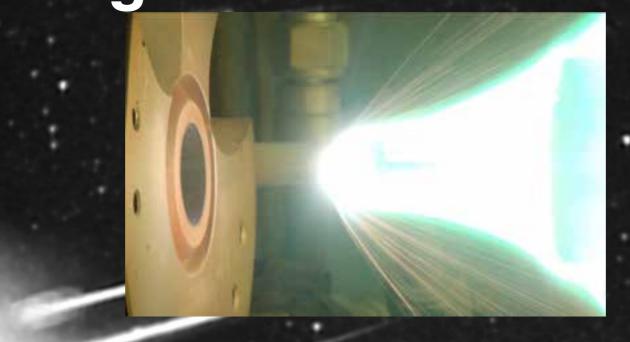
Artificial Meteor and Chelyabinsk Ablation Test using Arc-heated Wind Tunnel





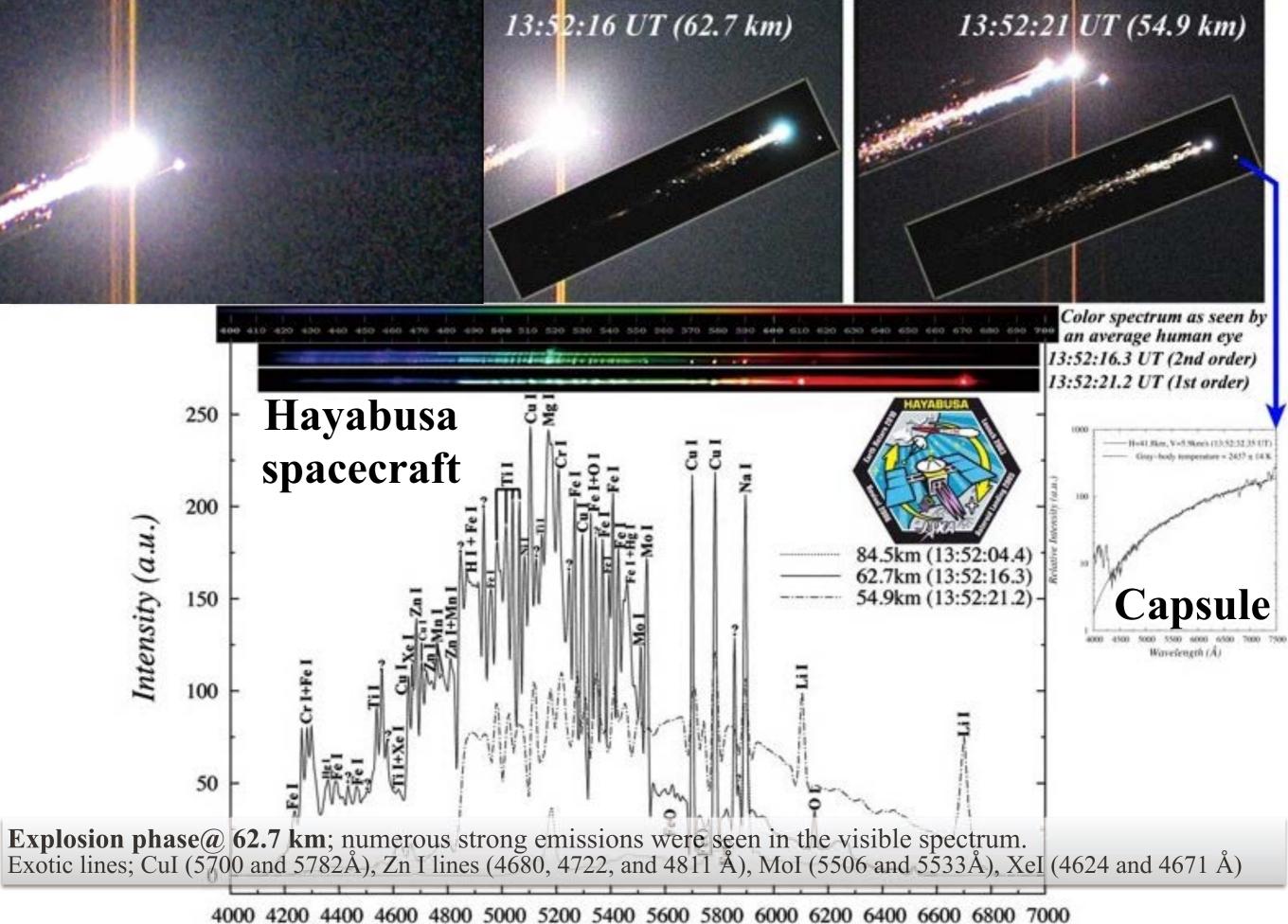
Collaborators;

Shinsuke ABE

Nihon University, Dept. Aerospace Engineering

K. Araki, T. Iwasaki, K. Toen (Nihon Univ.) Hironori Sahara (Tokyo Metropolitan Univ.)

Takeo Watanabe (Teikyo Univ.) Lena Okajima (ALE Co. Ltd.,)



4000 4200 4400 4600 4800 5000 5200 5400 5600 5800 6000 6200 6400 6600 6800 7000 Wavelength (Å)

S. Abe et al. 2011

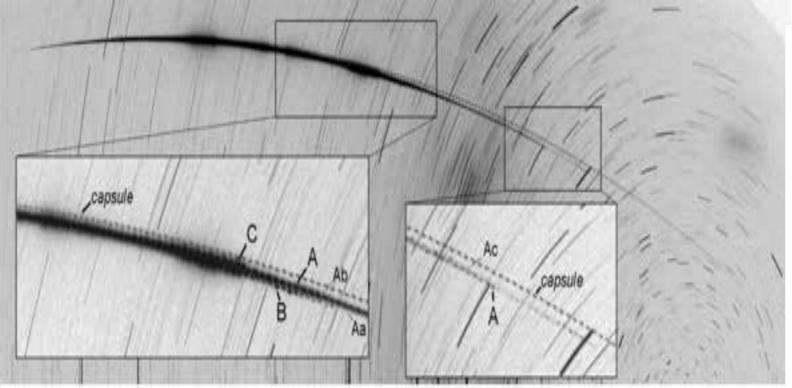
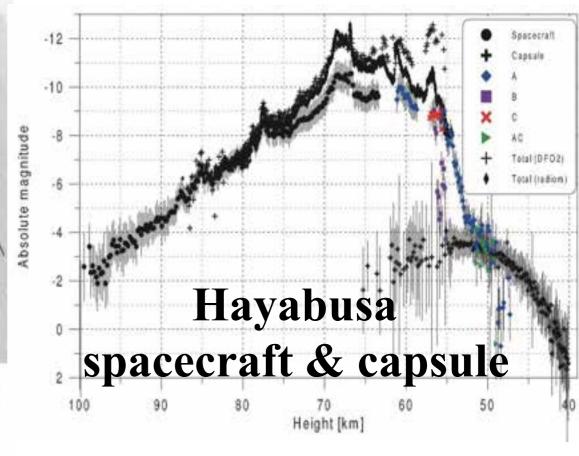


Fig. 2. Hayabusa re-entry as photographed from station GOS4. The exposure was 27 minutes long, from 13:37:00 to 14:04:00 UT. The fireball flew from left to right and was interrupted be the rotating shutter 10 times per second. The closest horizon lies upwards. The fragments mentioned in the text are identified in the insets.



- The dynamic pressures acting on the spacecraft at the fragmentation points were only 1–50 kPa
- No spacecraft fragment was seen to survive below a height of 47 km
- The integral luminous efficiency of the spacecraft was 1.3% and the capsule was 0.03%

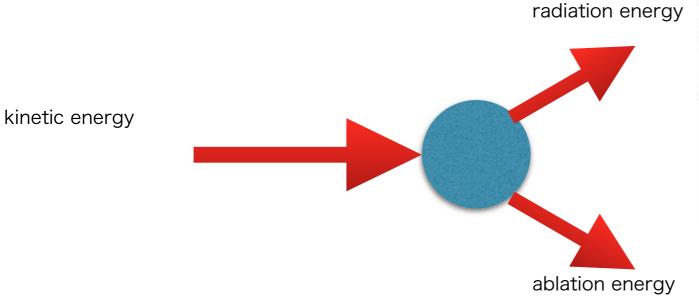


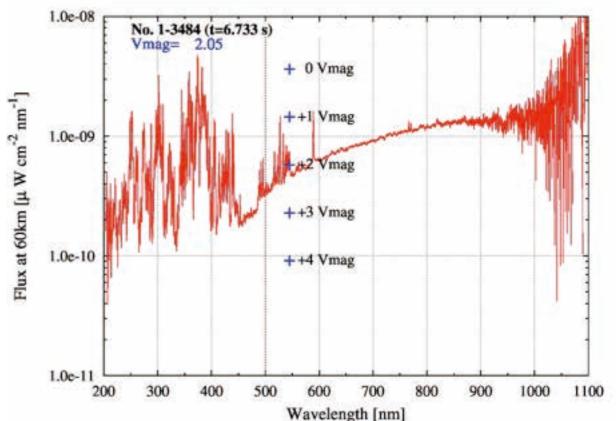
Eccentricity	1.32
Inclination	34.52°
Right ascension of the ascending node	7.58°
Pericenter distance	6310 km
Longitude of pericenter	255.58°
Heliocentric orbit (J2000.0)	
Semimajor axis	1.278 AU
Perihelion distance	0.9824 AU
Eccentricity	0.231
Inclination	1.59°
Argument of perihelion	145.63°
Longitude of the ascending node	82.360°

Borovická, Abe, Shrbený, Spurný, Bland, 2011

To understand ablation processes of atmospheric entry, artificial meteor test is carried out using JAXA's facility.

