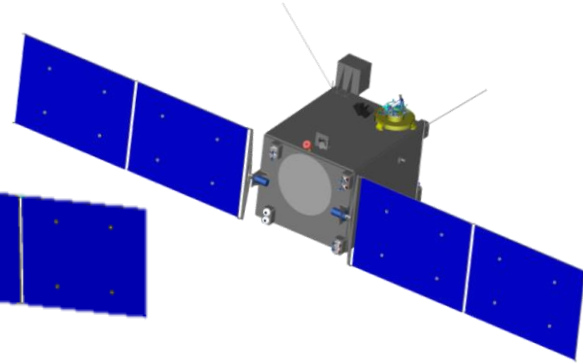
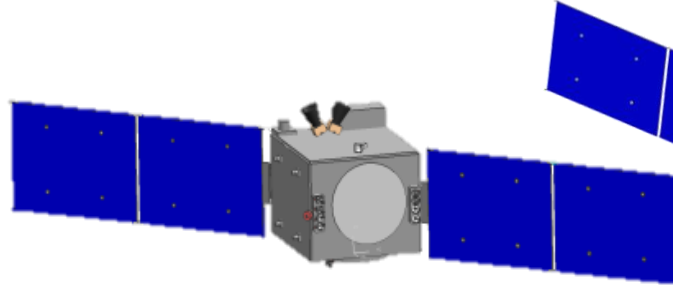
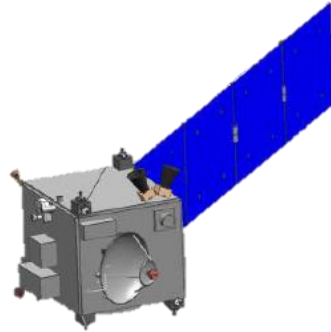
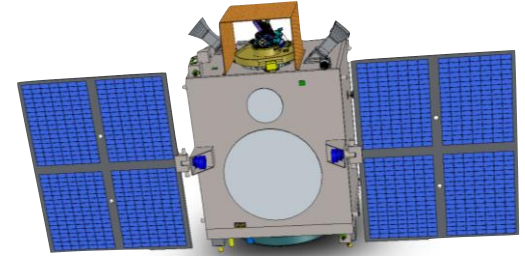
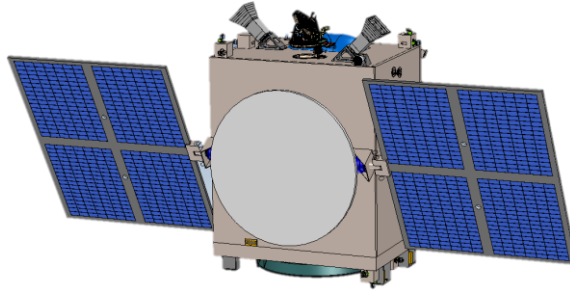
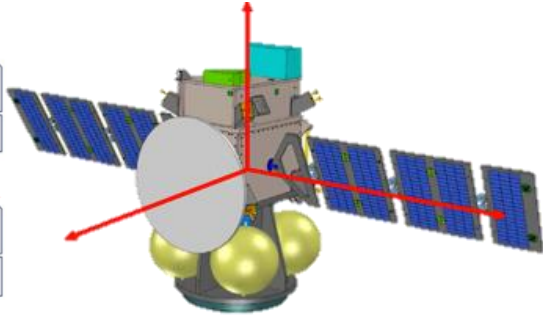
- Preliminary feasibility confirmed

→ ENABLING APPROACH

- Cost and schedule driven
- Platform and payload “integrated” teams
- Early OPS and FDyn teams support (Rosetta)
- Early GNC testing and validation in lab
- Reuse of flight spares (e.g. DAWN framing camera)



CURRENT SYSTEM DESIGN (B1)



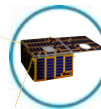
PAYLOAD ACTIVITIES: TOWARDS COMPLETION

→ AFTER

2015

2016

MASCOT
CONCERT



Exploration

Optel-μ
LADEE



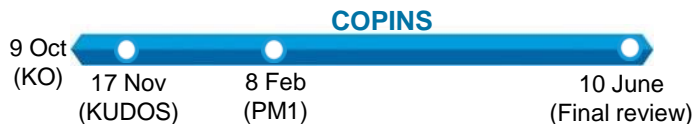
Mars,
human
exploration,
L-missions

WISDOM



Planetary
exploration

Aalto-1, Picasso,
Xatcobeo, HumSat-D,
Optos, SEAM...



