



# Apophis Recon Concepts

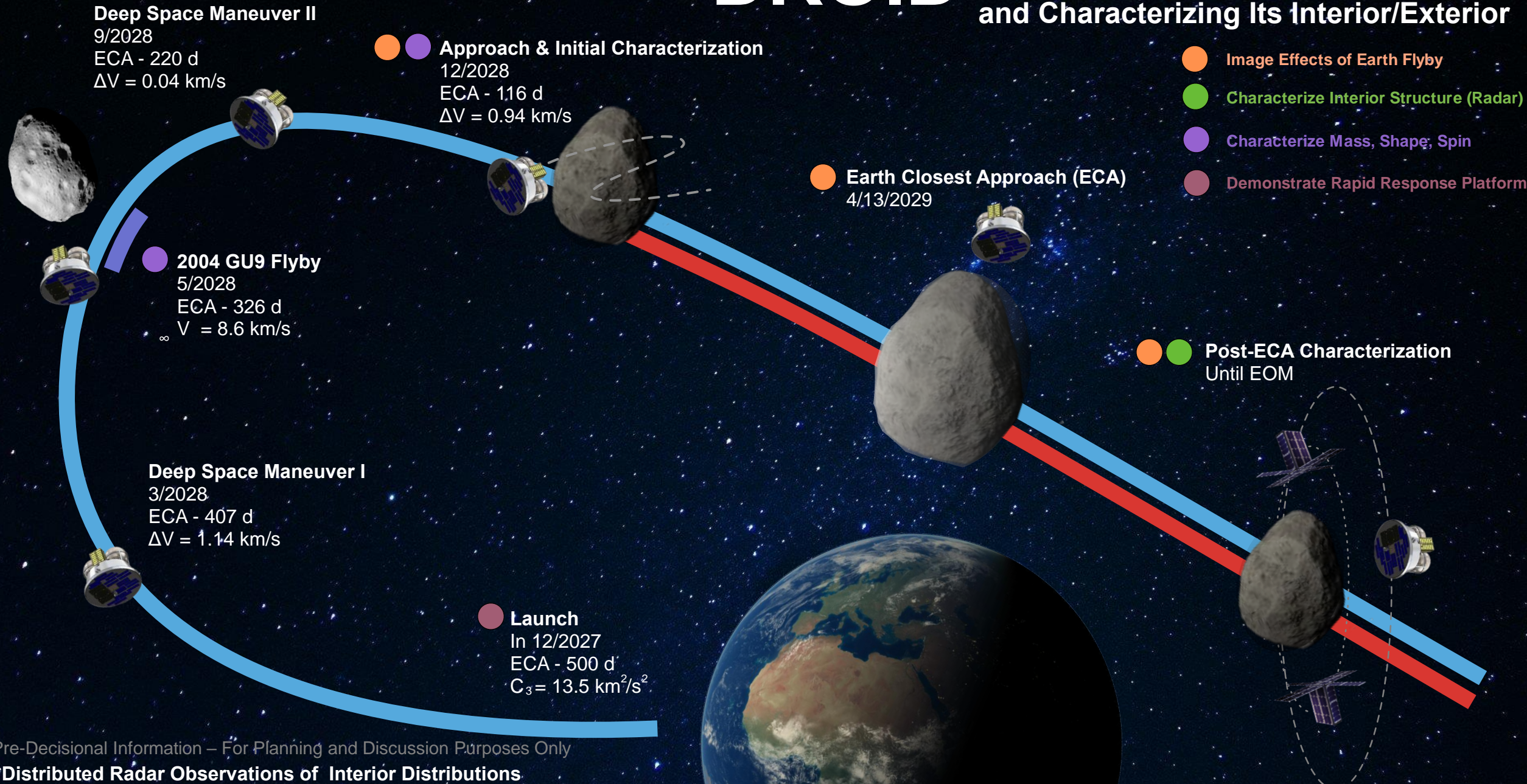
## SMPAG Mtg #21

Lindley Johnson, Planetary Defense Officer

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# DROID\* Shepherding Apophis Through Close Approach, Imaging Surface Changes, and Characterizing Its Interior/Exterior