JAXA/ISAS Arc-heated Wind Tunnel High enthalpy conditions T~10,000K, V~6km/s, 0.6MPa



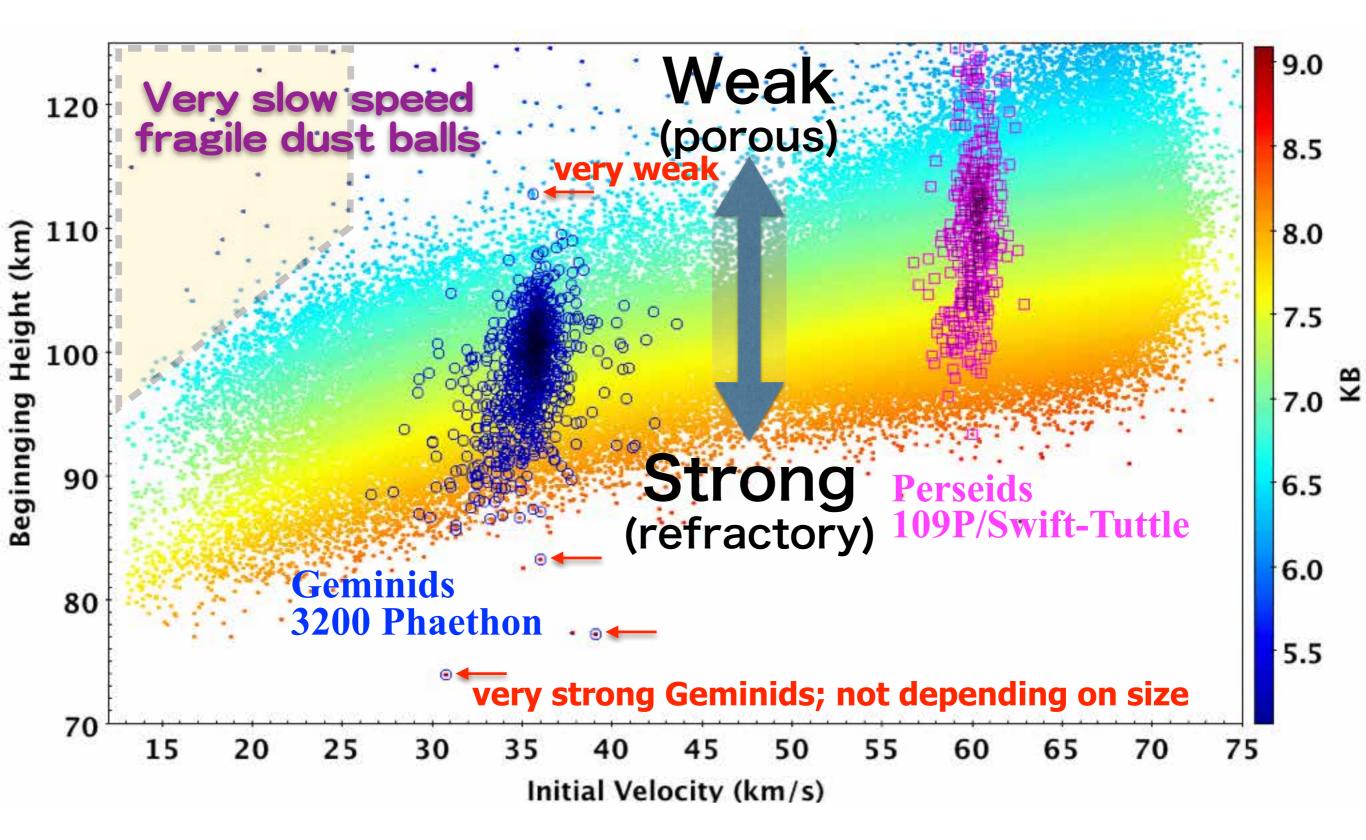


UV-VIS-NIR spectroscopy

$$\frac{dT_m}{dt} = \frac{A}{cm^{1/3}\rho_m^{2/3}} \left(\frac{C_h \rho_a V^3}{2} - 4\sigma \epsilon (T_m^4 - T_a^4) - L \frac{dm}{dt} \right)$$



