

# solar orbiter

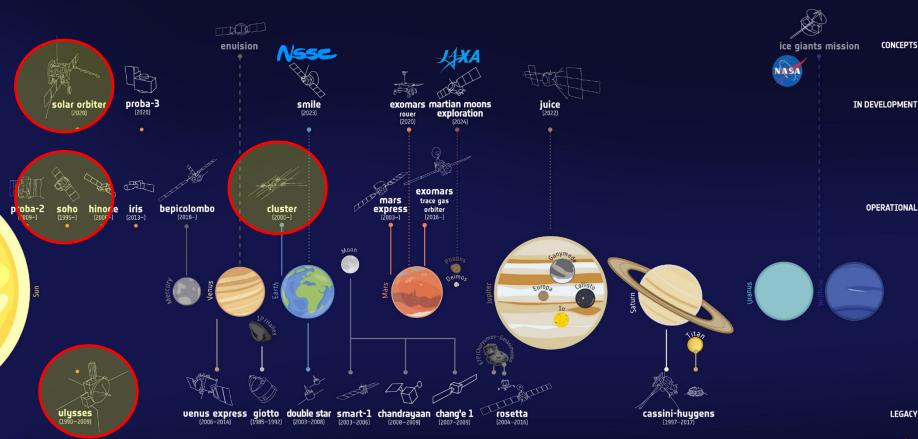
# → THE SUN UP CLOSE

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#### The ESA Fleet in the Solar System

#### → SOLAR SYSTEM EXPLORERS

#Space19plus



Space19 🧿

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#### Science objectives



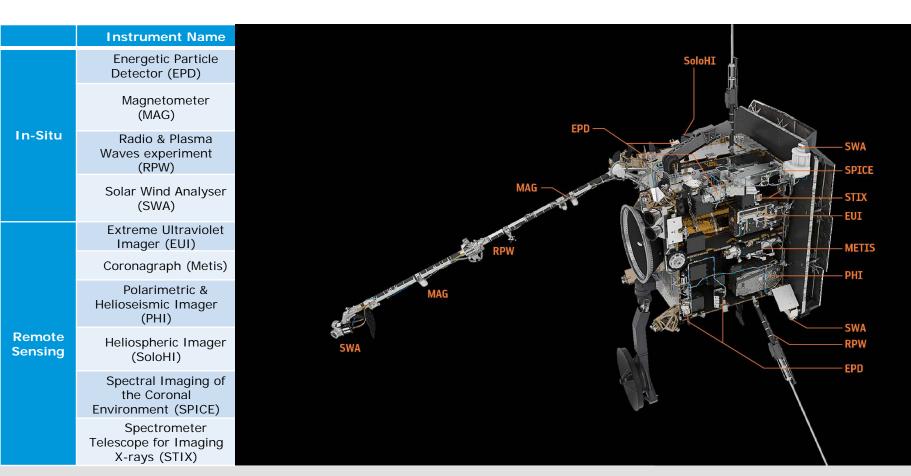
- Solar Orbiter addresses central questions concerning our Sun:
  - How does the Sun create and control the heliosphere?
  - What drives the solar wind?
  - □ Why does solar activity change over time?
- Closest approach 0.28 AU (42 million km within the orbit of Mercury) closest-ever images and following features at the surface
- Later in the mission, orbit change to a highly inclined orbit (up to 32° of solar latitude) first images of the poles
- Unique combination of in-situ and remote-sensing instruments correlate what we see and what we measure

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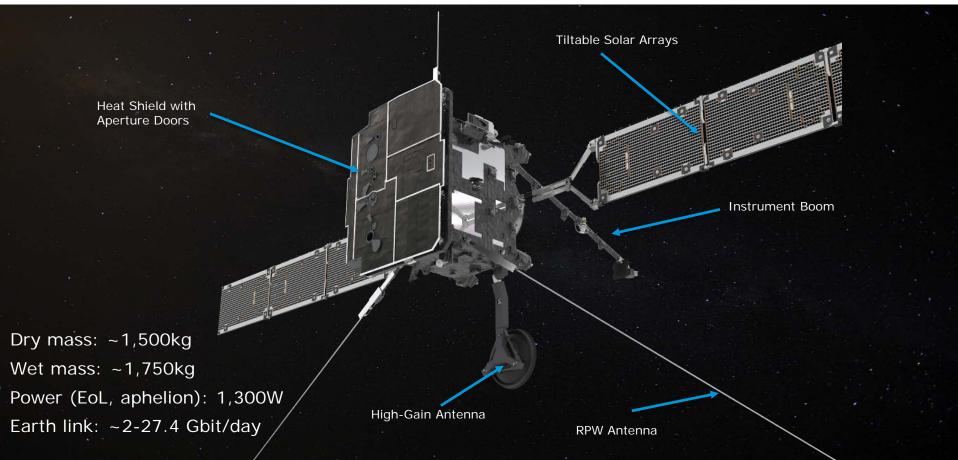
#### The scientific payload complement