**Deep Space Maneuver II**  
 9/2028  
 ECA - 220 d  
 $\Delta V = 0.04 \text{ km/s}$

**Approach & Initial Characterization**  
 12/2028  
 ECA - 116 d  
 $\Delta V = 0.94 \text{ km/s}$

- Image Effects of Earth Flyby
- Characterize Interior Structure (Radar)
- Characterize Mass, Shape, Spin
- Demonstrate Rapid Response Platform

**Earth Closest Approach (ECA)**  
 4/13/2029

**2004 GU9 Flyby**  
 5/2028  
 ECA - 326 d  
 $V_{\infty} = 8.6 \text{ km/s}$

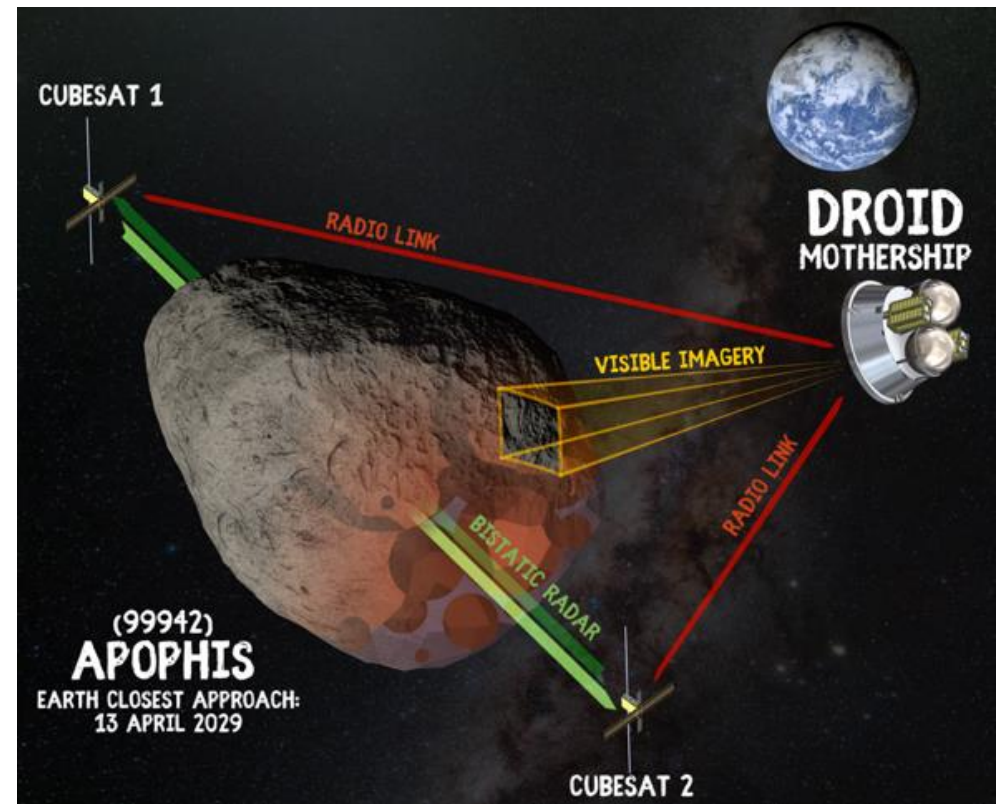
**Post-ECA Characterization**  
 Until EOM