Strength of Meteoroids by MU Radar Meteor head-echo

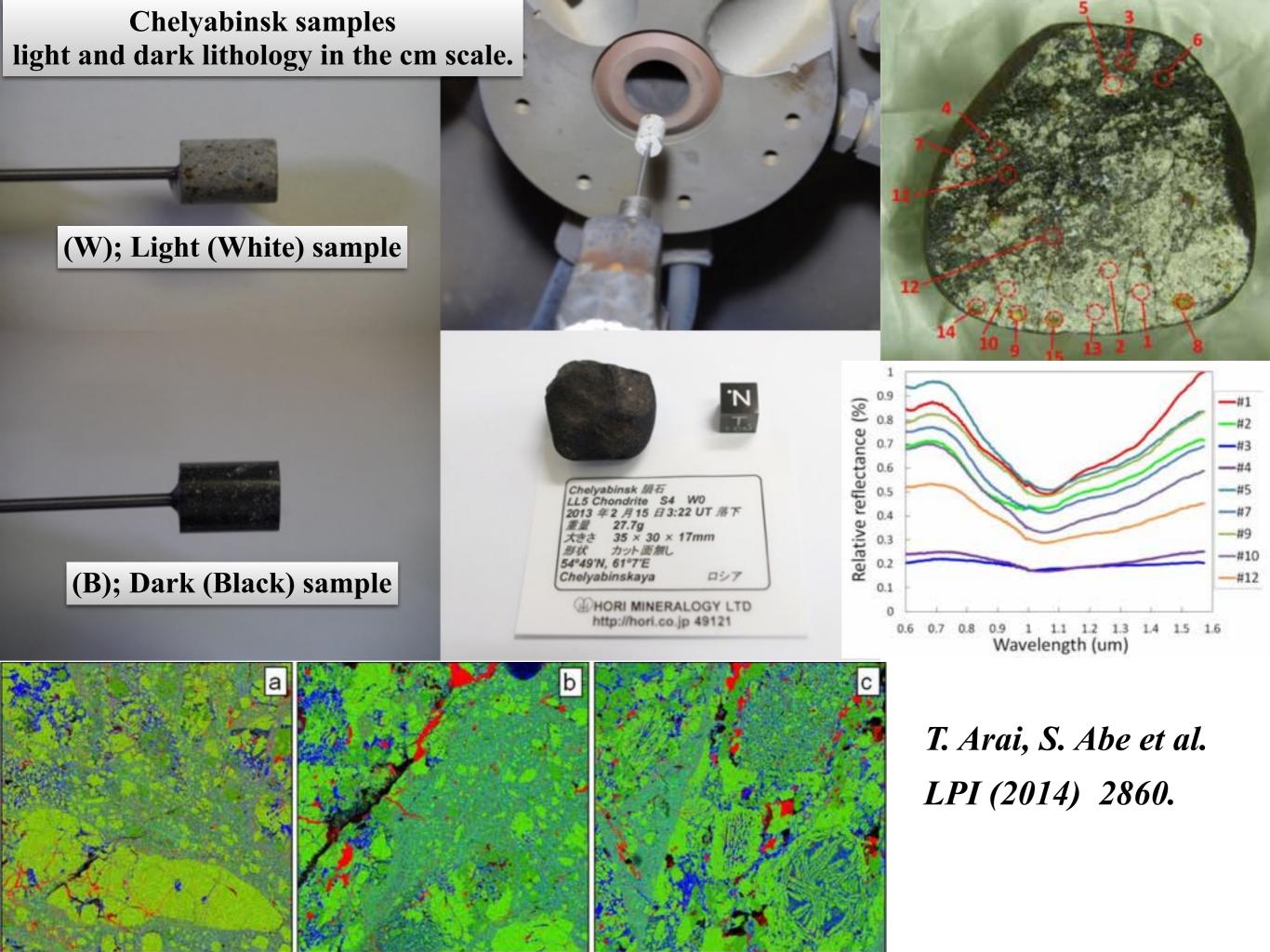


S. Abe, J. Kero, T. Nakamura et al. (in prep)



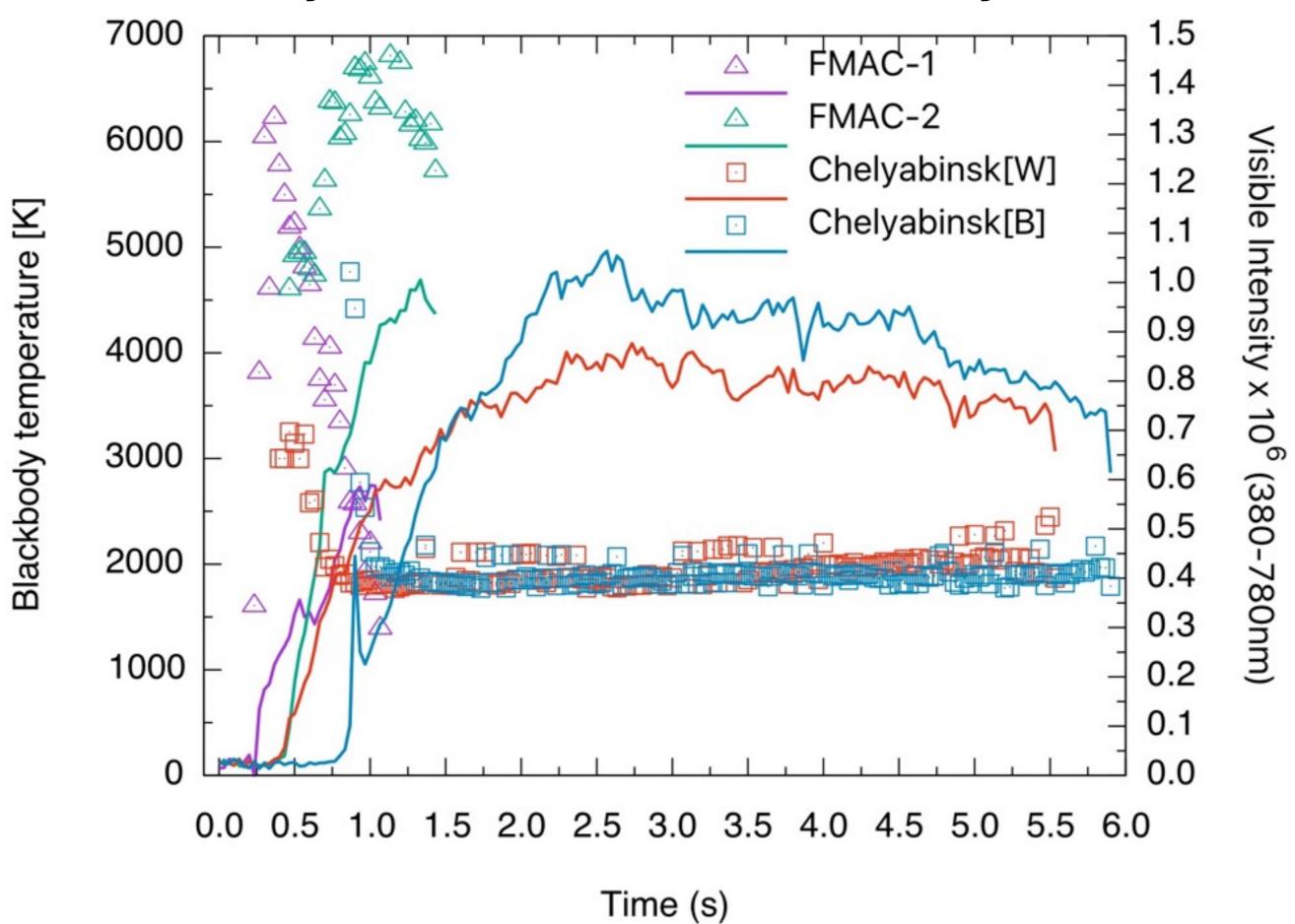
Chelyabinsk Meteorite; 2013/2/15

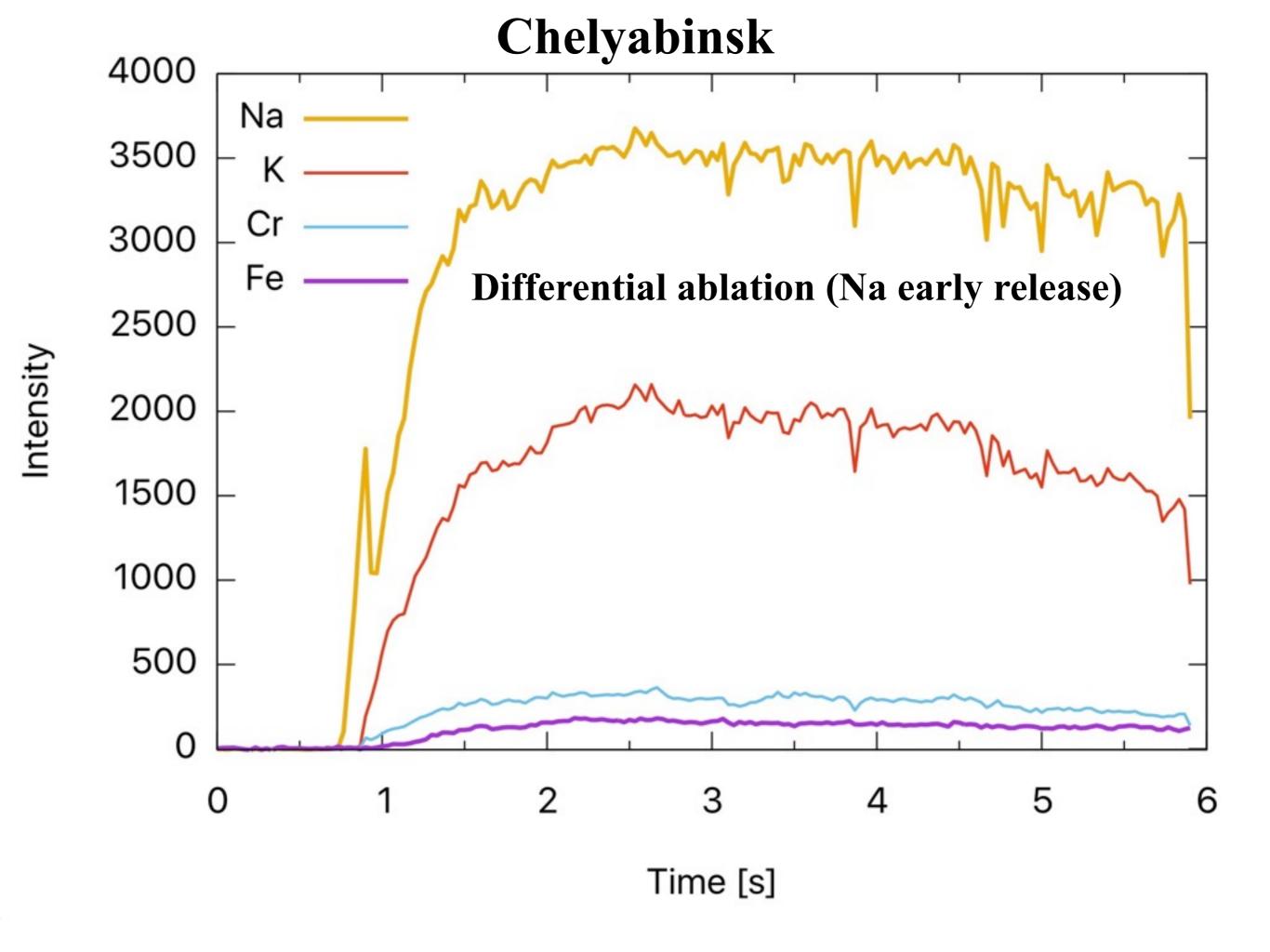


