Evaluation of disturbances due to test mass charging for LISA and LISA Pathfinder

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Talk Outline

Electrostatic analyses (LTP GRS) - FE model of entire GRS - FE CM submodels + Comparison to // plate LORENTZ Charging COULOMB Disturbance **Estimates LISA & LISA PF** Solar Max & Min Method to Cope with Charging

Why do TMs charge up?

• TMs can charge up due to:



More in later talks

Why does charging matter?

Charging of TMs disturbs their geodesic motion



- Unwanted forces from:
 - Coulomb interactions with surrounding conductors of GRS

Charge-dependent Coulomb accn:



=> Terms $\sim \Delta \mathbf{k}$

=> Terms~∆V & V∆k

 Lorentz interactions as it moves through magnetic fields

Lorentz accn:

$$\boldsymbol{a}_L = (Q/m) \boldsymbol{V} \times \boldsymbol{B}$$



Main charging Disturbances

- 1. Acceleration Noise (from fluctuations in: charge, TM position & velocity, voltages, magnetic field)
- 2. Modification of stiffness (from position dependence)
- 3. Coherent Fourier components (as Q(t) ~t)



Electrostatic FE model of LTP GRS

Coulomb accn is capacitance and hence geometry dependent

$$a_{Qk} = \frac{Q^2}{2C_T^2 m} \frac{\partial C_T}{\partial k} + \frac{QV_T}{mC_T} \frac{\partial C_T}{\partial k} - \frac{Q}{mC_T} \sum_{i=1}^{N-1} V_i \frac{\partial C_{i,N}}{\partial k}$$

Determine energy in system for given geometry & voltage distribution using FE analysis (ANSYS software)

Change V distribution => capacitances between specific conductors
 Change TM position => capacitance derivatives

Move in 1 DOF, Sym => $C_T(k) \approx C_T(k=0) + \frac{1}{2} \frac{\partial^2 C_T}{\partial k^2} k^2$	ANSYS (1 σ errors#~ 1%)	% > than // plate
"YZ-injection" sensor design* (ITEADELTRODEEAA design)	$C_T(k=0) = 33.7 \pm 0.2 \mathrm{pF}$	30%
(LISA PFLIP GRS EM design)	$\frac{\partial^2 C_T}{\partial x^2} = 1.15 \pm 0.01 \mu \text{F}/\text{m}^2$	21%
46mm side TM	$\frac{\partial^2 C_T}{\partial y^2} = 2.50 \pm 0.03 \mu\text{F}/\text{m}^2$	21%
Gaps: X face: 4 mm sens Y face: 2.9 mm sens, 4 mm inj Z face: 3.5 mm sens, 4 mm inj Z face: 3.5 mm sens, 4 mm inj	$\frac{\partial^2 C_T}{\partial z^2} = 1.62 \pm 0.03 \mu\text{F}/\text{m}^2$	27%
* B Weber et al. '03 SPIE	#Shaul & Sumner '04 CNME	



• For either set of features, $C_T \& \frac{\partial^2 C_T}{\partial z^2}$ unchanged to within 0.45 & 1.25 respectively

Submodel CM => estimate CM influence more accurately

- Positions of borders chosen as compromise between model size & distance from RoI
 => minimise effect on couplings of interest.
- BCs were not derived from the full model => avoid confounding models' accuracies



- Figures represent the properties of CM, NOT increase in capacitance when CM added as not compared "no CM" case
- Other capacitances will be affected by the inclusion of CM.
- These figures indicate CM level of influence.

Comparison of CM to full sensor

- Total $C_{TM,CM} < 3\% C_T$ for both types of stopper
- Total $\frac{\partial^2 C_{TM,CM}}{\partial z^2}$ < 4% & 9% $\frac{\partial^2 C_T}{\partial z^2}$ for CM with bbg stoppers & elongated stoppers
 - Only 0.7% from plungers
 - In dC_T/dk, this gets multiplied by TM offset from the centre of opposing features

 $a_{Qk} = \frac{Q^2}{2C_T^2 m} \frac{\partial C_T}{\partial k} + \frac{QV_T}{mC_T} \frac{\partial C_T}{\partial k} - \frac{Q}{mC_T} \sum_{i=1}^{N-1} V_i \frac{\partial C_{i,N}}{\partial k} = \gamma \text{ Terms} \sim \frac{\partial^2 C_T}{\partial z^2} \Delta z \Rightarrow \text{ requirements of the retracted CM}$

(E.g. TM offset of 50μm from the centre of the plungers => contribution of ~3% of that when the TM is offset 10μm wrt entire sensor)

 $\partial C_{TM,CM}$

 $\partial z < 9\%$, 6% & 14% of that of a single *z*-face, sensing electrode for a pyramidal plunger, a bbg stopper & an elongated stopper

$$a_{Qk} = \dots - \frac{Q}{mC_T} \sum_{i=1}^{N-1} V_i \frac{\partial C_{i,N}}{\partial k} \implies \text{Terms} \sim \frac{\partial C_i}{\partial z} \Delta V_i$$

- => Need to maintain a high level of uniformity in the CM surfaces, to minimise work function differences, as patch effects could multiply these gradients into significant effects
- Similarly, to minimise the CM contribution to stiffness, the mean voltage of opposing features should also be minimised.