#### Spacecraft overview

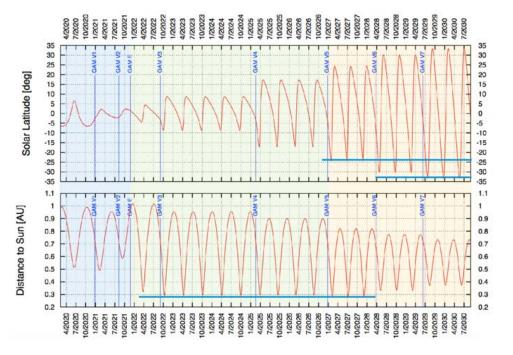




## Launch and trajectory

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- Launch from KSC, Florida, by NASA-provided Atlas V 411 in February 2020
- Direct injection escape trajectory to Venus, first acquisition expected at New Norcia ESA ground station
- Trajectory design driven by required planetary alignment for gravity assist maneuvers
- Highly elliptical orbit



Cruise phase: 1.8 years - Nominal mission: 4 years

Orbital period: 150-180 days

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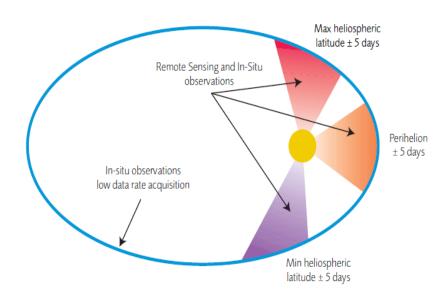
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#### Concept of operations

- Nominally sun pointing, with possibility to point at the limb for off-limb observations
- Telemetry-constrained
  - In-situ data continuously acquired
  - Remote-sensing operations planned in three 10-day Remote Sensing Windows per orbit
  - Planning cycles (long-term to very-shortterm)
  - Inter-instrument communication capability on board



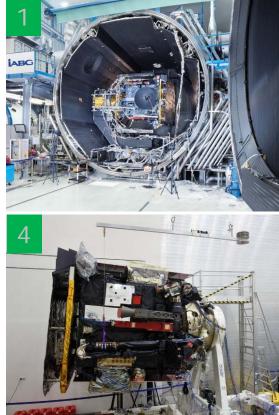


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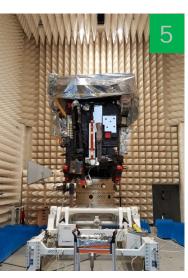
## Testing for the harsh space environment





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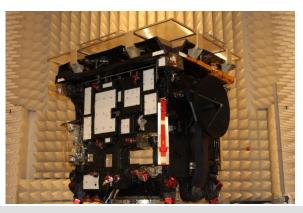




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# Engineering challenges: thermal

- Heat shield:
  - SolarBlack coating
  - □ 40 cm thick (high-T MLI, star brackets, low-T MLI)
  - □ Surface temperature between -200°C and +520°C
- SORA radiators and thermal straps
  - Stand-off from main structure to limit TED
  - Pyrolitic graphite: very high conductivity (1 W/K) for very low structural stiffness







# Engineering challenges: EMC, magnetic cleanliness and charging

- Specific design of electronic boards to minimize EMC emissions (and definition of EMC-quiet operations)
- Selection of non-magnetic materials
- Shielding of magnetic components
- Harness layout to avoid creating current loops
- Ensuring conductivity of all external surfaces



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# Engineering challenges: contamination control



- Stringent requirements on molecular contamination (UV instruments) and particulate contamination (coronograph and high-voltage instruments)
- Mitigations:

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- Selection of low-outgassing materials and systematic bake-out
- Purge system with high purity nitrogen
- Monitoring during Spacecraft TVAC
- Heat shield doors (+Metis cap), instrument doors
- Regular inspections and cleaning
- Extensive modelling of in-flight molecular contamination
- In-flight measurements by the Contamination Monitoring System (CMS)

#### CMS-1 on front panel of SPICE

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## An international enterprise



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- Payload and Spacecraft systems provided by European contributors from 17 ESA Member and Cooperating States
- Collaboration with NASA



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#### For more information and latest news



ESA general portal:

https://www.esa.int/Science\_Exploration/Space\_Science/Solar\_Orbiter

ESA science portal:

https://sci.esa.int/web/solar-orbiter

Solar orbiter's journey around the Sun: animation showing the orbital manoeuvres: <u>http://www.esa.int/spaceinvideos/Videos/2019/10/Solar\_Orbiter\_s\_journey\_around</u> <u>\_the\_Sun</u>

Twitter: @ESASolarOrbiter

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