**Deep Space Maneuver I**  
 3/2028  
 ECA - 407 d  
 $\Delta V = 1.14 \text{ km/s}$

**Launch**  
 In 12/2027  
 ECA - 500 d  
 $C_3 = 13.5 \text{ km}^2/\text{s}^2$

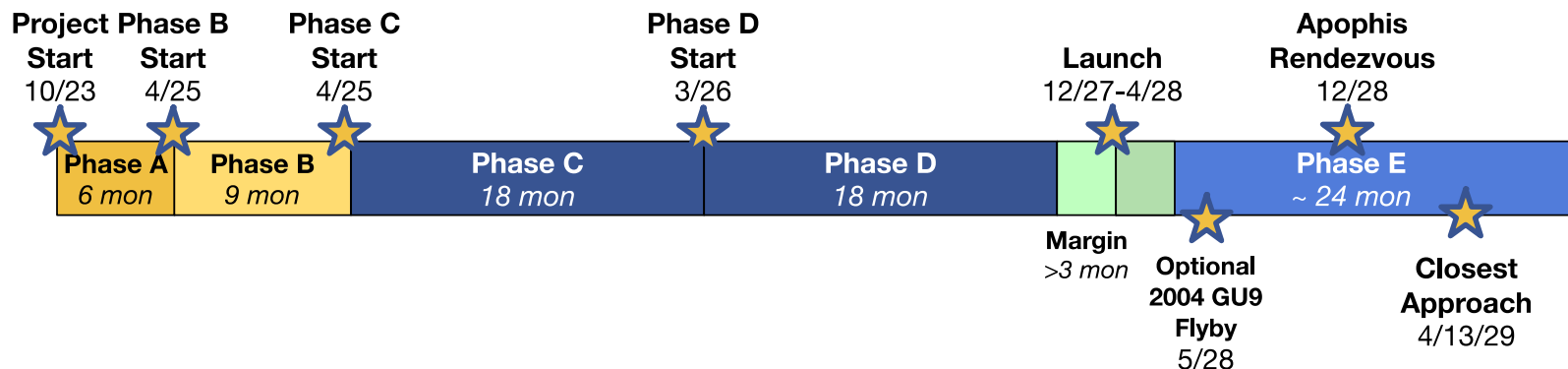
# DROID is an International Collaboration

- **CNES:** *CubeSats, Wide Angle Cameras, Inter-Satellite Links*
- **Rocket Lab:** *Mothership, Launch Vehicle*
- **University of Grenoble, Luxembourg:** *Monostatic/Bistatic Radar*
- **JPL:** *Management, Narrow Angle Camera*



*CubeSats use their radars to characterize interior structure and dielectric distribution as the Mothership develops a <5 cm/px map.*