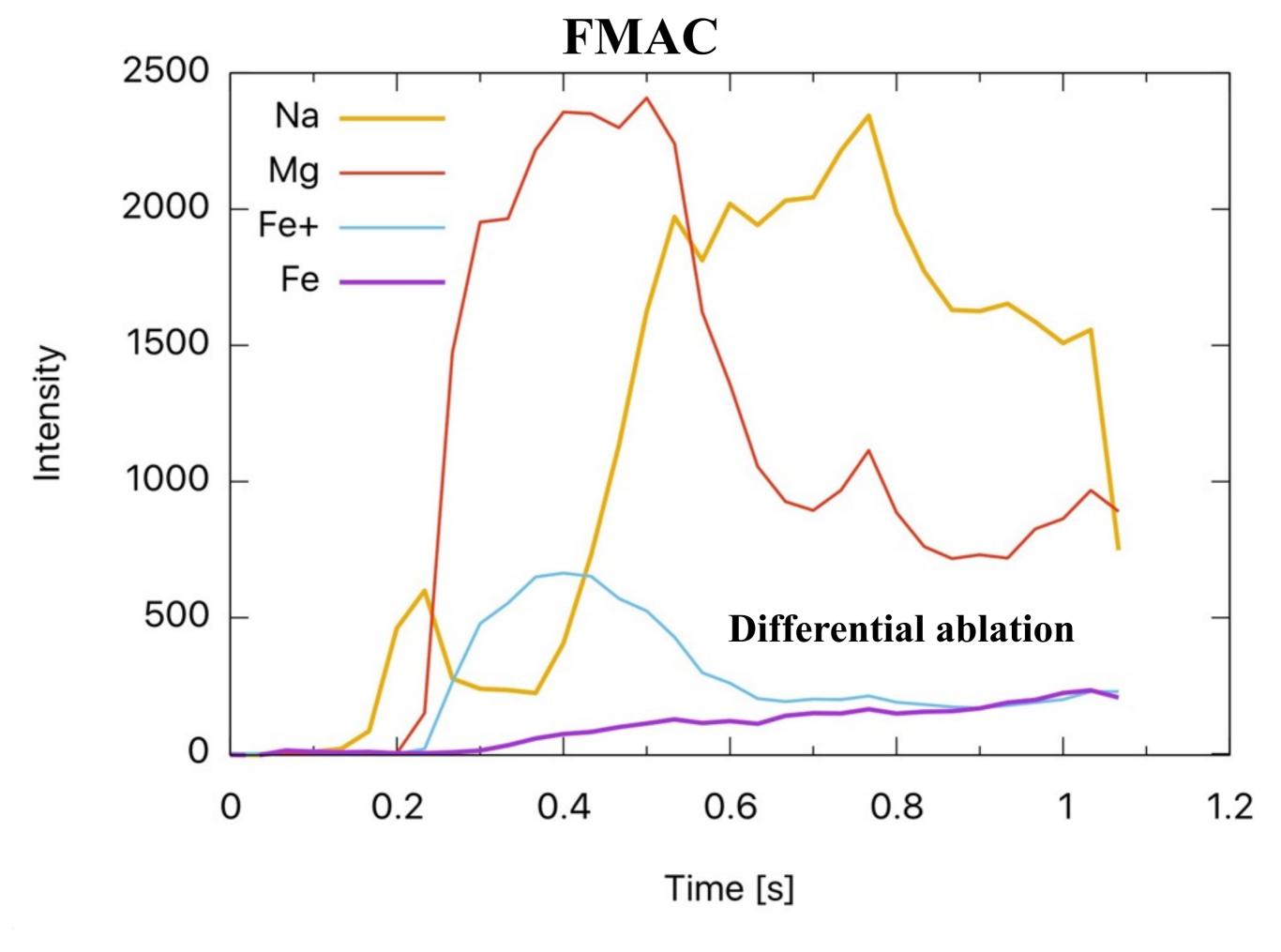




Blackbody is dominant for low-velocity meteors

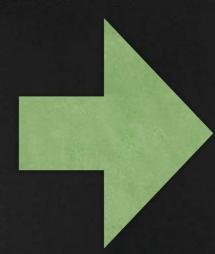








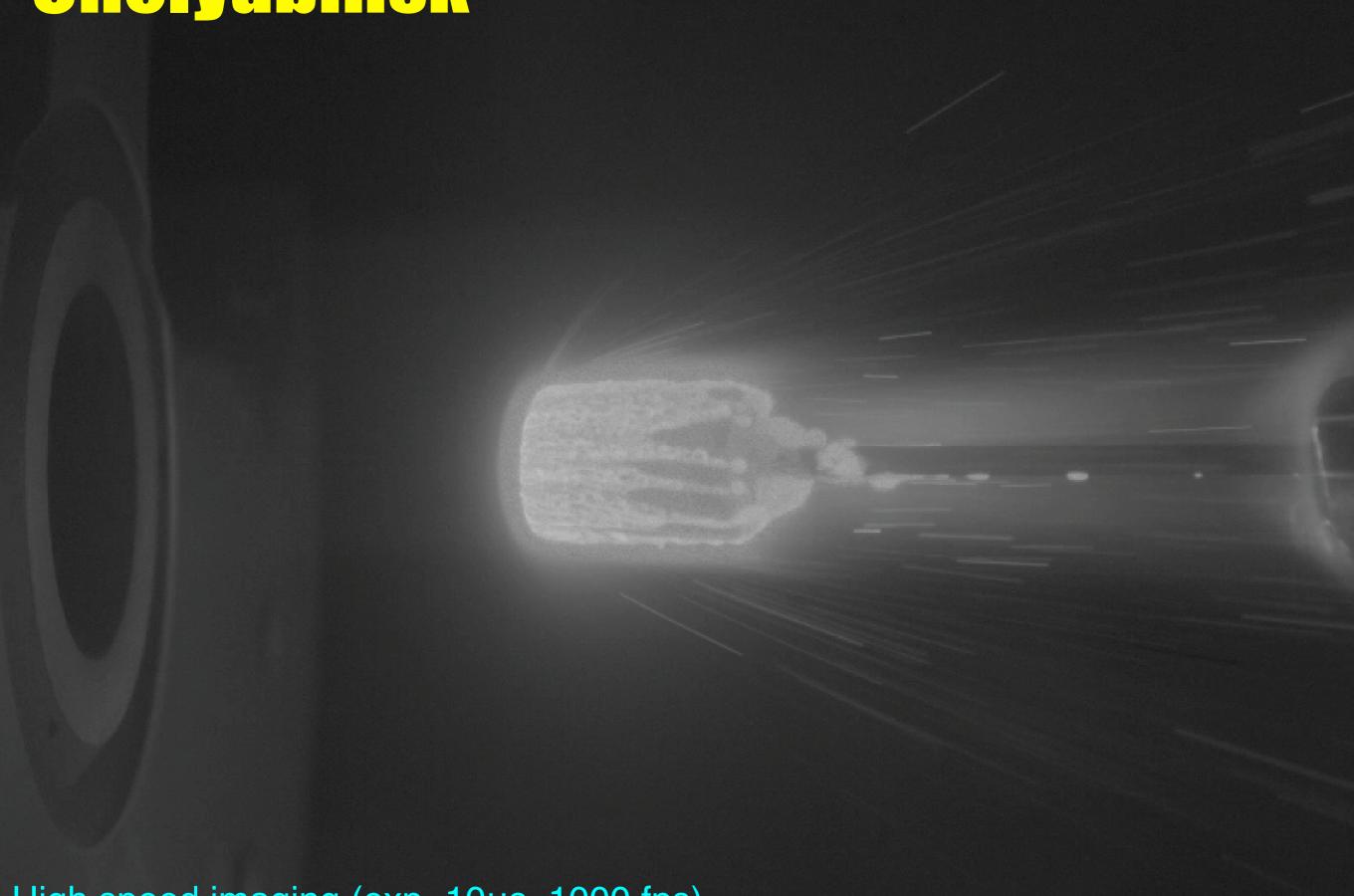
T~10,000 K, v~6km/s



Heating rate~ 30 MW/m^2

High speed imaging (exp=10µs, 1000 fps)