ESA CDF
Study (Jan
17)

Earth
observation,
lunar explo

MERTIS, CAMIR,
Hypercube...

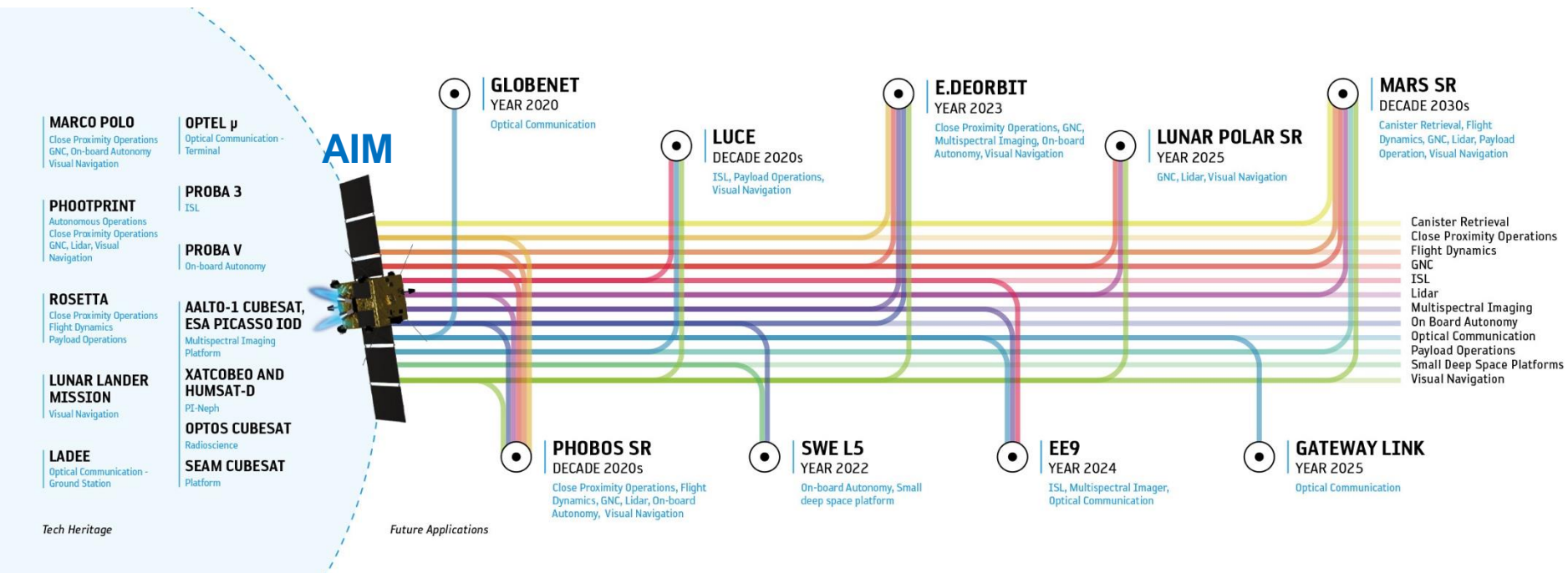


RDV &
docking
EOP,
CubeSats,
planetary
Exploration

CIVA, X-CAM,
Clupi...



aim is...TECHNOLOGY OF THE FUTURE



aim is... INSPIRATION AND OUTREACH

(416.000 results on Google for "asteroid impact mission")