## Preliminary Schedule



# FLARE Mission Overview

**Launch 6/1/2028**

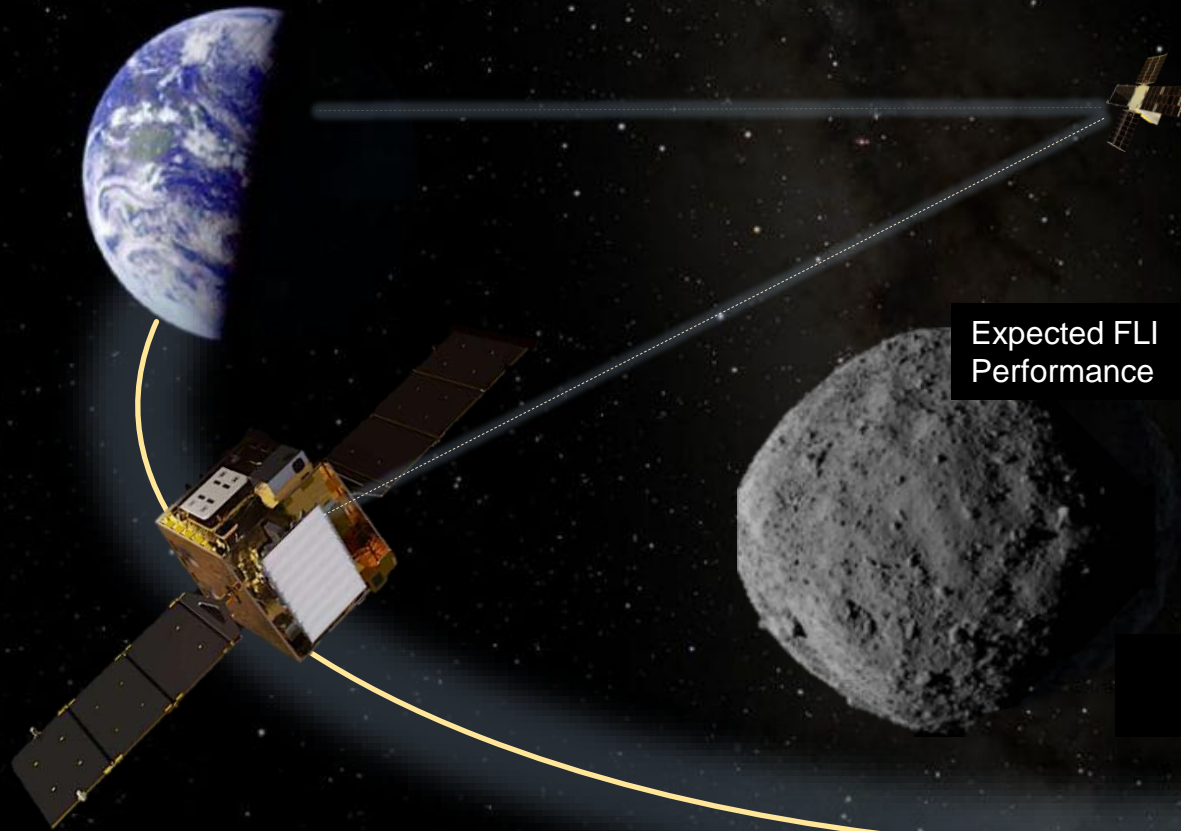
VSFB or KSC

Falcon 9