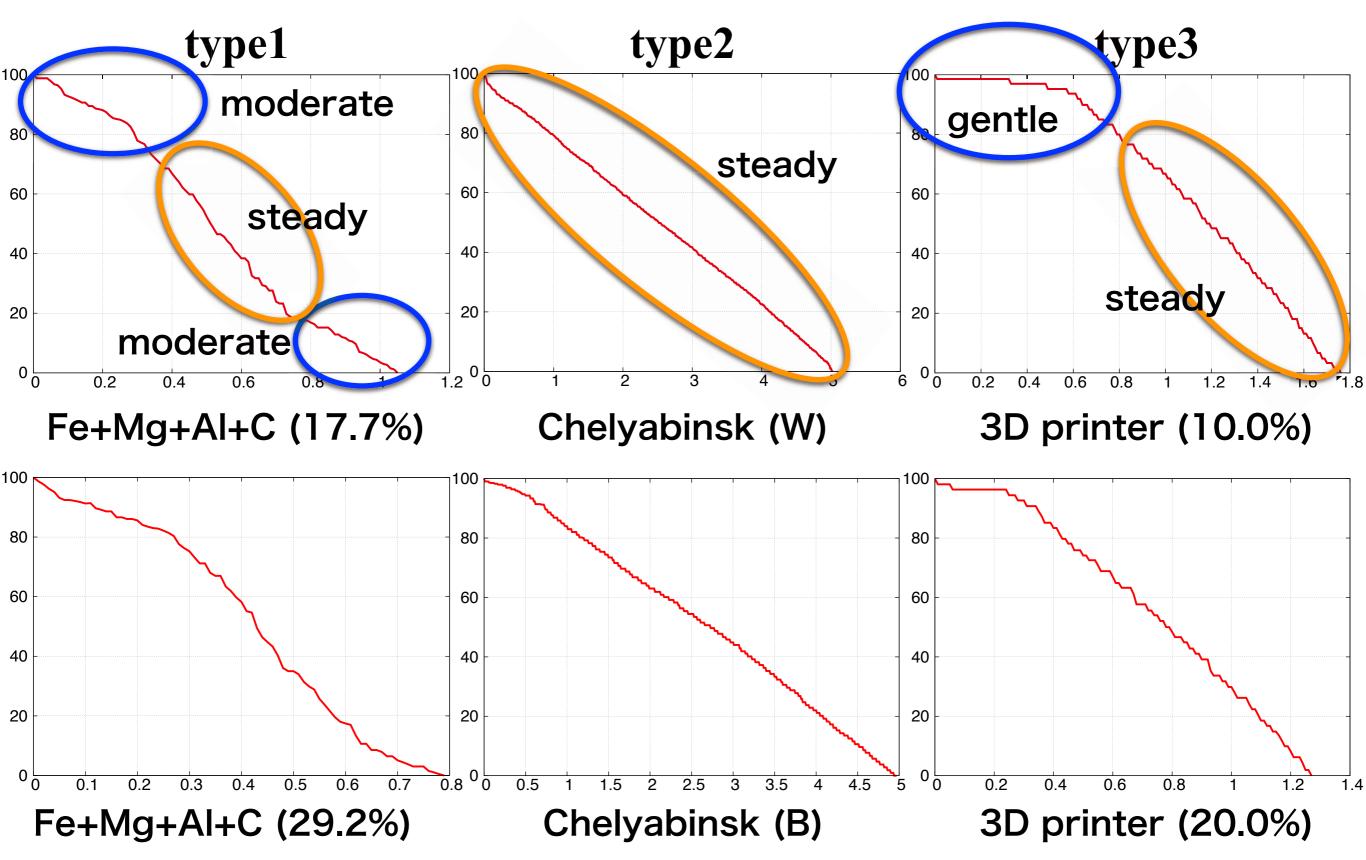
Chelyabinsk



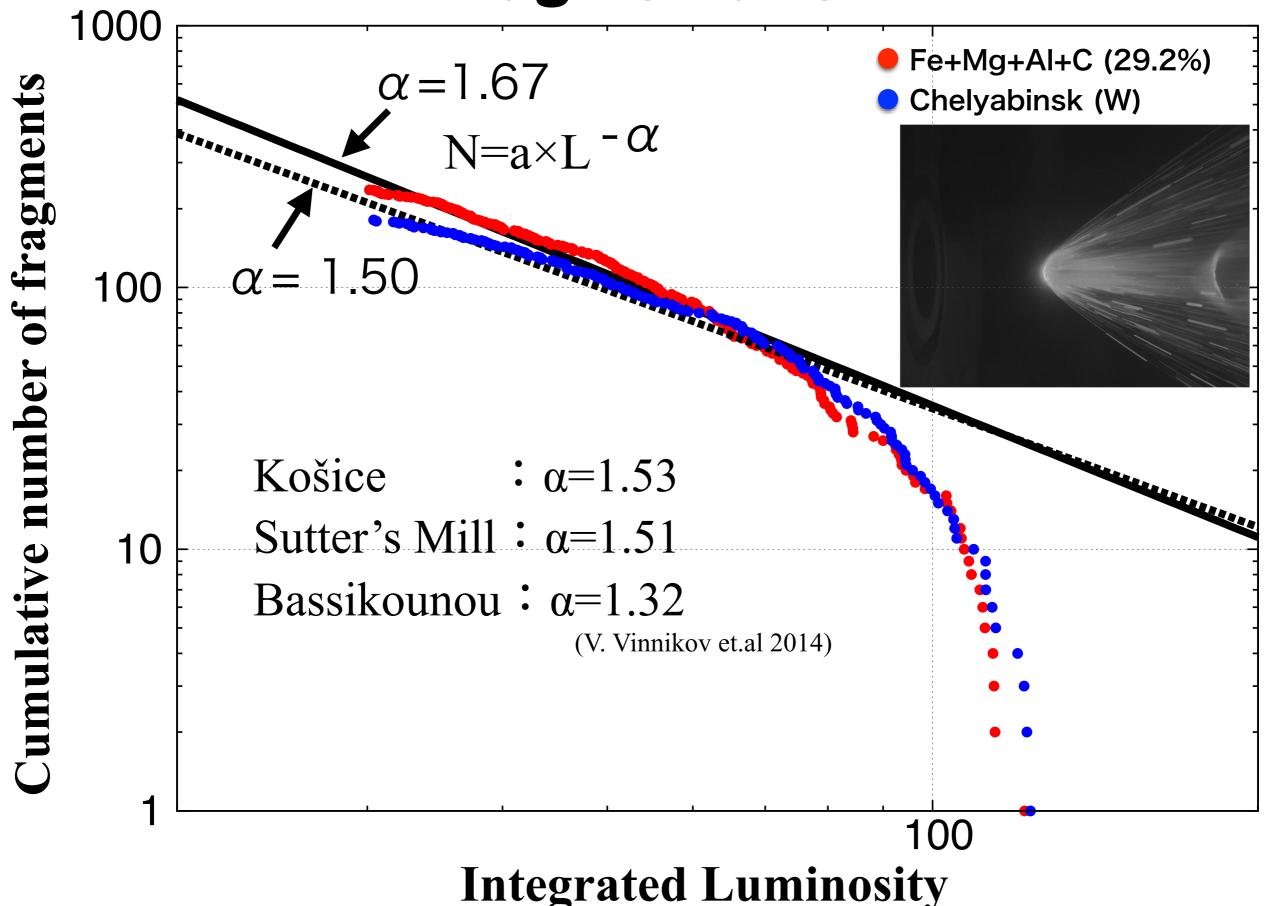
High speed imaging (exp=10µs, 1000 fps)

Mass Loss Rate, dm/dt

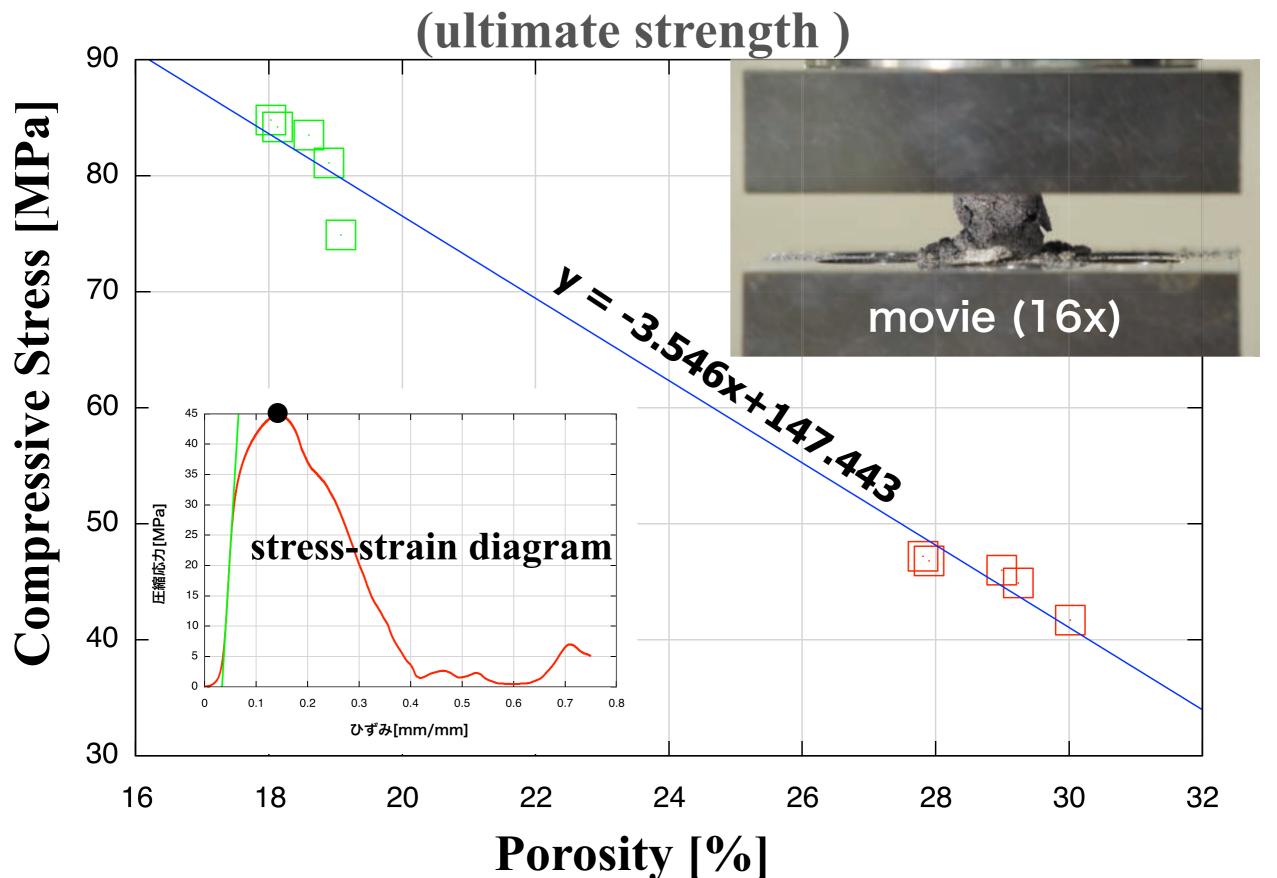
Y-axis: Rest of mass [%] X-axis: Time [s]