Magnitude of Charging Disturbances (using ANSYS results for

<u>sensitive x-axis)</u>

Charging Rates

	LISA, solar min	LISA, solar max	LISA PF, solar min	LISA PF, solar max
Net charging rate (GEANT GCR) (+e/s)	100	47	88	<i>43</i>
"Worst case" net charging rate assumed (+e/s)	<i>191</i>	<i>99</i>	176	<i>93</i>
Effective charging rate, R _{eff} (+e/s)	708	462	746	469

- Assume "worst case" charging conditions:
 - + error margin of ~30% on the GEANT GCR charging rates

+ 60 +e/s for kinetic low-energy secondary electron emission (may ~cancel in the actual sensor)

 Rate of SEP events expected to be low enough that data acquisition could be suspended for their duration, but needs to be verified

• [$\delta Q (CHz^{-0.5}) = (2R_{eff})^{0.5}e/2\pi f$]

LISA: Araujo H *et al* submitted to Astroparticle Physics, arXiv:astro-ph/0405522. LISA PF: Wass P *et al* in preparation.

<u>Coherent Charging Signals</u>

For f ~ 10 ⁻⁴ Hz ⇔ estimates largest (τ = 1 yr; T = 1day)	LISA, solar min	LISA, solar max	LISA PF, solar min	LISA PF, solar max
e _x (t) (S/N)	4,833.57	2,498.52	4,447.51	2,351.47
f _x (t) (S/N)	850.87	227.35	720.38	201.38
l_(t) (S/N)	0.57	0.30	0.53	0.28

S/N is the ratio of the charging signal to the acceleration noise budget for LISA



$$e_{k}(t) \equiv \Theta_{k}(t) \equiv \frac{\overline{\dot{Q}}V_{T}t}{mC_{T}} \frac{\partial C_{T}}{\partial k} - \frac{\overline{\dot{Q}}t}{mC_{T}} \sum_{i=1}^{N-1} V_{i} \frac{\partial C_{iN}}{\partial k}$$
$$f_{k}(t) \equiv \Xi_{k}(t)^{2} \equiv \frac{\overline{\dot{Q}}^{2}}{2C_{T}^{2}m} \frac{\partial C_{T}}{\partial k} t^{2}$$
$$l_{x}(t) \equiv \Phi_{x}(t) \equiv \left(\eta \overline{\dot{Q}}t / m \left[\overline{V}_{I} \times \overline{B}_{IP}\right] \cdot \hat{x}$$

Coulomb signals dependent on: $\Delta k (10 \mu m), \Delta V (1mV), V (100 mV), V_T (100 mV)$ Lorentz signal dependent on mean B_{IP} (~ 0.25nT -> ~ 40nT; median ~6nT). Even estimates @ 40nT < accn noise) (η = 0.01) Exact spectral shape depends on discharging scheme & variability of e.g. mean charging rate (Shaul et al, CQG '04)

Stiffness: GRS Requirement: -2x10-8 -> 8x10-8 s-2

	LISA, solar min	LISA, solar max	LISA PF, solar min	LISA PF, solar max	<u>Withir</u> <u>Limit?</u>
Stiffness, s _{Qx} (s ⁻²)	-1.11E-08	-5.27E-09	-1.00E-08	-4.93E-09	\checkmark



Coulomb Noise

For f ~ 10 ⁻⁴ Hz (Charge noise~1/f)	LISA, solar min	LISA, solar max	LISA PF, solar min	LISA PF, solar max
Displacement noise (ms ⁻² Hz ^{-0.5})	1.70E-17	8.54E-18	1.55E-17	8.02E-18
Charge noise (ms ⁻² Hz ^{-0.5})	3.97E-16	3.07E-16	4.04E-16	3.09E-16
Voltage noise (ms ⁻² Hz ^{-0.5})	4.62E-16	2.39E-16	4.25E-16	2.25E-16