AIM sand art performance



Asteroid Day press conf AIM @ planetarii Astrofest (London, June 2016)



Science&Vie Magazine



AIM videogame



Design your asteroid school contest



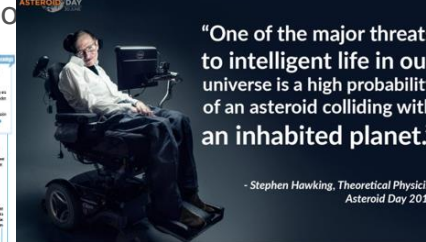
"Men vs Asteroid" Discovery Channel documentary



Asteroid Day (Bucharest, 2015)



CNN TV news



Stephen Hawking supporting Asteroid Day 2016



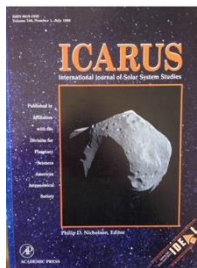
European Space Agency

aim is... HIGH SCIENCE RETURN



Fly-by of Lutetia (Rosetta) in 2010:

- Special issue of Science
- 47 referred articles (source ADS)



Fly-by of Steins (Rosetta) in 2008:

- 33 referred articles (source ADS)

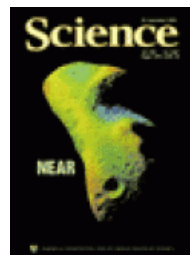
Fly-by of Mathilde (NEAR) in 1997:

- Special issue of Icarus
- 19 referred articles



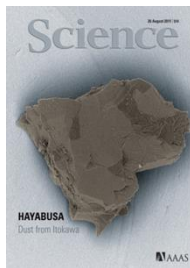
Radar observations of Kleopatra:

- Cover of Science



Visit of Eros (NEAR) in 2000-2001:

- Special Issue of Science
- One book (Cambridge Press)
- 156 referred articles (source ADS)



Visit of Itokawa (Hayabusa) in 2005:

- Two Special Issues of Science
- 83 referred articles (source ADS)
- Three Hollywood-like movies



Visit of Ceres (DAWN) in 2015:

- Special Issues of Science
- 206 referred articles (source ADS) for Ceres and Vesta