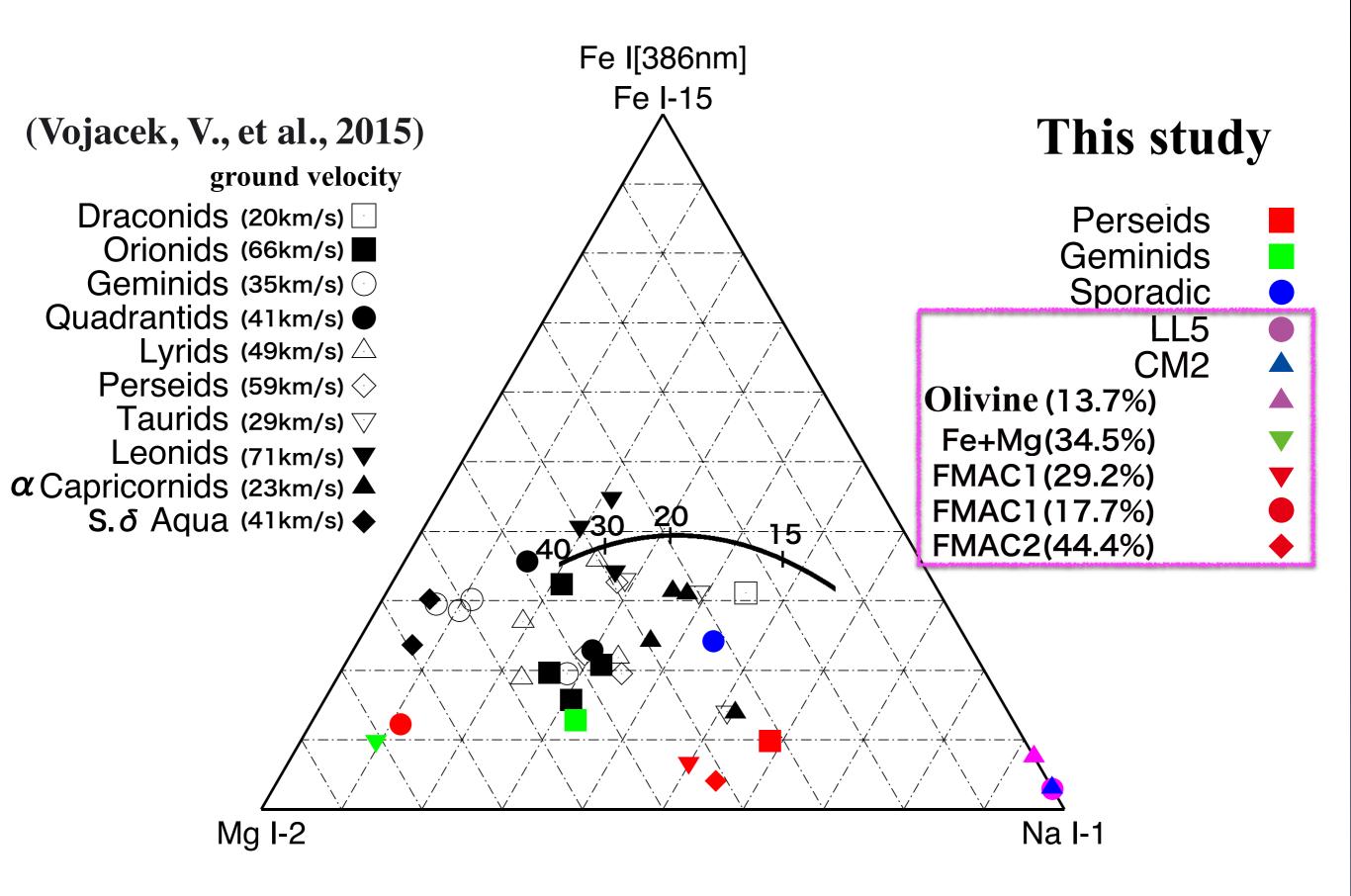


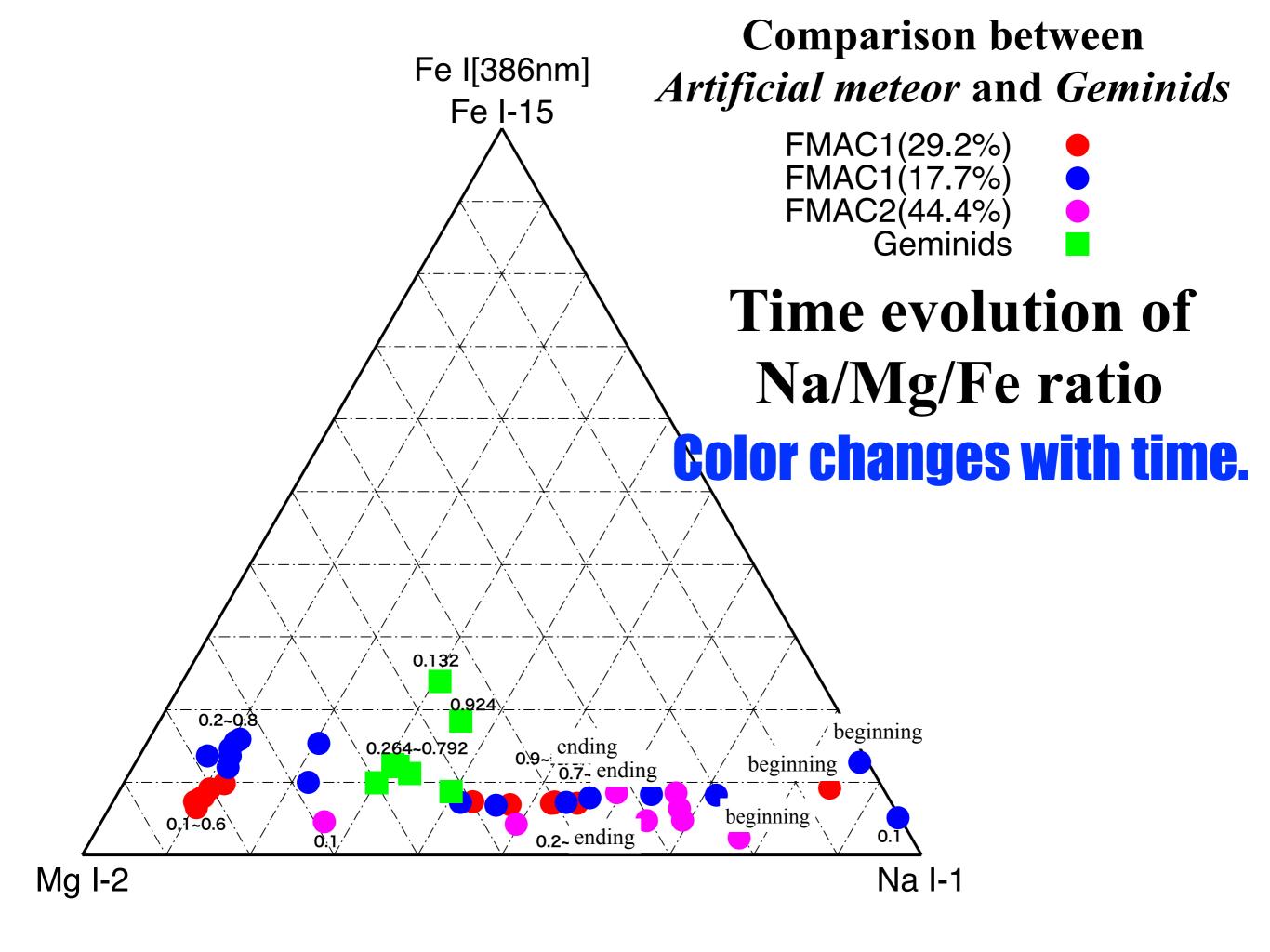
Fragmentation



Fe+Mg+Al+C (FMAC): Maximum compressive stress as a function of Porosity









Artificial Meteors for Business & Science ALE Co. Ltd., Tokyo Metropolitan Univ., Nihon Univ., Teikyo Univ.