<u>Lorentz Noise</u>

For f ~ 10 ⁻⁴ Hz (Lorentz noise~1/f ^{1.6})	LISA, solar min	LISA, solar max	LISA PF, solar min	LISA PF, solar max
Lorentz noise (ms ⁻² Hz ^{-0.5})	1.22E-16	6.32E-17	1.12E-16	5.95E-17

$$\overline{B}_{IP} = 3 \times 10^{-8} \text{ T}$$
 $\delta B_{IP} = 3 \times 10^{-7} \text{ THz}^{-0.5}$ $\overline{B}_{SC} = 8 \times 10^{-7} \text{ T}$ $\delta B_{SC} = 1 \times 10^{-7} \text{ THz}^{-0.5}$

Lorentz accn: $a_{L} \approx \left(\eta \overline{Q} t \overline{V}_{I} \times \overline{B}_{IP} + \eta \overline{Q} t \overline{V}_{I} \times \delta B_{IP} + \eta \overline{Q} t \delta V_{I} \times \overline{B}_{IP} + \overline{Q} t \delta V_{SC} \times \overline{B}_{SC} + \eta \delta Q \overline{V}_{I} \times \overline{B}_{IP} \right) / m$ Fluctuations Fluctuations in Charging in B_{IP} TM velocity "shot noise" Dominant term by > 3 orders of magnitude so ~ insensitive to range of B_{IP}

Magnitude of Charging Disturbances

	LISA, solar min	LISA, solar max	LISA PF, solar min	LISA PF, solar max	With Limit
Net charging rate (GEANT GCR) (+e/s)	100	47	88	<u>43</u>	
"Worst case" net charging rate assumed (+e/s)	<i>191</i>	<i>99</i>	176	<i>93</i>	
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e _x (t) (S/N)	4,833.57	2,498.52	4,447.51	2,351.47	
f _x (t) (S/N)	850.87	227.35	720.38	201.38	
I _x (t) (S/N)	0.57	0.30	0.53	0.28	
Stiffness, s _{Qx} (s ⁻²)	-1.11E-08	-5.27E-09	-1.00E-08	-4.93E-09	\checkmark
Displacement noise (ms ⁻² Hz ^{-0.5})	1.70E-17	8.54E-18	1.55E-17	8.02E-18	
Charge noise (ms ⁻² Hz ^{-0.5})	3.97E-16	3.07E-16	4.04E-16	3.09E-16	
Voltage noise (ms ⁻² Hz ^{-0.5})	4.62E-16	2.39E-16	4.25E-16	2.25E-16	
Lorentz noise (ms-2Hz-0.5)	1.22E-16	6.32E-17	1.12E-16	5.95E-17	
Total noise (ms ⁻² Hz ^{-0.5})	6.22E-16	3.95E-16	5.98E-16	3.87E-16	~ ~

Noise Allocation = 20% of LISA noise budget = $0.6 \times 10^{-15} [1 + (f/3mHz)^2] ms^{-2}Hz^{-0.5}$

- Reduce T by ~2 hrs => noise within spec
- If no contribution of secondaries -> net charging rate, total noise P~10-20%
- If V=1mV, the e(t) 0 ~55% , stiffness 0 ~90%, total noise 0 ~20-30%

Comparison to //plate

- // plate approx overestimates
 - Coulomb noise, stiffness, $e_x(t)$ by ~10%
 - f_x(t) by ~40%
- BUT LEVEL OF AGREEMENT IS DEPENDENT ON SENSOR GEOMETRY
- E.g. For similar "Trento torsion" sensor design, FE results (validated in expt: **Carbone** *et al* 2003 **Physical Review Letters** 91 **151101**) & // plate approx give similar levels of agreement
 - But if exclude guard rings for this design, agreement \oplus :
 - $C_{\rm T}({\rm FE}) = C_{\rm T}(//) \times 1.56$ (from 1.30)
 - agreement of 2nd derivatives wrt x, y & z ~ same (~20-30%)
 - //plate approx overestimates:
 - noise, stiffness, $e_x(t)$ by ~30-40%
 - f_x(t) by ~110%
 - guard rings minimise fringing fields
 - exact level of agreement between derived quantities also dependent on other parameters e.g. V
 - => USE CAUTION WHEN EMPLOY // PLATE APPROX



Management of disturbances.

- UV light => discharge the TMs via pe effect.
- Nominally, T ~ 1 day => accn noise & stiffness within budget
- But coherent charging signals above noise target
- To remove charging signals:
 - spectral analysis?
 - continuous discharging of the TMs at a rate exceeding the charging rate? Disadvantage = increased noise

Variation of total noise for LISA solar min, with discharging rate, assuming Q=0



⇒Discharging rate = ~6 × Charging rate, to reach nominal noise budget
⇒Variation in charging rate between solar min & max ~ 50% => plausible that discharging rate could be set high enough to cope with variations in the GCR flux, without exceeding budgets

•Still need to quantify e.g. impact of restoration of equilibrium of charging and discharging currents following changes in the mean charging rate

