

Mission Selection Review summary

M5 Public presentation, 29 April 2021 Frédéric Safa (ESA)

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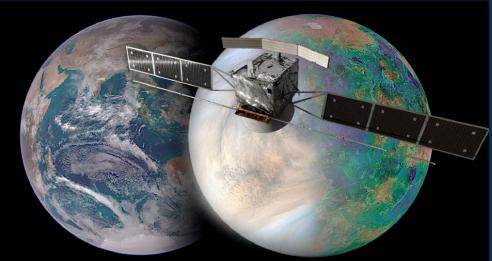


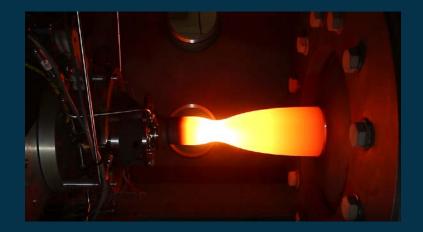
- ESA Independent Review concluding the Phase A, organised in two Panels, Technical and Programmatic, for each mission
- Objectives: Confirm the candidate missions feasibility and enable the SPC to make an informed decision for the M5 mission selection
- Review inputs produced by the Phase A studies
 - For ESA provision: Data packages produced by the two parallel industrial contracts for the spacecraft definition, for each mission candidate
 - For nationally provided payload elements: Data packages produced by the instrument teams (I-PRR)
- NASA provision to Envision was subject to NASA internal reviews, with visibility provided to ESA, and positive outcome

EnVision Technical Assessment: Spacecraft



- Good design maturity achieved
 - ✓ SC 1,400 kg (dry)/ 3 kW, incl. P/L 240 kg / 1,600 W
- Compatibility with Ariane 6.2 confirmed
- Critical functions designed with robust margins and within heritage, e.g. Venus Orbit Insertion, Aerobraking, and thermal management.
- Overall, medium development risk
 - Available heritage & know-how in Europe for both the SC development and operations
 - ✓ High Thrust (1 kN) engine early qualification recommended





LEROS-4 1 kN engine fire test

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EnVision Technical Assessment: Payload



 All instruments are new HW developments, however with recent relevant heritage

=> Payload-driven development schedule

- NASA VenSAR predevelopments initiated to secure the schedule
- VenSpec-H change of detector assembly with respect to NOMAD (ExoMars): Alternative identified but dedicated activities needed in Phase B1

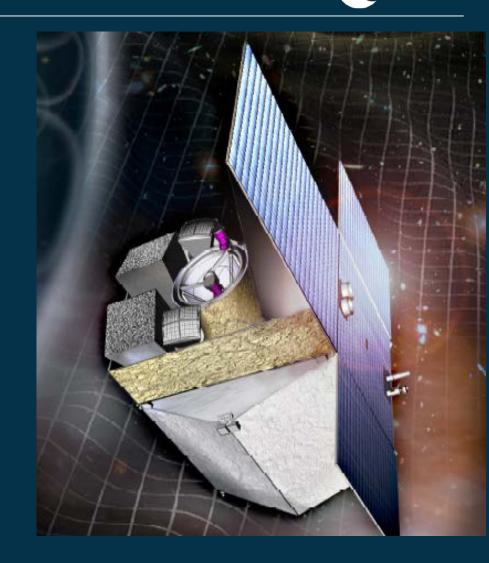


(**) Deployable dipole antenna excluded (12 kg) , procured by ESA

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THESEUS Technical Assessment: Spacecraft

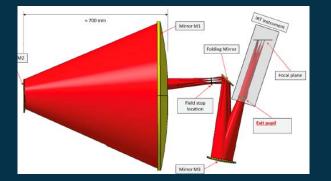
- Satisfactory design maturity achieved
 SC 1,900 kg (dry) / ~ 2 kW, incl. PLM ~ 1,000 kg / 1,000 W
- Compatibility with VEGA-C confirmed
- Spacecraft platform using available technologies and equipment, no predevelopment identified
- Main challenge will be on the development schedule (driven by the payload)
- Overall, medium development risk considering the payload I/F and risks



THESEUS Technical Assessment: Payload



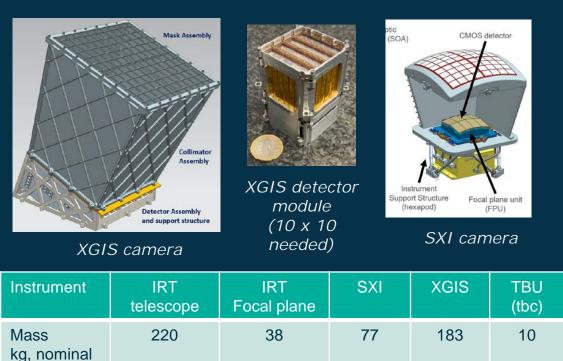
- All instruments are new HW developments, however with heritage
 - => Payload-driven development schedule
 - Complex interface management expected at Prime level, to be de-risked through appropriate early models (e.g. STMs & functional models)
 - ✓ No basic technical feasibility issue
 - IRT payload defines the critical path, closely followed by SXI & XGIS
- Effort needed in Phase B1 to consolidate the development approach and schedule, while providing flexibility for the instrument deliveries