**Flyby 1/21/2029**

Speed 2 km/s

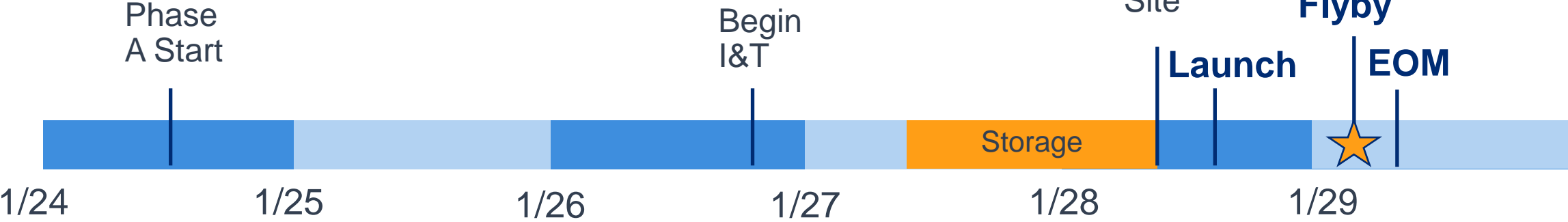
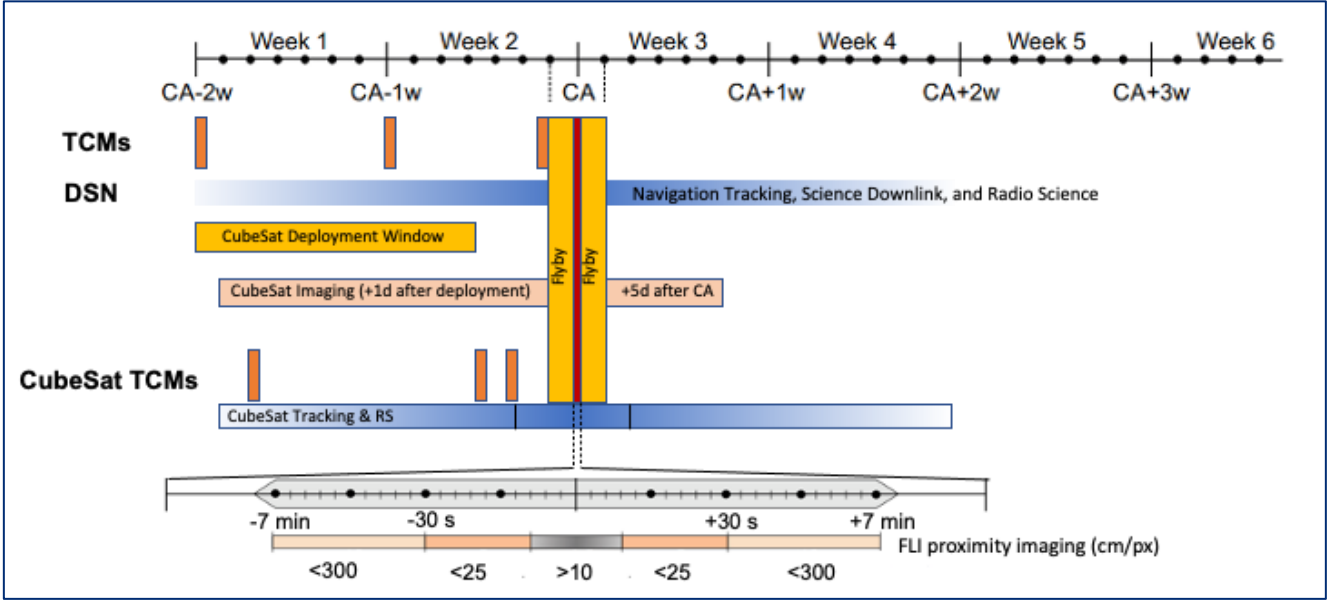
Closest Approach Distance 40 km