PROJECT

TEAM

DEVELOPMENT

FUTURE

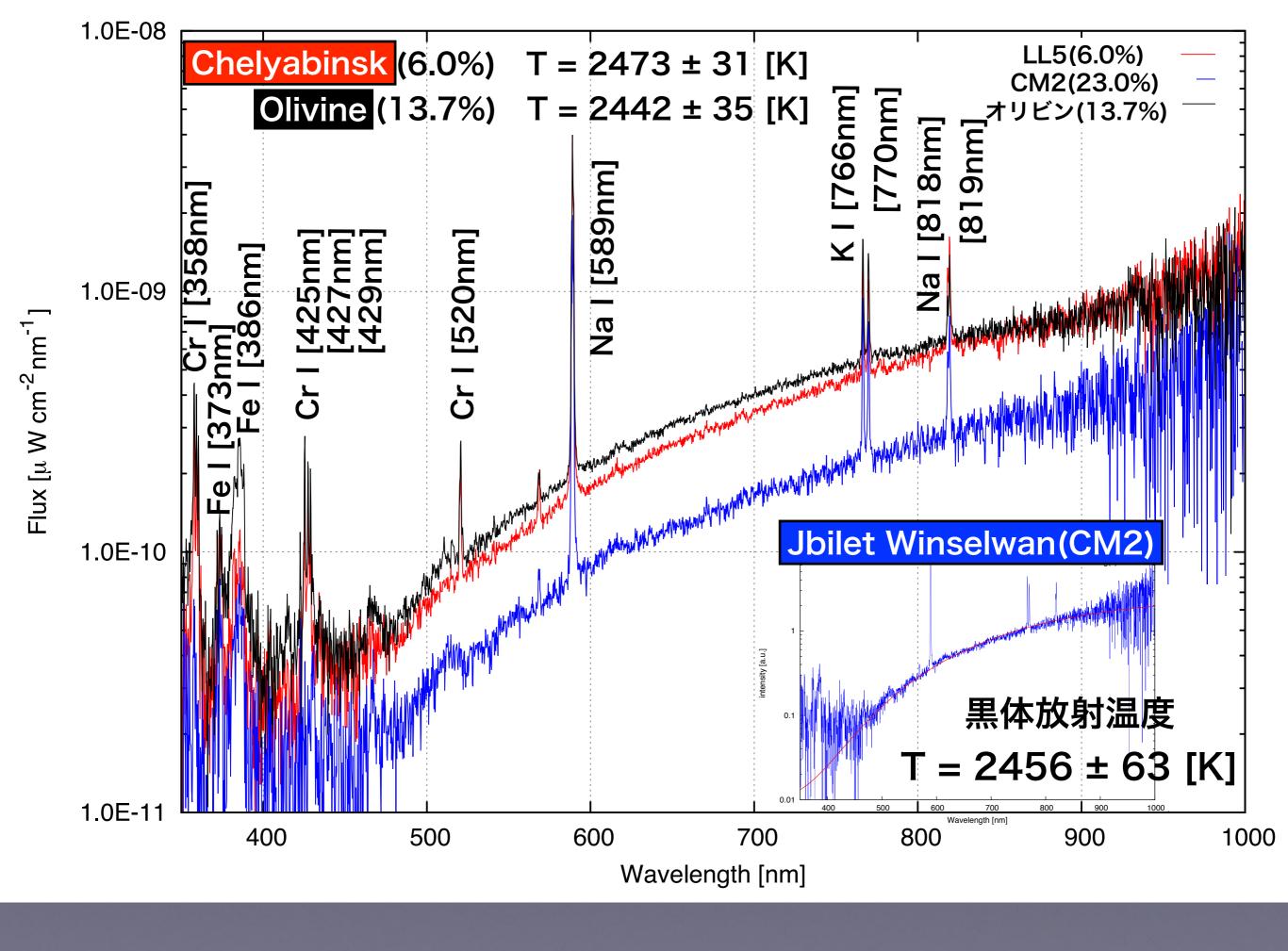
ALE Co., Ltd.



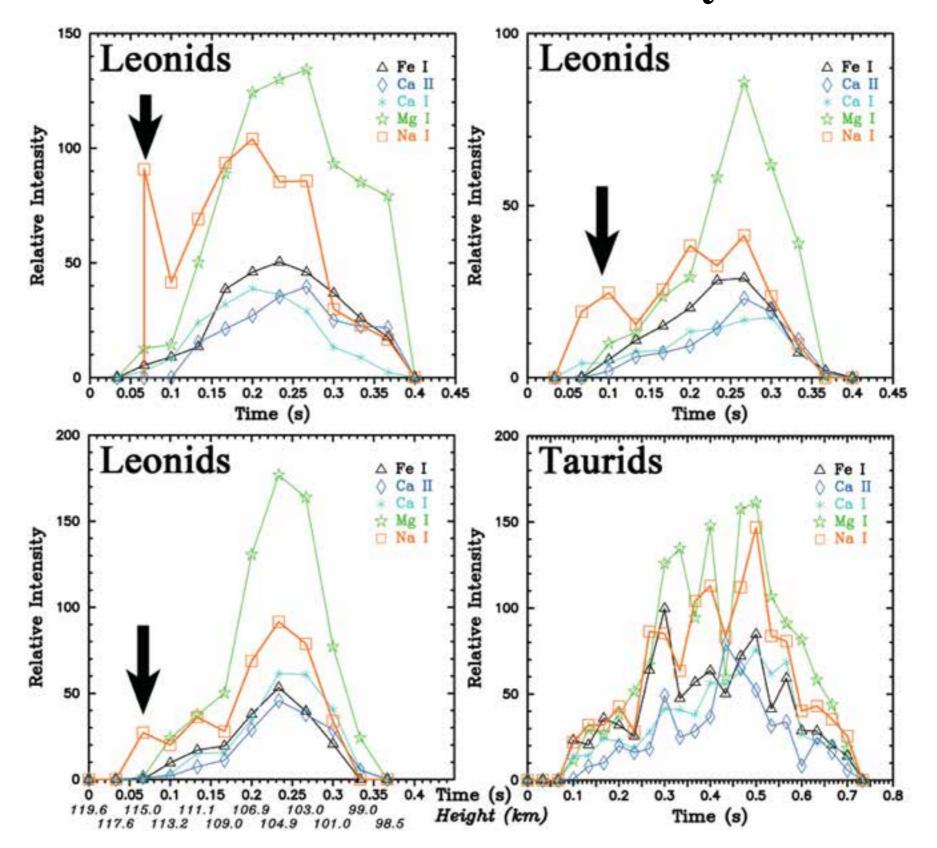
未来のエンターティンメント

Thank You

On-demand Meteor Shower First test in 2018



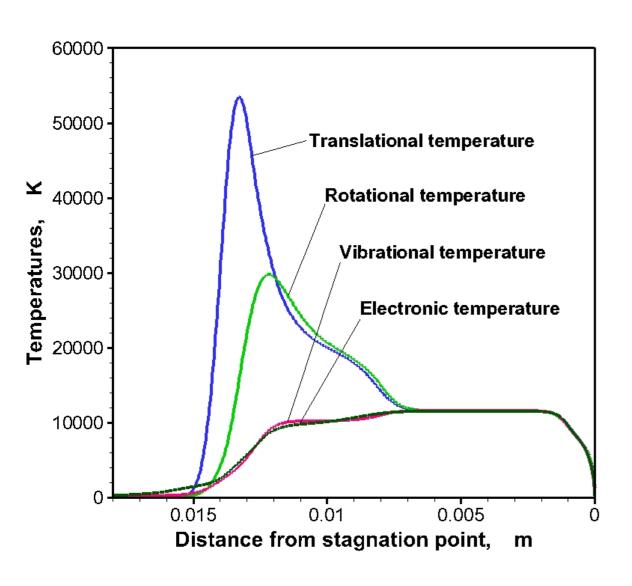
Differential ablation (Na early release) signature of structure & volatility of meteoroids