AIM SCHEDULE & STATUS



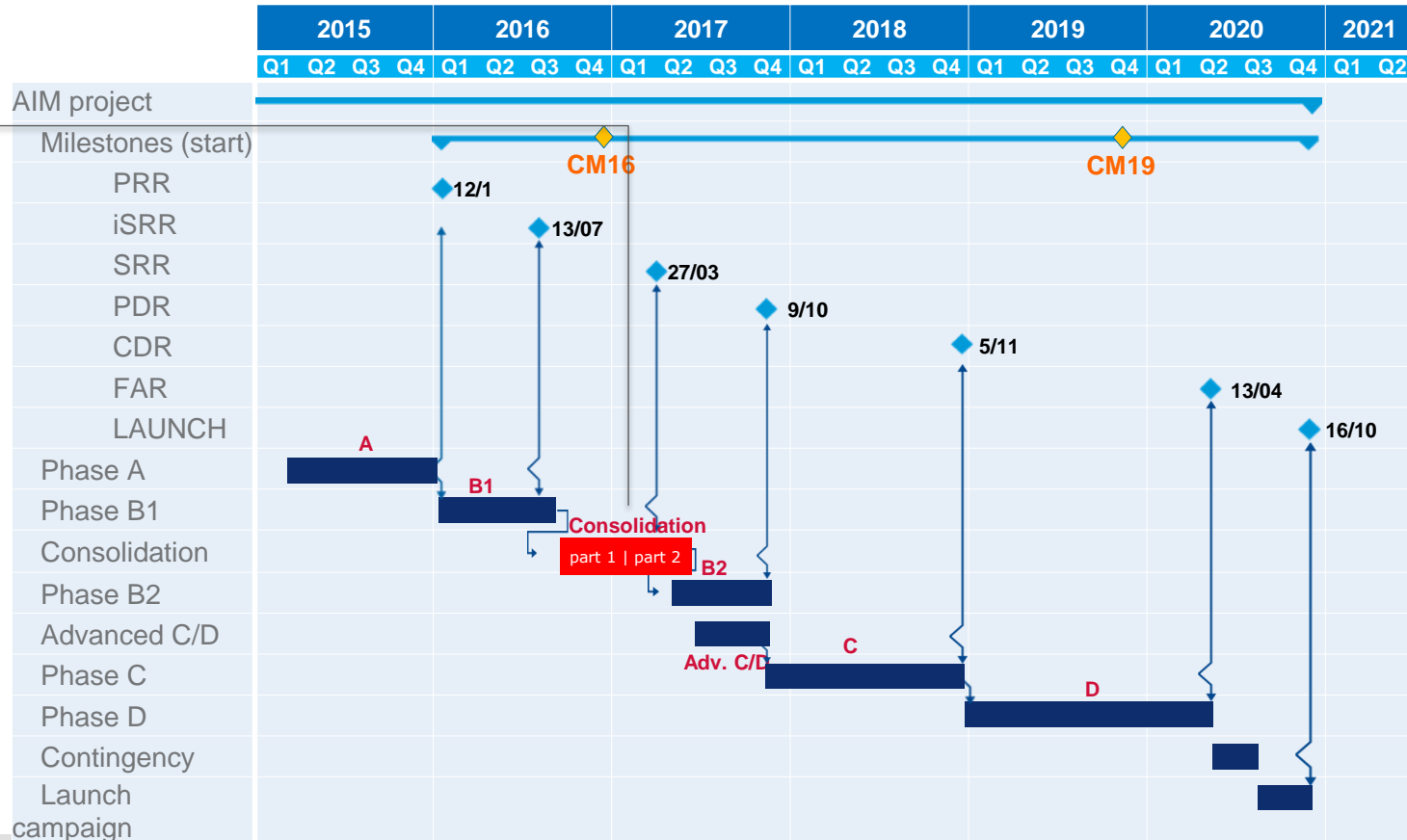
→ NEXT

ITT Consolidation Phase published (4.5M€)

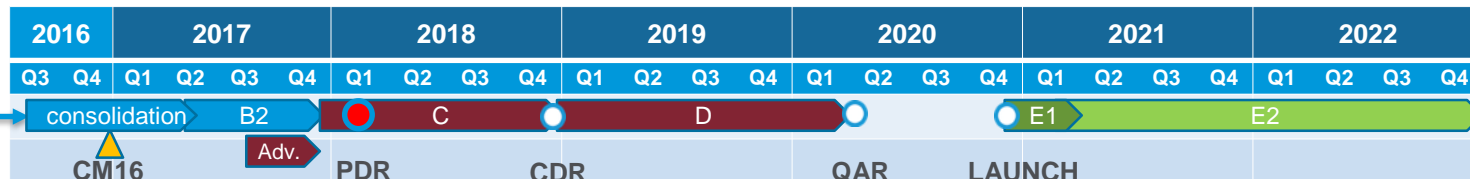
- Spacecraft design consolidation
- Team organization
- Consolidation of CaC and implementation plan

Supported by:

- Germany
- Belgium
- Spain
- Portugal
- Romania
- Poland



AIM CONSOLIDATION PHASE



Consolidation phase approved:

PART 1

consolidation phase (4.5M€)

PART 2

KO 16
OCT

- PART 1 ITT released on EMITS (Aug 2016)

Consolidation of the spacecraft design down to equipment level, preliminary design of the GNC and associated FDIR algorithms, operations concept definition, program implementation definition, preliminary subco/suppliers identification, consolidation of cost estimates