**FLARE** (Flyby Asteroid REconnnaissance) Mission uses a unique opportunity to establish the utility of flyby data by:

1. Characterizing the key physical properties of Apophis that are important for planetary defense, including shape and mass of the asteroid, using two instruments: a Narrow Angle Imager (FLI) and CubeSat Mass experiment (FLAME)
2. Testing rapid response capabilities in a compressed schedule.
3. Validating measurements with higher quality "truth data" from the OSIRIS - APEX rendezvous mission
4. Establishing the surface conditions of Apophis before its close approach

# Baseline Conops, Schedule, and Cost



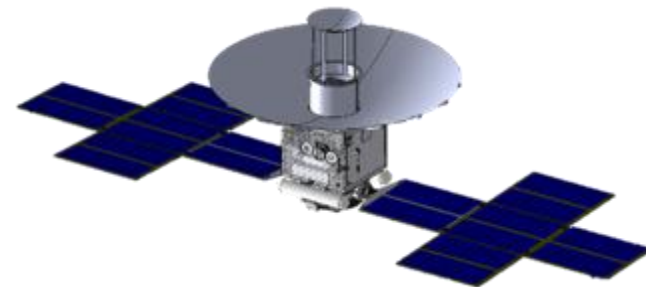
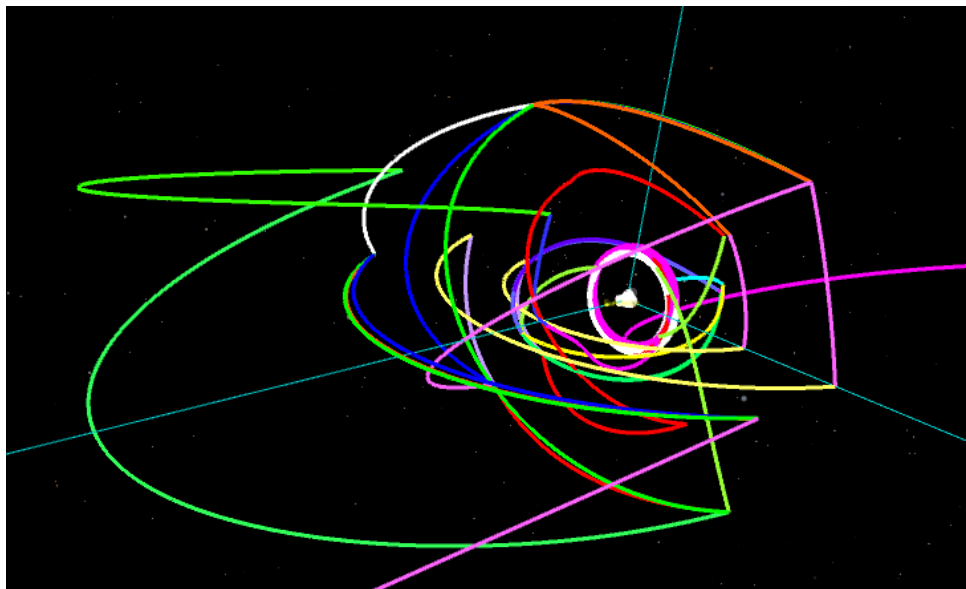
\*Schedule Challenges can be implemented to further reduce durations