700 mm IRT Korsch telescope layout

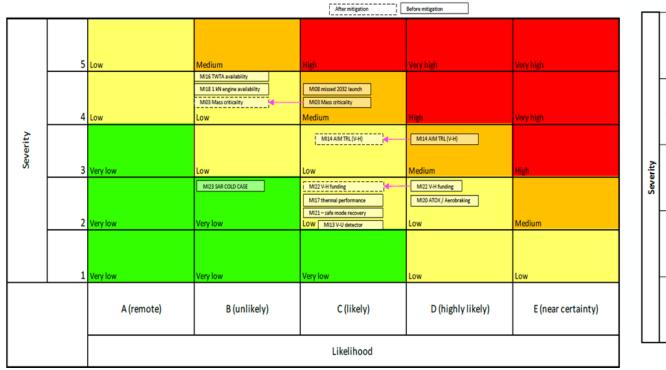
IRT Focal Plane Assembly



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Risk bubble charts





	5	Low	Medium		Very high	Very high
Severity	4	Low	MI10 - Delta development required for CMOS detector to enable Mission Adoption	MI05 Mass growth close to limit of LV adapter qualification (growth of both platform and PL) Medium	High	Very high
	3	Very low	MI02- Payload/SC thermal interfaces MI03- Autonomy concept (GS and OBC)	Low	Medium	MI04-IRT camera delivery need date requires early freezing of interfaces (i.e. early PDR)
	2	Very low	Very low	Low	Low	Medium
	1	Very low	MI01 - S-band unavailability Very low	Very low	Low	Low
		A (remote)	B (unlikely)	C (likely)	D (Highly likely)	E (near certain)
		Likelihood				

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EnVision

- ESA: 1 kN thruster qualification, ideally to be completed by mission adoption
- ESA: Procurement of VenSpec-H detector assembly, to verify and confirm the baseline compatibility with EnVision
- Member States: Payload preparation activities (design & breadboarding for raising TRLs) in continuation of the Phase A to secure the schedule.

THESEUS

- ESA: SXI CMOS detector EM development and performance verification
- ESA: Activities TBD to support the payload interface freezing before adoption
- Member States: Payload preparation activities (design & breadboarding for raising TRLs) in continuation of the Phase A to secure the schedule

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Schedule Analyses



SPC adoption targeted early 2024 for both candidates

- EnVision
 - Baseline schedule judged realistic: launch in June 2032, science operations at Venus starting in Q1 2035. Back-up launch windows available in Dec 2032 and May 2033.
 - May 2031 launch window is possibly feasible (subject to compatibility with all instrument schedules, including VenSAR) and is being investigated.

• THESEUS

- ✓ A launch in 2032 would be possible, subject to overall development approach consolidation and steady progress on instrument preparation activities during the phase B1.
- ✓ No stringent launch window constraints for THESEUS

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Programmatic assessment: responsibility scheme



ESA have consulted the Member States on the payload funding scheme for the Phase A consolidated baseline.

For THESEUS, the ground VHF network provision is not confirmed.

- ✓ Not mandatory for THESEUS core science objectives
- Basic mitigation is to use the SC communication system for the trigger alert and ESA ground station(s), with reduced performance
- All other elements are supported.

Following the MSR, a second consultation is planned on the funding of payload activities in phase B1.

Phase A baseline responsibility scheme				
EnVision	THESEUS			
ESA in charge of: SC development; Launch services; In-orbit operations (MOC and SOC);	ESA in charge of: SC development; Launch services; In-orbit operations (MOC and SOC);			
EnVision Payload: VenSAR: NASA All other instruments: Member States, with the following ESA contributions: 1- VenSpec-H detector assembly, with integrated cryocooler 2- SRS deployable antenna	THESEUS PayloadMember States provision, with the following ESA contributions:1- IRT telescope assembly2- IRT cryo-cooling chain3- IRT focal plane detector and proximity electronics4- SXI CMOS detectors			

Programmatic aspects: ESA cost estimates



ESA CaC estimated by the MSR Programmatic Panel for the Phase A baseline is within 610 M€(e.c. 2021) for both missions.

- \checkmark Exceeding the target by ~ 5%
- Launcher costs based on recent consultations of Arianespace for other projects
- Similar CaC for both missions, but with completely different profiles: Much larger ESA contribution to THESEUS payload, approximately balanced by lower costs for launcher (VEGA-C vs A62) and operations (LEO vs Venus)

ESA and NASA have started their discussion on the implementation details in preparation of the MoU

MSR conclusion



- For both EnVision and THESEUS, the Phase A converged to a feasible baseline by following a design-to-cost approach while preserving the core scientific objectives
- The Member States have been consulted for the funding of nationally provided payload elements. The payload funding scheme is confirmed.
- ESA anticipate a few predevelopment activities for each mission in Phase B1 for securing the implementation schedule and risks
- Predevelopments will also be needed for the nationally provided payload elements, and are expected to be funded in Phase B1
- ESA cost at completion estimates are within **610 M€e.c. 2021** for both missions



The end

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