CM16

payload
AO
JAN17

Consolidation of MS interests following CM16 subscriptions

RFQ
FEB17

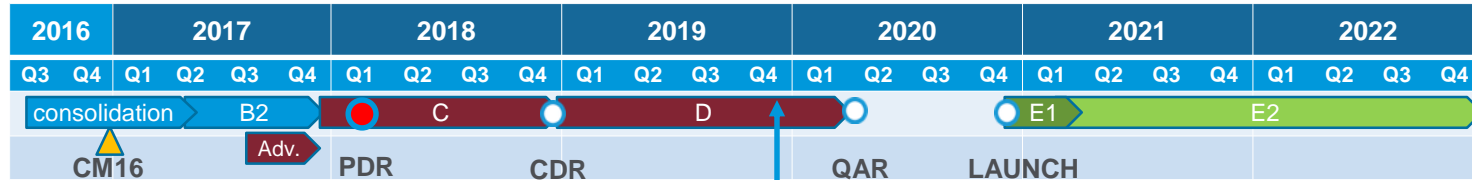
- PART 2 RFQ to be released in Feb 2017, team consolidation reflecting CM16 level subscriptions

- KO planned Apr 17, covering activities up to Sep 2017 (pre-PDR)

Further consolidation of the spacecraft design down to equipment level, down-flow of requirements and specifications. GNC development. Industrial team consolidation also on the basis of the MS

subscriptions. Identification of LLIs and critical pre-developments.

AIM IMPLEMENTATION

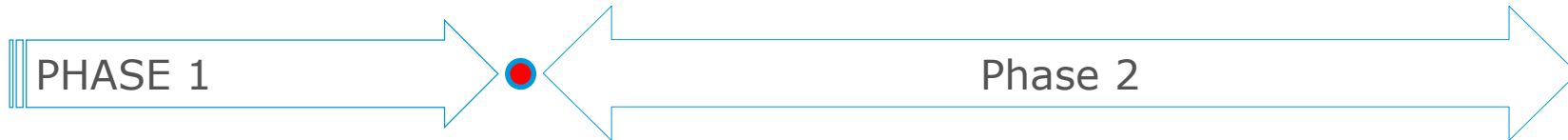


▲ Implementation Decision B2/C/D/E

● Authorization to CDE

- Successful PDR
- Schedule confirmation
- Status of DART mission

Cover remaining common costs (operations, launcher) and risks



Completion of phase B2 activities up to PDR, finalization of LLI procurement

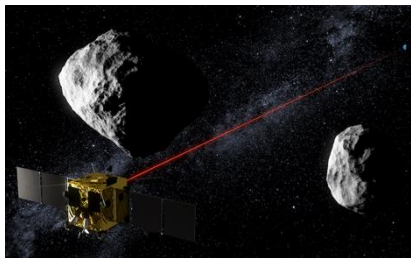
Spacecraft detailed design definition and successful critical design review (CDR), spacecraft production and ground qualification, successful qualification review (QR) and acceptance review (AR). Completion of Phase E activities.



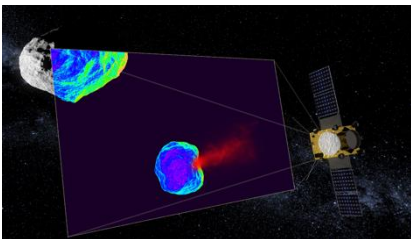
CONCLUSIONS



First mission to prove **asteroid deflection** with an experiment visible from ground, to reach a **binary asteroid** and to deploy small **CubeSats**. Great **inspiration for European citizens**.



Maintain European industry **leadership** in **GNC**, **complex proximity operations**, forefront of deep-space **optical communications**, first to **operate multiple-platforms** in close vicinity, demonstrate new approach for **fast missions in deep-space**.



Enabling future **space mission architectures** (swarms), **new applications** for Earth monitoring and services, capabilities for **autonomous operations**, demonstrate key technologies for **exploration missions**, **opportunity for new industries** to gain deep-space heritage.



For more information
www.esa.int/AIM