# Reconnaissance of Apophis (RA)

NASA/Goddard Space Flight Center (GSFC)

*RA is Carefully Designed to Provide Significant Science On a Smallsat Budget*

RA is a capable smallsat rendezvous mission able to detect any disturbances to Apophis during the 2029 Earth approach, through both real-time observations and comparison of pre-/post-Earth close approach survey data. The shape models will have a resolution of  $\leq 30$  cm, with resolution as fine as 5 cm in select areas. Proximity operations have been designed based on OSIRIS-REx heritage, by personnel with OSIRIS-REx flight operations experience.

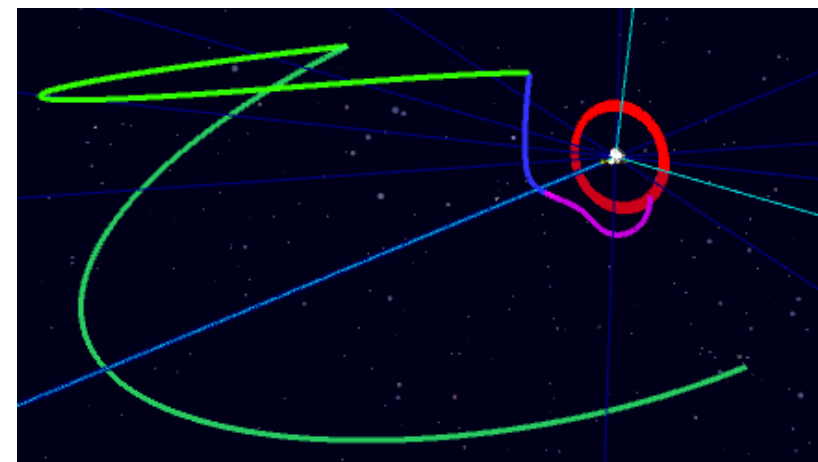


# Notional Schedule for RA



*Note: notional schedule is an unapproved draft.*

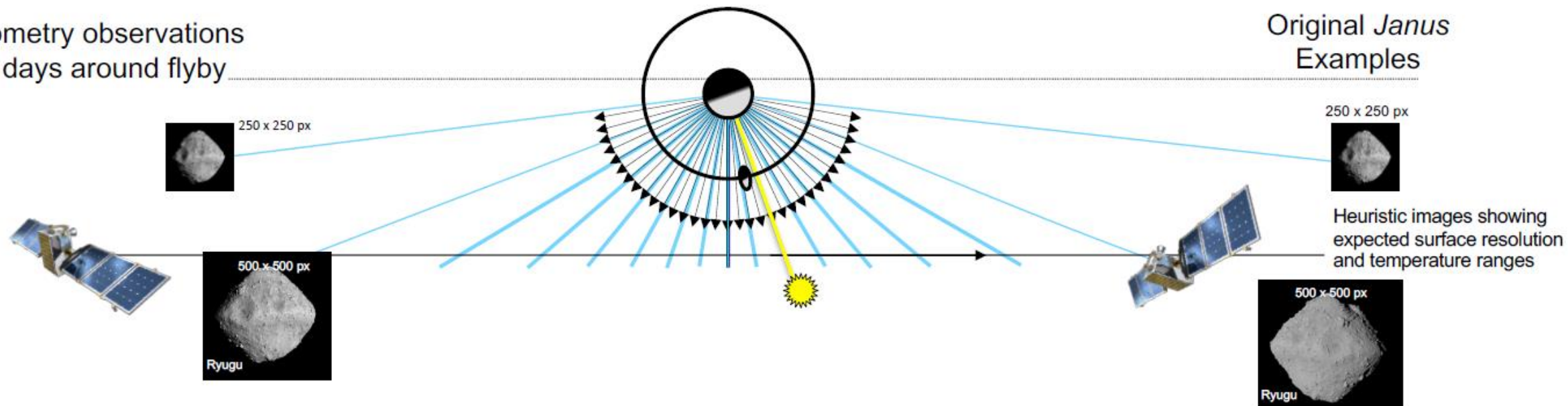
RA can follow Apophis through Earth close approach while both maintaining passive safety and remaining close enough to collect real-time observations.



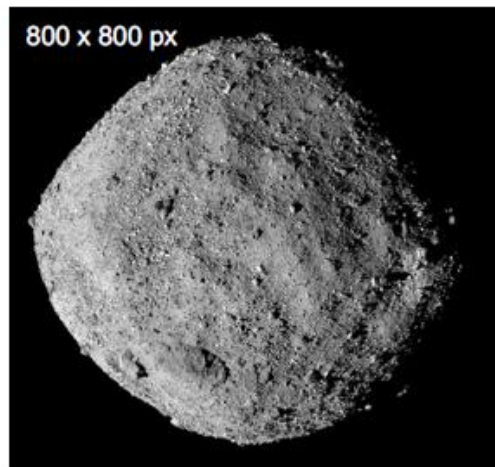
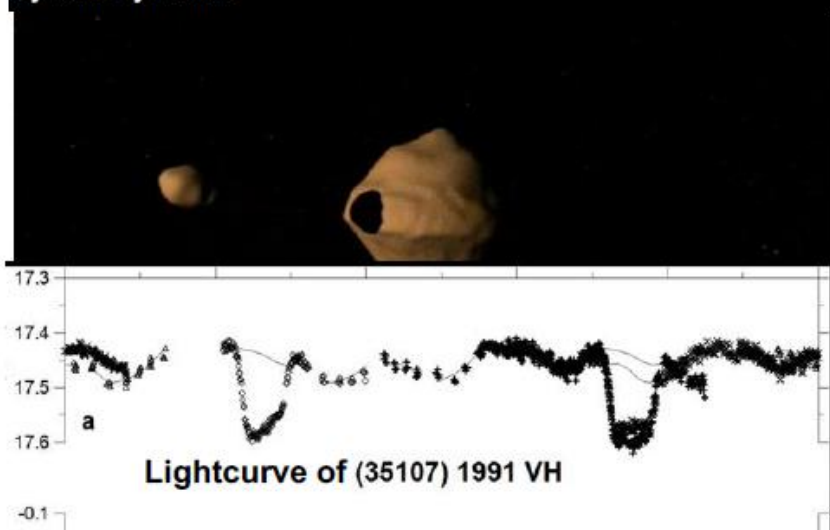
# Janus Apophis observations use the same design as the original Janus