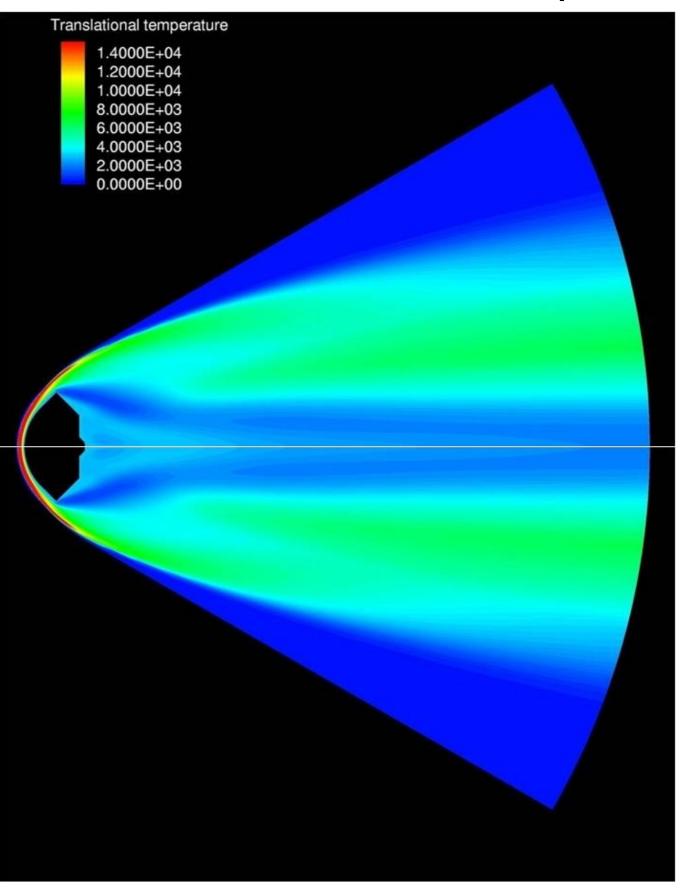


S. Abe (2009), Meteoroids and Meteors - observations and connection to parent bodies, Springer

CFD simulation of HAYABUSA capsule

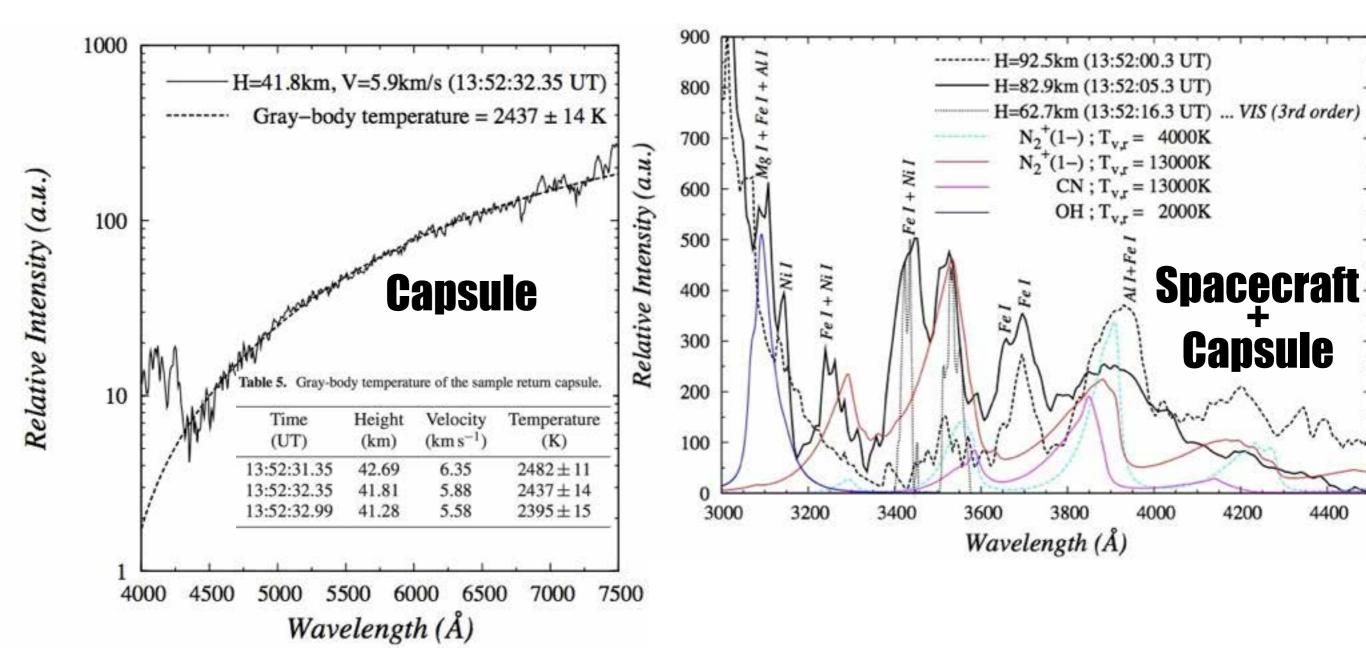
Vibration-rotation temperature of $\sim 13,000 \text{ K for } N_2^+ (1^-) \text{ and CN is reasonable.}$

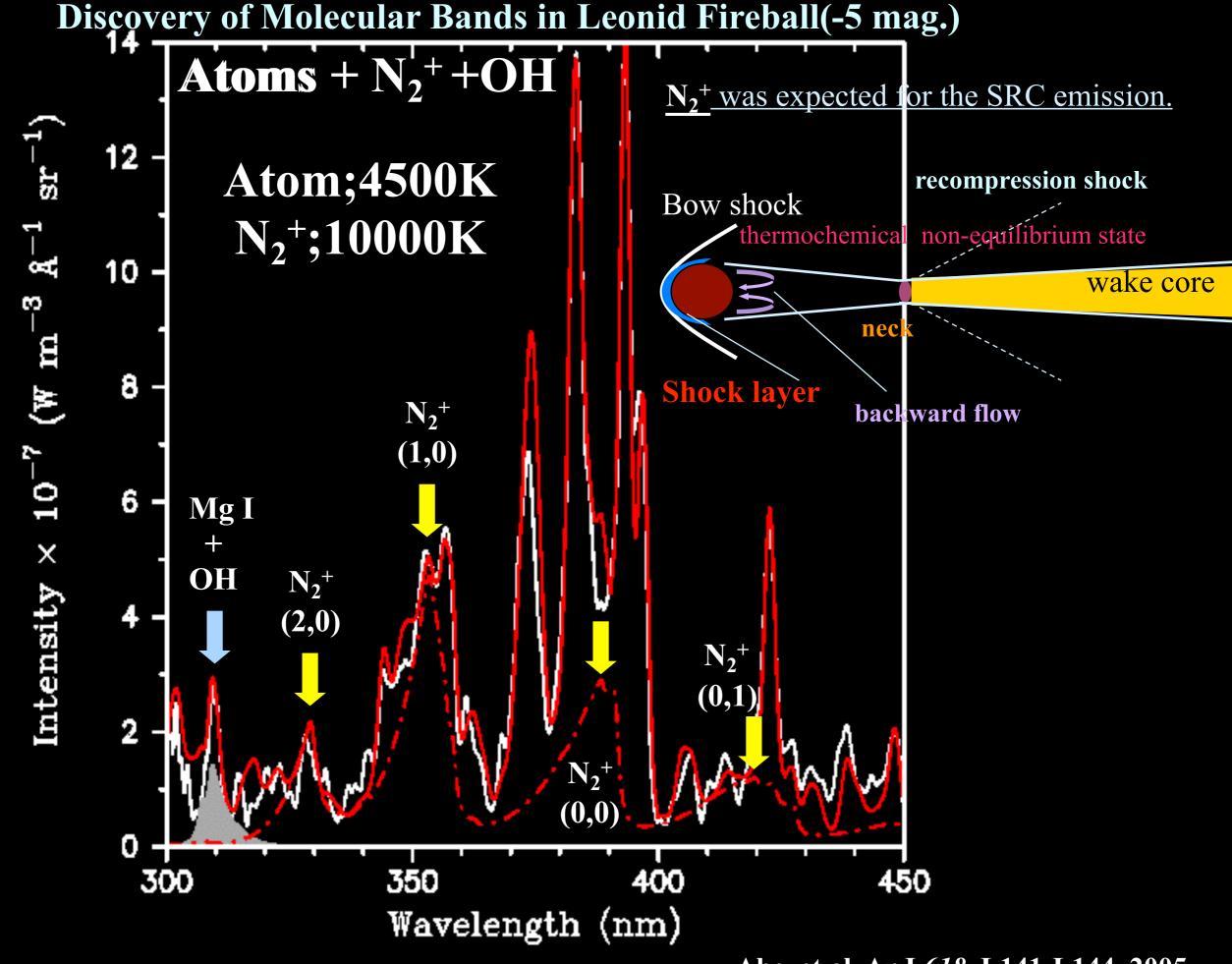




Fujita, Abe et al. (2013)

- The vibration temperature of molecular N_2^+ (1⁻) as dramatically changed from 4,000 K @ 92.5km to 13,000 K @ 82.9km.
- The observed spectra are a superposition of the post shock plasma radiation which is mixed with a shock layer heating and downward plasma. Thus, it is logical to understand that the high temperature region was induced by a shock layer of the spacecraft which rapidly grew between 92.5 km and 82.9 km in height.
- N₂⁺ (1⁻)bands originated from the spacecraft was much stronger than CN bands originated from the capsule in which Carbon was the major erosion product of the Carbon-Phenol heat shield of the capsule as seen by the Stardust capsule (Jenniskens 2010; Winter & Trumble 2011).





Abe, et al. ApJ 618, L141-L144, 2005.

Kyoto University, RISH

Middle and Upper Atmosphere Radar

Monostatic coherent pulse Doppler radar VHF (46.5 MHz), 1MW peak power, 475 crossed Yagi antennas Pulse length: $1-500\mu$ s, Antenna aperture: 8330m^2 (D=103m)