Photometry observations  
~±2 days around flyby

Original *Janus*  
Examples

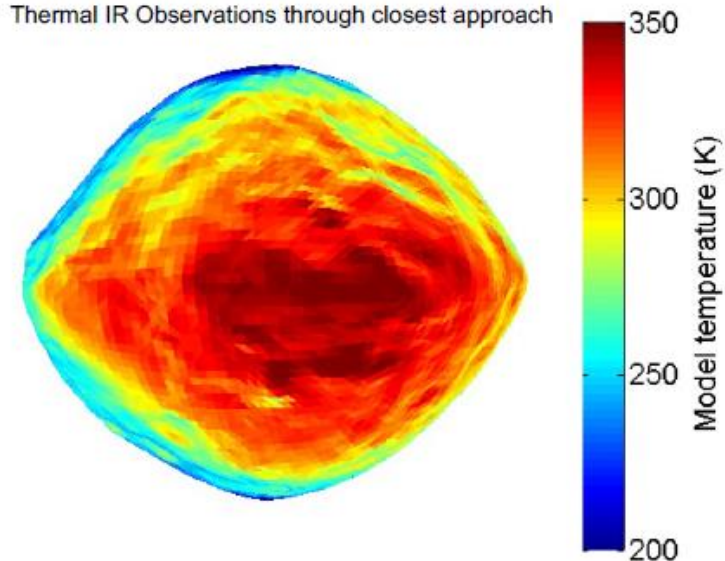


Photometry Observations acquire "light curves" showing system dynamics



Visible imaging will provide surface morphology at a range of phase angles.

Thermal IR Observations through closest approach





# Small Bodies Assessment Group (SBAG) Specific Action Team - Apophis Report

**Table 2.** Apophis science in 2029 enabled by ground-based and space-based optical facilities

Optical wavelength	Immediate data products	Apophis science	Example Facilities
IR	unresolved images	<ul style="list-style-type: none"> <li>• effective diameter;</li> <li>• thermal inertia</li> <li>• limited mineralogy</li> </ul>	NEO Surveyor
	spectra	<ul style="list-style-type: none"> <li>• mineralogy and bulk composition;</li> <li>• space weathering;</li> <li>• taxonomy;</li> <li>• inferred microscopic surface density</li> </ul>	SpeX@NASA IRTF
mm to sub-mm	unresolved? images	<ul style="list-style-type: none"> <li>• thermal inertia</li> </ul>	ALMA
Visible	light curves	<ul style="list-style-type: none"> <li>• 3D convex shape model;</li> <li>• spin state;</li> <li>• moment of inertia ratios;</li> </ul>	
	spectra	<ul style="list-style-type: none"> <li>• taxonomy;</li> <li>• compositional analogs</li> </ul>	ubiquitous
	occultation chords	<ul style="list-style-type: none"> <li>• size;</li> <li>• limited shape information;</li> <li>• heliocentric orbit improvement</li> </ul>	
	adaptive optics	<ul style="list-style-type: none"> <li>• 10-m resolution images</li> </ul>	SPHERE@VLT
	speckle imaging	<ul style="list-style-type: none"> <li>• 10-m resolution images</li> </ul>	QWSSI@LDT
	polarimetry	<ul style="list-style-type: none"> <li>• regolith properties;</li> <li>• optical albedo;</li> <li>• surface heterogeneity</li> </ul>	WIRC+Pd@Palomar 200"

SPHEREx launches  
in early 2025

NOTE—NASA IRTF - NASA Infrared Telescope Facility at Mauna Kea, Hawaii

ALMA - The Atacama Large Millimeter/submillimeter Array in the Atacama desert, northern Chile

VLT - ESO's Very Large Telescope in Chile

LDT - Lowell Discovery Telescope, Arizona, USA

Palomar 200" - The Hale telescope at the Palomar Observatory, California, USA

[https://www.lpi.usra.edu/sbag/documents/Apophis\\_SAT.pdf](https://www.lpi.usra.edu/sbag/documents/Apophis_SAT.pdf)

$\Delta v$  for spacecraft flyby missions

